## PET4I101 ELECTROMAGNETICS ENGINEERING (4 ${ }^{\text {th }}$ Sem ECEETC)

Module-I (10 Hours)

1. Cartesian, Cylindrical and Spherical Coordinate Systems; Scalar and Vector Fields; Line, Surface and Volume Integrals.
2. Coulomb's Law; The Electric Field Intensity; Electric Flux Density and Electric Flux;
Gauss's Law; Divergence of Electric Flux Density: Point Form of Gauss's Law; The Divergence Theorem; The Potential Gradient; Energy Density; Poisson's and Laplace's Equations.
3. Ampere's Magnetic Circuital Law and its Applications; Curl of H; Stokes' Theorem;

Divergence of B; Energy Stored in the Magnetic Field.
Module-II (8 Hours)

1. The Continuity Equation; Faraday's Law of Electromagnetic Induction;

Conduction
Current: Point Form of Ohm's Law, Convection Current; The Displacement Current;
2. Maxwell's Equations in Differential Form; Maxwell's Equations in Integral Form; Maxwell's Equations for Sinusoidal Variation of Fields with Time; Boundary Conditions; The Retarded Potential; The Poynting Vector; Poynting Vector for Fields Varying Sinusoid ally with Time
Module-III (8 Hours)

1. Solution of the One-Dimensional Wave Equation; Solution of Wave Equation for Sinusoid ally Time-Varying Fields; Polarization of Uniform Plane Waves; Fields on the Surface of a Perfect Conductor; Reflection of a Uniform Plane Wave Incident Normally on a Perfect Conductor and at the Interface of Two Dielectric Regions; The Standing Wave Ratio; Oblique Incidence of a Plane Wave at the Boundary between Two Regions; Oblique Incidence of a Plane Wave on a Flat Perfect Conductor and at the Boundary between Two Perfect Dielectric Regions;
Module-IV (8 Hours)
2. Types of Two-Conductor Transmission Lines; Circuit Model of a Uniform Two-

Conductor Transmission Line; The Uniform Ideal Transmission Line; Wave Reflection at a Discontinuity in an Ideal Transmission Line; Matching of Transmission Lines with Load.
Additional Module (Terminal Examination-Internal) (8 Hours)

1. Formulation of Field Equations; Wave Types; the Parallel-Plate Waveguide; the Rectangular Waveguide.
2. Radiation Properties of a Current Element; Radiation Properties of a Half-Wave Dipole; Yagi-Uda Antenna; the Parabolic Reflector Antenna.
3. The Vector Magnetic Potential; Energy stored in a capacitor, Graphical field mapping; Continuity of Current in a Capacitor; Critical Angle of Incidence and Total Reflection; Brewster Angle.

## El3I001 ELECTROMAGNETIC FIELD THEORY ( $3^{\text {rd }}$ Sem I\&E

 Engg.)MODULE - I (13 Hours)

1. Vectors and Fields: Cartesian Coordinate System, Cylindrical and Spherical coordinate
system, Vector Algebra, Scalar and Vector Fields, gradient, divergence, curl operations, The
Laplacian, Divergence Theorem, Stoke’s Theorem ,Useful vector identities and their derivations.
(selected portions from 1.01 to 1.05 of TB-1)
2.: Electric and Magnetic fields: Field due to a line/sheet/volume charge, Biot _Savart
Law,Gauss's Law for Electric Field and Magnetic Field, Fields of electric and magnetic dipoles,
Applications of electrostatics and magnetostatics, Faraday's Law, Ampere's Circuital Law.
(portions 3.4.to 3.6, 4.4.3,4.6, 4.8,4.9, 8.3 to 8.8 and 9.2 of TB-2)
2. Maxwell's Equations: Divergence and Differential Form, Line Integral, Surface Integral and
Integral form, Faradays Law, Ampere's Circuital Law, Gauss's Law for Electric Field and
Magnetic Field. (portions 4.01 to 4.03 of TB-1)
MODULE - II (13 Hours) (Portions 5.01 to 5.13 of TB-1)
3. Wave Propagation in Free Space: The electromagnetic wave equation and its solution,
Uniform Plane Waves, Direction cosines, Concept on TEM mode, Poynting Vector and Power
density
4. Wave Propagation in Material Media: Conductors and Dielectrics, Magnetic Materials, Wave

Equation and Solution, Uniform Plane Waves in Dielectrics and Conductors, Polarization,
Boundary Conditions, Reflection and Transmission of Uniform Plane Waves at the boundary of
two media for normal and oblique incidence, Brewster's angle.
MODULE - III (10 Hours)
6. Transmission Line Analysis: Transmission lines, Circuit representation of a parallel plane
transmission line, Parallel plane transmission lines with loss, E and H about long parallel
cylindrical conductors of arbitrary cross section, Transmission line theory, UHF lines as circuit
elements (portions 7.10 to 7.16 of TB-1)
7. Wave Guide Principles: Rectangular guides, TM waves in rectangular guides, TE waves in
rectangular guides, Impossibility of TEM wave in wave guides, wave impedance and characteristic impedances, Attenuation factor and Q of wave guides, Dielectric Slab Guide
,(portions 8.01 to $8.04,8.08,8.10,8.11$ of TB-1)

## PET6J012 ANTENNAS \& WAVE PROPAGATION ( 6 th Sem ECEETC) <br> MODULE-I

Electromagnetic radiation and antenna fundamentals- Review of electromagnetic
theory: Vector potential, Solution of wave equation, retarded case, Hertizian dipole.
Antenna characteristics: Radiation pattern, Beam solid angle, Directivity, Gain, Input
impedance, Polarization, Bandwidth, Reciprocity, Equivalence of Radiation patterns, Equivalence of Impedances, Effective aperture, Vector effective length, Antenna temperature.
MODULE-II
Wire antennas- Short dipole, Radiation resistance and Directivity, Half wave Dipole,
Monopole, Small loop antennas. Antenna Arrays: Linear Array and Pattern Multiplication,
Two-element Array, Uniform Array, Polynomial representation, Array with nonuniform
Excitation-Binomial Array
MODULE- III
Aperture Antennas- Magnetic Current and its fields, Uniqueness theorem, Field equivalence principle, Duality principle, Method of Images, Pattern properties, Slot antenna,
Horn Antenna, Pyramidal Horn Antenna, Reflector Antenna-Flat reflector, Corner Reflector,
Common curved reflector shapes, Lens Antenna

## MODULE- IV

Special Antennas-Long wire, V and Rhombic Antenna, Yagi-Uda Antenna, Turnstile
Antenna, Helical Antenna- Axial mode helix, Normal mode helix, Biconical Antenna, Log
periodic Dipole Array, Spiral Antenna, Microstrip Patch Antennas.
Antenna Measurements- Radiation Pattern measurement, Gain and Directivity Measurements, Anechoic Chamber measurement.
ADDITIONAL MODULE (TERMINAL EXAMINATION-INTERNAL)
Radio wave propagation- Calculation of Great Circle Distance between any two points on
earth, Ground Wave Propagation, Free-space Propagation, Ground Reflection, Surface waves, Diffraction, Wave propagation in complex Environments, Tropospheric Propagation,
Tropospheric Scatter. Ionospheric propagation: Structure of ionosphere, Sky waves, skip
distance, Virtual height, Critical frequency, MUF, Electrical properties of ionosphere, Effects
of earth's magnetic fields, Faraday rotation, Whistlers.

## PEL3I001 ELECTROMAGNETIC THEORY ( ${ }^{\text {rd }}$ Sem EEE)

Module - I (8 hours)
University Portion (80\%):
Co-ordinate systems \& Transformation: Cartesian co-ordinates, circular cylindrical coordinates, spherical co-ordinates.
Vector Calculus: Differential length, Area \& volume, Line surface and volume Integrals, Del
operator, Gradient of a scalar, Divergence of a vector \& divergence theorem, curl of a vector \&
Stoke's theorem, laplacian of a scalar (Text Book 1: Chapter- 1, Chapter-2)
College/Institute Portion (20\%):
Field: Scalar Field and Vector Field. Or related advanced topics as decided by the concerned faculty
teaching the subject.
Module - II (11 hours)
University Portion (80\%):
Electrostatic Fields: Coulomb's Law, Electric Field Intensity, Electric Fields due to point, line,
surface and volume charge, Electric Flux Density, Gauss's Law - Maxwell's Equation, Applications
of Gauss's Law, Electric Potential, Relationship between E and V -Maxwell's Equation An Electric
Dipole \& Flux Lines, Energy Density in Electrostatic Fields., Electrostatic Boundary Value

Problems: Possion's \& Laplace's Equations, Uniqueness theorem, General procedures for solving
possion's or Laplace's Equation. (Textbook-1: Chapter- 3, 4, 5.1 to 5.5)

## College/Institute Portion (20\%):

Nature of current and current density, the equation of continuity. Or related advanced topics as
decided by the concerned faculty teaching the subject.
Module - III (8 hours)
University Portion (80\%):
Magnatostatic Fields: Magnetic Field Intensity, Biot-Savart's Law, Ampere's circuit lawMaxwell
Equation, applications of Ampere's law, Magnetic Flux Density-Maxwell's equations. Maxwell's
equation for static fields, Magnetic Scalar and Vector potentials. (Textbook-1:
Chapter- 6.1 to 6.8)
College/Institute Portion (20\%): (2 hours)
Energy in Magnetic Field Or related advanced topics as decided by the concerned faculty teaching the
subject.
Module - IV (7 hours)
University Portion (80\%):
Electromagnetic Fields and Wave Propagation: Faraday's Law, Transformer \& Motional Electromagnetic Forces, Displacement Current, Maxwell's Equation in Final forms, Time Varying
Potentials, Time-Harmonic Field. Electromagnetic Wave Propagation: Wave Propagation in lossy
Dielectrics, Plane Waves in loss less Dielectrics, Power \& pointing vector. (Textbook-1:
Chapter-8.1
to 8.7 , Ch. 9.1 to $9.3 \& 9.6$ )
College/Institute Portion (20\%):
General Wave Equation, Plane wave in dielectric medium, free space, a conducting medium, a good
conductor and good dielectric, Polarization of wave. Or related advanced topics as decided by the concerned faculty teaching the subject.

## MODULE-I

## Elementary Coordinate Systems

We know that the cylindrical coordinates are given by $(\rho, \phi, z)$. Basically, the $x y$ - plane is transformed into polar coordinates while the $z$-axis remains the same in both the cylindrical and the rectangular system of coordinates. We solve a couple of numerical to explain the basics of the three orthogonal coordinate systems, namely, rectangular, cylindrical and spherical.

Ex.1.1 Relate the differentials in cylindrical coordinates to the rectangular coordinates.
$x=\rho \cos \phi \quad y=\rho \sin \phi \quad \rho=\sqrt{x^{2}+y^{2}}, \tan \phi=\frac{y}{x}$
$\frac{\partial}{\partial x}=\cos \phi \frac{\partial}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial}{\partial \phi}$
$\frac{\partial}{\partial y}=\sin \phi \frac{\partial}{\partial \rho}+\frac{\cos \phi}{\rho} \frac{\partial}{\partial \phi}$
$A_{x}=A_{\rho} \cos \phi-A_{\phi} \sin \phi$
$A_{y}=A_{\rho} \sin \phi+A_{\phi} \cos \phi$
Ex.1.2 Compute $\nabla . \mathbf{A}=\frac{\partial A_{x}}{\partial x}+\frac{\partial A_{y}}{\partial y}+\frac{\partial A_{z}}{\partial z}$ in cylindrical coordinates
Soln: Substitution of the results in Ex. 1.1 in the above expression gives us

$$
\begin{aligned}
& \nabla . \mathbf{A}=\left(\cos \phi \frac{\partial}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial}{\partial \phi}\right)\left(A_{\rho} \cos \phi-A_{\phi} \sin \phi\right) \\
& +\left(\sin \phi \frac{\partial}{\partial \rho}+\frac{\cos \phi}{\rho} \frac{\partial}{\partial \phi}\right)\left(A_{\rho} \sin \phi+A_{\phi} \cos \phi\right)+\frac{\partial A_{z}}{\partial z}
\end{aligned}
$$

This is equal to
$\nabla . \mathbf{A}=\cos ^{2} \phi \frac{\partial A_{\rho}}{\partial \rho}-\frac{\sin \phi(-\sin \phi)}{\rho} A_{\rho}+\frac{\sin ^{2} \phi}{\rho} \frac{\partial A_{\phi}}{\partial \phi}$
$+\sin ^{2} \phi \frac{\partial A_{\rho}}{\partial \rho}+\frac{\cos \phi}{\rho} \cdot A_{\rho} \cos \phi+\frac{\cos ^{2} \phi}{\rho} \frac{\partial A_{\phi}}{\partial \phi}+\frac{\partial A_{z}}{\partial z}$
Therefore,

$$
\begin{aligned}
& \nabla \cdot \mathbf{A}=\left(\cos ^{2} \phi+\sin ^{2} \phi\right) \frac{\partial A_{\rho}}{\partial \rho}+\left(\frac{\sin ^{2} \phi}{\rho}+\frac{\cos ^{2} \phi}{\rho}\right) A_{\rho}+\left(\frac{\cos ^{2} \phi}{\rho}+\frac{\sin ^{2} \phi}{\rho}\right) \frac{\partial A_{\phi}}{\partial \phi}+\frac{\partial A_{z}}{\partial z} \\
& =\frac{\partial A_{\rho}}{\partial \rho}+\frac{A_{\rho}}{\rho}+\frac{\partial A_{\phi}}{\rho \partial \phi}+\frac{\partial A_{z}}{\partial z} \\
& =\frac{1}{\rho} \frac{\partial}{\partial \rho}\left(\rho A_{\rho}\right)+\frac{\partial A_{\phi}}{\rho \partial \phi}+\frac{\partial A_{z}}{\partial z}
\end{aligned}
$$

Ex 1.3. Transform the equation $\frac{\partial^{2} \Psi}{\partial x^{2}}+\frac{\partial^{2} \Psi}{\partial y^{2}}=0$ into polar coordinates.
Soln: We know $x=\rho \cos \phi \quad y=\rho \sin \phi \quad \rho=\sqrt{x^{2}+y^{2}}, \phi=\tan ^{-1} \frac{y}{x}$
Hence, $\frac{\partial \rho}{\partial x}=\frac{1}{2}\left(x^{2}+y^{2}\right)^{-1 / 2} .2 x=\frac{x}{\sqrt{x^{2}+y^{2}}}=\frac{\rho \cos \phi}{\rho}=\cos \phi$
And $\frac{\partial \phi}{\partial x}=\frac{\partial}{\partial x}\left(\tan ^{-1} \frac{y}{x}\right)=\frac{1}{1+\left(\frac{y}{x}\right)^{2}} \cdot\left(-\frac{y}{x^{2}}\right)=-\frac{y}{x^{2}+y^{2}}=-\frac{\rho \sin \phi}{\rho^{2}}=-\frac{\sin \phi}{\rho}$
Thus,
$\frac{\partial}{\partial x}=\cos \phi \frac{\partial}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial}{\partial \phi}$
$\frac{\partial}{\partial y}=\sin \phi \frac{\partial}{\partial \rho}+\frac{\cos \phi}{\rho} \frac{\partial}{\partial \phi}$
Thus $\frac{\partial \Psi}{\partial x}=\frac{\partial \Psi}{\partial \rho} \cdot \frac{\partial \rho}{\partial x}+\frac{\partial \Psi}{\partial \phi} \cdot \frac{\partial \phi}{\partial x}=\cos \phi \frac{\partial \Psi}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial \Psi}{\partial \phi}$
Therefore,
$\frac{\partial^{2} \Psi}{\partial x^{2}}=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial x}\right)=\left(\cos \phi \frac{\partial}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial}{\partial \phi}\right)\left(\cos \phi \frac{\partial \Psi}{\partial \rho}-\frac{\sin \phi}{\rho} \frac{\partial \Psi}{\partial \phi}\right)$
$=\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}-\frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}-\frac{\sin \cos \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\sin ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}$
$-\cos \phi \sin \phi \frac{\partial \Psi}{\partial \phi} \frac{\partial}{\partial \rho}\left(\frac{1}{\rho}\right)-\sin \phi \cos \phi \frac{\partial \Psi}{\partial \phi} \frac{\partial}{\partial \rho}\left(\frac{1}{\rho}\right)-\frac{\sin \phi}{\rho} \frac{\partial}{\partial \phi}(\cos \phi) \frac{\partial \Psi}{\partial \rho}$
$=\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}-2 \frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\sin ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}-2 \cos \phi \sin \phi\left(-\frac{1}{\rho^{2}}\right)-\frac{\sin \phi}{\rho}(-\sin \phi) \frac{\partial \Psi}{\partial \rho}$
$=\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}-2 \frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\sin ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}+\frac{2 \cos \phi \sin \phi}{\rho^{2}} \frac{\partial \Psi}{\partial \phi}+\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}$

Similarly,

$$
\begin{aligned}
& \frac{\partial^{2} \Psi}{\partial y^{2}}=\frac{\partial}{\partial y}\left(\frac{\partial \Psi}{\partial y}\right)=\left(\sin \phi \frac{\partial}{\partial \rho}+\frac{\cos \phi}{\rho} \frac{\partial}{\partial \phi}\right)\left(\sin \phi \frac{\partial \Psi}{\partial \rho}+\frac{\cos \phi}{\rho} \frac{\partial \Psi}{\partial \phi}\right) \\
& =\sin ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}+\frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\cos \phi}{\rho} \frac{\partial}{\partial \phi}(\sin \phi) \frac{\partial \Psi}{\partial \rho}+\frac{\sin \cos \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi} \\
& +\frac{\cos ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}+\sin \phi \cos \phi \frac{\partial}{\partial \rho}\left(\frac{1}{\rho}\right) \frac{\partial \Psi}{\partial \phi}+\cos \phi \sin \phi \frac{\partial}{\partial \rho}\left(\frac{1}{\rho}\right) \frac{\partial \Psi}{\partial \phi} \\
& =\sin ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}+2 \frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\cos ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}-2 \frac{\sin \phi \cos \phi}{\rho^{2}} \frac{\partial \Psi}{\partial \phi}+\frac{\cos ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}
\end{aligned}
$$

Adding the two terms, we get

$$
\begin{aligned}
& \frac{\partial^{2} \Psi}{\partial x^{2}}+\frac{\partial^{2} \Psi}{\partial x^{2}}=\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}-2 \frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\sin ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}+\frac{2 \cos \phi \sin \phi}{\rho^{2}} \frac{\partial \Psi}{\partial \phi}+\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}+ \\
& \sin ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}+2 \frac{\cos \phi \sin \phi}{\rho} \frac{\partial^{2} \Psi}{\partial \rho \partial \phi}+\frac{\cos ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}-2 \frac{\sin \phi \cos \phi}{\rho^{2}} \frac{\partial \Psi}{\partial \phi}+\frac{\cos ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}} \\
& =\left(\cos ^{2} \phi+\sin ^{2} \phi\right) \frac{\partial^{2} \Psi}{\partial \rho^{2}}+\left(\frac{\sin ^{2} \phi}{\rho}+\frac{\cos ^{2} \phi}{\rho}\right) \frac{\partial \Psi}{\partial \rho}+\frac{\sin ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}}+\frac{\cos ^{2} \phi}{\rho^{2}} \frac{\partial^{2} \Psi}{\partial \phi^{2}} \\
& =\frac{\partial^{2} \Psi}{\partial \rho^{2}}+\frac{1}{\rho} \frac{\partial \Psi}{\partial \rho}+\frac{\partial^{2} \Psi}{\rho^{2} \partial \phi^{2}} \\
& =\frac{1}{\rho} \frac{\partial}{\partial \rho}\left(\rho \frac{\partial \Psi}{\partial \rho}\right)+\frac{\partial^{2} \Psi}{\rho^{2} \partial \phi^{2}}
\end{aligned}
$$

Ex. 1.4 Derive a suitable expression for $\frac{\partial \Psi}{\partial x}$ in cylindrical coordinates.
Soln: We have
$x=\rho \cos \phi \quad y=\rho \sin \phi \quad \rho=\sqrt{x^{2}+y^{2}}, \phi=\tan ^{-1} \frac{y}{x}$ $\frac{\partial \Psi}{\partial x}=\frac{\partial \Psi}{\partial \rho} \frac{\partial \rho}{\partial x}$
Now, $\frac{\partial \rho}{\partial x}=\frac{1}{2}\left(x^{2}+y^{2}\right)^{-\frac{1}{2}} \cdot 2 x=\frac{x}{\sqrt{x^{2}+y^{2}}}=\frac{\rho \cos \phi}{\rho}=\cos \phi$

$$
\begin{aligned}
& \begin{aligned}
& \frac{\partial \Psi}{\partial x}=\frac{\partial \Psi}{\partial \rho} \frac{\partial \rho}{\partial x}=\cos \phi \frac{\partial \Psi}{\partial \rho} \\
& \Rightarrow \frac{\partial^{2} \Psi}{\partial x^{2}}=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial x}\right)=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial \rho} \cdot \frac{\partial \rho}{\partial x}\right) \\
&=\frac{\partial \Psi}{\partial \rho} \frac{\partial^{2} \rho}{\partial x^{2}}+\frac{\partial \rho}{\partial x} \frac{\partial^{2} \Psi}{\partial x \partial \rho} \\
& \frac{\partial^{2} \rho}{\partial x^{2}}=\frac{\sqrt{x^{2}+y^{2}}-\frac{1}{2} x\left(x^{2}+y^{2}\right)^{-\frac{1}{2}} \cdot 2 x}{x^{2}+y^{2}} \\
&=\frac{1}{\sqrt{x^{2}+y^{2}}}-\frac{x^{2}}{\left(x^{2}+y^{2}\right)^{\frac{3}{2}}} \\
& \text { Now, }=\frac{1}{\rho}-\frac{\rho^{2} \cos ^{2} \phi}{\rho^{3}} \\
&=\frac{1}{\rho}\left(1-\cos ^{2} \phi\right) \\
&=\frac{1}{\rho} \sin ^{2} \phi
\end{aligned}
\end{aligned}
$$

Substitution of this in the above gives us

$$
\begin{aligned}
& \frac{\partial \Psi}{\partial x}=\frac{\partial \Psi}{\partial \rho} \frac{\partial^{2} \rho}{\partial x^{2}}+\frac{\partial \rho}{\partial x} \frac{\partial^{2} \Psi}{\partial x \partial \rho} \\
& =\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}+\cos \phi \frac{\partial}{\partial \rho}\left(\frac{\partial \Psi}{\partial x}\right) \\
& =\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}+\cos \phi \cdot \frac{\partial}{\partial \rho}\left(\cos \phi \frac{\partial \Psi}{\partial \rho}\right) \\
& =\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}+\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}} \\
& =\cos ^{2} \phi \frac{\partial^{2} \Psi}{\partial \rho^{2}}+\frac{\sin ^{2} \phi}{\rho} \frac{\partial \Psi}{\partial \rho}
\end{aligned}
$$

Ex. 1.5 For a spherical wave, the displacement $\Psi$ is a function of $r$ and $t$ where $r$ is the magnitude of the distance from a fixed point. Obtain a general equation for the spherical wave.

Soln: We need to show the following
$\nabla^{2} \Psi=\frac{\partial^{2} \Psi}{\partial r^{2}}+\frac{2}{r} \frac{\partial \Psi}{\partial r}=\frac{1}{r^{2}} \frac{\partial}{\partial r}\left(r^{2} \frac{\partial \Psi}{\partial r}\right)$
We understand that

$$
\begin{aligned}
\frac{\partial \Psi}{\partial x}= & \frac{\partial \Psi}{\partial r} \frac{\partial r}{\partial x} \\
& r^{2}=x^{2}+y^{2}+z^{2} \\
& \Rightarrow 2 r \frac{\partial r}{\partial x}=2 x \\
\text { Now, } & \Rightarrow \frac{\partial r}{\partial x}=\frac{x}{r} \\
& \Rightarrow \frac{\partial \Psi}{\partial x}=\frac{\partial \Psi}{\partial r} \frac{x}{r}
\end{aligned}
$$

Let us go for the second order differential which is obtained by differentiating the above function once more with respect to $x$.Thus,
$\frac{\partial^{2} \Psi}{\partial x^{2}}=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial x}\right)=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial r} \frac{x}{r}\right)=\frac{\partial}{\partial x}\left(\frac{\partial \Psi}{\partial r}\right) \frac{x}{r}+\frac{\partial}{\partial x}\left(\frac{x}{r}\right) \frac{\partial \Psi}{\partial r}$
$=\frac{\partial}{\partial r}\left(\frac{\partial \Psi}{\partial r}\right) \frac{\partial r}{\partial x} \frac{x}{r}+\frac{1}{r} \frac{\partial \Psi}{\partial r}$
$=x \frac{\partial}{\partial r}\left(\frac{1}{r} \frac{\partial \Psi}{\partial r}\right) \frac{\partial r}{\partial x}+\frac{1}{r} \frac{\partial \Psi}{\partial r}$
$=x \frac{x}{r} \frac{\partial}{\partial r}\left(\frac{1}{r} \frac{\partial \Psi}{\partial r}\right)+\frac{1}{r} \frac{\partial \Psi}{\partial r}$
$=\frac{x^{2}}{r}\left(\frac{1}{r} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{1}{r^{2}} \frac{\partial \Psi}{\partial r}\right)+\frac{1}{r} \frac{\partial \Psi}{\partial r}$
$=\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{x^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{x^{2}}{r^{3}} \frac{\partial \Psi}{\partial r}$
Similarly, we get for the $y$ component
$\frac{\partial^{2} \Psi}{\partial y^{2}}=\frac{\partial}{\partial y}\left(\frac{\partial \Psi}{\partial y}\right)=\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{y^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{y^{2}}{r^{3}} \frac{\partial \Psi}{\partial r}$
and for the $z$ component
$\frac{\partial^{2} \Psi}{\partial z^{2}}=\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{z^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{z^{2}}{r^{3}} \frac{\partial \Psi}{\partial r}$
Therefore,

$$
\begin{aligned}
& \frac{\partial^{2} \Psi}{\partial x^{2}}+\frac{\partial^{2} \Psi}{\partial y^{2}}+\frac{\partial^{2} \Psi}{\partial z^{2}}=\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{x^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{x^{2}}{r^{3}} \frac{\partial \Psi}{\partial r} \\
& +\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{y^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{y^{2}}{r^{3}} \frac{\partial \Psi}{\partial r} \\
& +\frac{1}{r} \frac{\partial \Psi}{\partial r}+\frac{z^{2}}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{z^{2}}{r^{3}} \frac{\partial \Psi}{\partial r} \\
& =\frac{3}{r} \frac{\partial \Psi}{\partial r}+\frac{\left(x^{2}+y^{2}+z^{2}\right)}{r^{2}} \frac{\partial^{2} \Psi}{\partial r^{2}}-\frac{\left(x^{2}+y^{2}+z^{2}\right)}{r^{3}} \frac{\partial \Psi}{\partial r} \\
& =\frac{\partial^{2} \Psi}{\partial r^{2}}+\frac{2}{r} \frac{\partial \Psi}{\partial r} \Theta x^{2}+y^{2}+z^{2}=r^{2} \\
& =\frac{1}{r^{2}} \frac{\partial}{\partial r}\left(r^{2} \frac{\partial \Psi}{\partial r}\right)
\end{aligned}
$$

Ex 1.6. Plane $z=10 \mathrm{~m}$ carries a surface charge density of $20 \mathrm{nc} / \mathrm{m}^{2}$. What is the electric field at the origin?

Soln: The unit normal to the plane of $z=10 \mathrm{~m}$ is $a_{z}$. As this is above the origin, the unit normal for the electric field at the origin becomes $-a_{z}$. Thus, the electric field strength becomes

$$
E=\frac{\rho_{s}}{2 \varepsilon_{0}}\left(-a_{z}\right)=\frac{20 \times 10^{-9}}{2 \times \frac{1}{36 \pi} \times 10^{-9}}\left(-a_{z}\right)=-360 \pi a_{z} \mathrm{~V} / \mathrm{m}
$$

Ex 1.7. Two point charges $(\mathrm{Q} 1=\mathrm{Q}, \mathrm{Q} 2=2 \mathrm{Q})$ and an infinite ground plane are shown in the following figure. Find the ratio of the forces experienced by both $Q_{1}$ and $Q_{2}$.


Soln: The ground plane acts like a mirror for the two charges and charges of opposite polarity with same magnitude are induced on both sides of the plane. Hence the diagram looks as the following

$$
-Q_{2}=-2 Q
$$



Force experienced by $Q_{1}$ would be due to three charges; $-Q_{1}$ at $2 d, Q_{2}=2 Q$ at $3 d$ and $-Q_{2}=-2 Q$ at $d$. The net force is

$$
\begin{aligned}
& F_{1}=\frac{1}{4 \pi \varepsilon_{0}}\left[\frac{Q_{1}\left(-Q_{1}\right)}{(2 d)^{2}}+\frac{Q_{1} Q_{2}}{(3 d)^{2}}+\frac{Q_{1}\left(-Q_{2}\right)}{(d)^{2}}\right] \\
& \Rightarrow \frac{1}{4 \pi \varepsilon_{0}}\left[\frac{Q \cdot(-Q)}{4 d^{2}}+\frac{Q \cdot 2 Q}{9 d^{2}}+\frac{Q(-2 Q)}{d^{2}}\right] \\
& =\frac{1}{4 \pi \varepsilon_{0}} \cdot \frac{Q^{2}}{d^{2}}\left(\frac{2}{9}-\frac{1}{4}-2\right) \\
& =-\frac{1}{4 \pi \varepsilon_{0}} \cdot \frac{Q^{2}}{d^{2}} \cdot \frac{73}{36}
\end{aligned}
$$

Ex 1.8. The electric field strength at a distance point, $P$, due to a point charge, $+q$ located on the origin, is $100 \mathrm{~V} / \mathrm{m}$. If the point charge is now enclosed by a perfectly conducting metal sheet sphere whose centre is at the origin, compute the electric field strength outside the sphere.

Soln: With a conducting metal sheet surrounding the point charge $+q$, equal and opposite charges are induced in inner and outer surfaces of the sphere. The net charge enclosed by Gaussian surface of radius ' $r$ ' is again $+q$. Thus, $\oint \mathbf{D} . d \mathbf{s}=q$
$\Rightarrow D=\frac{q}{4 \pi r^{2}}$
$\Rightarrow E=\frac{q}{4 \pi \varepsilon_{0} r^{2}}$
Thus, the electric field is same with and without the conducting sphere.

Ex 1.9. A straight current carrying conductor and two conducting loops $A$ and $B$ are shown in the following diagram. What are the directions of the induced currents in A and $B$ if the current in the straight conductor is decreasing?


Soln: The direction of the current as shown produces a magnetic field in the anticlockwise direction that encircles the straight wire. It becomes anticlockwise in B whereas A has clockwise encircled magnetic flux. If the current is decreasing in the straight conductor, the induced currents in the loops would be in such a direction so as to oppose this change. Hence, B would get a current in the clockwise direction and A would have its induced current in the anticlockwise direction.

Ex 1.10 In spherical coordinates region 1 is $r<a$, region 2 is $a<r<b$ and region 3 is $r>b$. Regions 1 and 3 are free space, while $\mu_{r 2}=500$. Given $\mathbf{B}_{1}=0.20 \hat{\mathbf{r}}$, find $\mathbf{H}$ in each region.

Soln: We know that, the normal components are related as

$$
\mu_{0} \mu_{r 1} \mathbf{H}_{1}=\mu_{0} \mu_{r 2} \mathbf{H}_{2}=\mu_{0} \mu_{r 3} \mathbf{H}_{3}
$$

Hence,

$$
\begin{aligned}
& \mathbf{H}_{1}=\frac{\mathbf{B}_{1}}{\mu_{0} \mu_{r 1}}=\frac{0.2}{\mu_{0}}=\mu_{0} \mu_{r 2} \mathbf{H}_{2} \\
& \Rightarrow \mathbf{H}_{2}=\frac{0.2}{500 \mu_{0}}=0.04 \times 10^{-2}=\frac{4 \times 10^{-4}}{\mu_{0}} \mathrm{~A} / \mathrm{m}
\end{aligned}
$$

and the corresponding magnetic field in region 3 is
$\mu_{0} \mu_{r 1} \mathbf{H}_{1}=\mu_{0} \mu_{r 3} \mathbf{H}_{3} \Rightarrow \mathbf{H}_{1}=\mathbf{H}_{3}=\frac{0.2}{\mu_{0}} \mathrm{~A} / \mathrm{m}$
Ex 1.11. Obtain the unit vector along the direction of propagation of a wave, the displacement of which is given by $\Psi(x, y, z, t)=a \cos (2 x+3 y+4 z-5 t)$.

Soln: The unit vector along the direction of propagation of the wave is corresponding to the plane $2 x+3 y+4 z=$ constant

Hence, it becomes

$$
\begin{aligned}
& \frac{2}{\sqrt{2^{2}+3^{2}+4^{2}}} \hat{\mathbf{x}}+\frac{3}{\sqrt{2^{2}+3^{2}+4^{2}}} \hat{\mathbf{y}}+\frac{4}{\sqrt{2^{2}+3^{2}+4^{2}}} \hat{\mathbf{z}} \\
& =\frac{2}{\sqrt{29}} \hat{\mathbf{x}}+\frac{3}{\sqrt{29}} \hat{\mathbf{y}}+\frac{4}{\sqrt{29}} \hat{\mathbf{z}}
\end{aligned}
$$

Ex 1.12. The displacement associated with a three-dimensional wave is given by $\Psi(x, y, z, t)=a \cos \left(\frac{\sqrt{3}}{2} k x+\frac{1}{2} k y-\omega t\right)$. Find the direction of wave propagation.
Soln: The generalized wave equation is
$\Psi(r, t)=\operatorname{Re}\left\{E_{0} \exp (\mathbf{k} \cdot \mathbf{r}-\omega t)\right\}$
For the given problem, the two components of the vector $\mathbf{k}$ are $k_{x}=k_{y}=k$ and the two components of the position vector are
$r_{x}=\cos ^{-1}\left(\frac{\sqrt{3}}{2}\right)=30^{0}$
$r_{y}=\cos ^{-1}\left(\frac{1}{2}\right)=60^{\circ}$
The wave propagates in the $x y$ - plane making an angle of $30^{\circ}$ with the positive $x$ - axis and $60^{\circ}$ with the positive $y$ - axis.

Ex 1.13 Find the force per unit area on the surface of a conductor, with surface charge density $\sigma$ in the presence of an electric field $\left(\mathbf{a}_{n}\right)$ is the unit outward normal to the surface.

Soln: The electric field at a given point due to the surface charge density is
$\mathbf{E}=\frac{\sigma}{2 \varepsilon_{0}} \mathbf{a}_{n}$
Force per unit area due to another sheet charge is
$\mathbf{F}=E . q=\frac{\sigma}{2 \varepsilon_{0}} . \sigma \mathbf{a}_{n}=\frac{\sigma^{2}}{2 \varepsilon_{0}} \mathbf{a}_{n}$
Ex 1.14 Inside a right circular cylinder, $\mu_{r 1}=1000$. The exterior is free space. If $\mathbf{B}_{1}=2.5 \hat{\boldsymbol{\varphi}}(\mathrm{~T})$ inside the cylinder, determine $\mathbf{B}_{2}$ just outside.

Soln: We know that the tangential component of $\mathbf{H}$ is continuous across the boundary. Thus,
$\frac{\mathbf{B}_{1}}{\mu_{0} \mu_{r 1}}=\frac{\mathbf{B}_{2}}{\mu_{0} \mu_{r 2}}$ and hence, $\mathbf{B}_{2}=\frac{2.5}{1000} \hat{\boldsymbol{\varphi}} \mathrm{mT}$
Ex 1.15. In free space, $\mathbf{B}=B_{m} e^{j(\omega t+\beta z)} \hat{\mathbf{y}}$. Find the corresponding electric field.

Soln: We know that
$\nabla \times \mathbf{H}=\varepsilon_{0} \frac{\partial \mathbf{E}}{\partial t}=\nabla \times \frac{\mathbf{B}}{\mu_{0}}=\varepsilon_{0} \frac{\partial \mathbf{E}}{\partial t} \Rightarrow \nabla \times \mathbf{B}=\mu_{0} \varepsilon_{0} \frac{\partial \mathbf{E}}{\partial t}=j \omega \mu_{0} \varepsilon_{0} \mathbf{E}$
If we expand the above expression we get
$\left[\begin{array}{ccc}\hat{\mathbf{x}} & \hat{\mathbf{y}} & \hat{\mathbf{z}} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ B_{x} & B_{y} & B_{z}\end{array}\right]=\left(\frac{\partial B_{z}}{\partial y}-\frac{\partial B_{y}}{\partial z}\right) \hat{\mathbf{x}}-\left(\frac{\partial B_{z}}{\partial x}-\frac{\partial B_{x}}{\partial z}\right) \hat{\mathbf{y}}+\left(\frac{\partial B_{y}}{\partial x}-\frac{\partial B_{x}}{\partial y}\right) \hat{\mathbf{z}}=j \omega \mu_{0} \varepsilon_{0} \mathbf{E}$
In the above expression we understand that the magnetic field has an $y$ component only and further $\frac{\partial}{\partial x}=0=\frac{\partial}{\partial y}, \frac{\partial}{\partial z}=j \beta$. Therefore,
$\frac{\partial B_{y}}{\partial z}=j \beta B_{m} e^{j(\omega t+\beta z)}$
and $\mathbf{E}=-\frac{j \beta B_{m}}{j \omega \mu_{0} \varepsilon_{0}} e^{j(\omega t+\beta z)} \hat{\mathbf{x}}=-\frac{\beta B_{m}}{\omega \mu_{0} \varepsilon_{0}} e^{j(\omega t+\beta z)} \hat{\mathbf{x}}$
As $\frac{1}{\mu_{0} \varepsilon_{0}}=\frac{\omega^{2}}{\beta^{2}} \Rightarrow \mu_{0} \varepsilon_{0}=\frac{\beta^{2}}{\omega^{2}}$
Substitution of the above in the electric field expression we get
$\mathbf{E}=-\frac{\beta B_{m}}{\omega \mu_{0} \varepsilon_{0}} e^{j(\omega t+\beta z)} \hat{\mathbf{x}}=-\frac{\beta B_{m}}{\omega\left(\beta^{2} / \omega^{2}\right)} e^{j(\omega t+\beta z)} \hat{\mathbf{x}}=-\frac{\omega B_{m}}{\beta} e^{j(\omega t+\beta z)} \hat{\mathbf{x}}$
This expression shows that the electric field $\mathbf{E}$ and the magnetic field $\mathbf{H}$ are in space quadrature but in time phase. Both of the fields propagate with the same velocity in the positive $z$ direction.

Ex.1.16 Find the electric field in the region between the two cones as shown in Fig.1.1.


Fig.1.1 Figure for Ex.1.16
Soln: The potential is observed to be a function of the azimuthal angle $\theta$ only and is a constant with respect to $r$ and $\phi$. Hence, Laplace's equation becomes
$\frac{1}{r^{2} \sin \theta} \frac{d}{d \theta}\left(\sin \theta \frac{d V}{d \theta}\right)=0$
$\frac{d}{d \theta}\left(\sin \theta \frac{d V}{d \theta}\right)=0$
$\Rightarrow \sin \theta \frac{d V}{d \theta}=K$
$\Rightarrow \frac{d V}{d \theta}=K \csc \theta$
$\Rightarrow V=K \ln (\csc \theta-\cot \theta)+C$
The term $\csc \theta-\cot \theta=\frac{1}{\sin \theta}-\frac{\cos \theta}{\sin \theta}=\frac{1-\cos \theta}{\sin \theta}=\frac{2 \sin ^{2}(\theta / 2)}{2 \sin (\theta / 2) \cdot \cos (\theta / 2)}=\tan (\theta / 2)$

$$
\begin{aligned}
& \frac{d V}{d \theta}=K \csc \theta \\
& \Rightarrow V=-K \ln |\csc \theta+\cot \theta|=-K \ln \left|\frac{1}{\sin \theta}+\frac{\cos \theta}{\sin \theta}\right| \\
& =-K \ln \left|\frac{1+\cos \theta}{\sin \theta}\right|=-\ln \left|\frac{2 \cos ^{2}\left(\frac{\theta}{2}\right)}{2 \sin \left(\frac{\theta}{2}\right) \cdot \cos \left(\frac{\theta}{2}\right)}\right|=\ln \cot \left(\frac{\theta}{2}\right)+C \\
& \mathbf{E}=-\nabla V=-\frac{1}{r} \frac{\partial V}{\partial \theta} \mathbf{a}_{\theta}=-\frac{1}{r} \frac{\partial}{\partial \theta}\left\{\left[K \ln \cot \left(\frac{\theta}{2}\right)+C\right]\right\} \mathbf{a}_{\theta} \\
& \frac{1}{r} \frac{\partial V}{\partial \theta}=-\frac{K}{2 r} \tan \left(\frac{\theta}{2}\right) \csc ^{2}\left(\frac{\theta}{2}\right)=-\frac{K}{2 r} \frac{\sin \left(\frac{\theta}{2}\right)}{\cos \left(\frac{\theta}{2}\right)} \cdot \frac{1}{\sin ^{2}\left(\frac{\theta}{2}\right)} \\
& \mathbf{E}=\frac{K}{r \sin \theta} \mathbf{a}_{\theta}
\end{aligned}
$$

The potential between the two cones equidistant from the origin is therefore
$V=-\int_{\theta_{1}}^{\pi-\theta_{1}} \mathbf{E} \cdot d \mathbf{l}=-\int_{\theta_{1}}^{\pi-\theta_{1}} \frac{K}{r \sin \theta} \cdot r d \theta$
$=-K \int_{\theta_{1}}^{\pi-\theta_{1}} \frac{K}{\sin \theta} \cdot d \theta$
$=K \ln \cot \left(\frac{\theta}{2}\right)$
Therefore, the solution to the potential becomes
$V=K \ln (\csc \theta-\cot \theta)+C$
$=K \ln \left[\tan \left(\frac{\theta}{2}\right)\right]+C$
Further, we observe that, $V=V_{1}$ at $\theta=\theta_{1}$ and $V=0$ at $\theta=\theta_{2}$. Therefore,
$V_{1}=K \ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]+C$ and
$0=K \ln \left[\tan \left(\frac{\theta_{2}}{2}\right)\right]+C$
$\Rightarrow C=-K \ln \left[\tan \left(\frac{\theta_{2}}{2}\right)\right]$
From these two, we obtain

$$
\begin{aligned}
& V_{1}=K \ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]-K \ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right] \\
& \Rightarrow K=\frac{V_{1}}{\ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]-\ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]}
\end{aligned}
$$

The potential is, therefore

$$
V=V_{1} \frac{\ln \left[\tan \left(\frac{\theta}{2}\right)\right]-\ln \left[\tan \left(\frac{\theta_{2}}{2}\right)\right]}{\ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]-\ln \left[\tan \left(\frac{\theta_{2}}{2}\right)\right]}
$$

The electric field is evaluated as

$$
\begin{aligned}
& \mathbf{E}=-\nabla V=-\frac{1}{r} \frac{\partial V}{\partial \theta} \mathbf{a}_{\theta}=-\frac{1}{r} \frac{\partial}{\partial \theta}\left\{\left[K \ln \tan \left(\frac{\theta}{2}\right)+C\right]\right\} \mathbf{a}_{\theta} \\
& \frac{1}{r} \frac{\partial V}{\partial \theta}=\frac{K}{2 r} \cot \left(\frac{\theta}{2}\right) \sec ^{2}\left(\frac{\theta}{2}\right)=\frac{K}{2 r} \frac{\cos \left(\frac{\theta}{2}\right)}{\sin \left(\frac{\theta}{2}\right) \cdot} \cdot \frac{1}{\cos ^{2}\left(\frac{\theta}{2}\right)} \\
& =\frac{K}{r \sin \theta}=\frac{V_{1}}{\ln \left[\tan \left(\frac{\theta_{1}}{2}\right)\right]-\ln \left[\tan \left(\frac{\theta_{2}}{2}\right)\right] r \sin \theta}=\frac{V_{1}}{\ln \left(\tan 10^{\circ}\right)-\ln \left(\tan 80^{\circ}\right) r \sin \theta} \\
& \mathbf{E}=\frac{0.288}{r \sin \theta} V_{1} \mathbf{a}_{\theta}
\end{aligned}
$$

The general electric field is evaluated as

$$
\begin{aligned}
& \mathbf{E}=-\nabla V=-\frac{1}{r} \frac{\partial V}{\partial \theta} \mathbf{a}_{\theta}=-\frac{1}{r}\left[K \ln \tan \left(\frac{\theta}{2}\right)+C\right] \mathbf{a}_{\theta} \\
& =-\frac{K}{2 r_{1}} \cot \left(\frac{\theta}{2}\right) \sec ^{2}\left(\frac{\theta}{2}\right) \mathbf{a}_{\theta}=-\frac{K}{2 r} \frac{\cos \left(\frac{\theta}{2}\right)}{\sin \left(\frac{\theta}{2}\right)} \cdot \frac{1}{\cos ^{2}\left(\frac{\theta}{2}\right)} \mathbf{a}_{\theta} \\
& =-\frac{K}{r \sin \theta} \mathbf{a}_{\theta}
\end{aligned}
$$

Hence, the total electric flux bounded by the conical surface is

$$
D=\iint_{S} \varepsilon \mathbf{E} \cdot d \mathbf{s}=\varepsilon \int_{0}^{\pi} \int_{0}^{R} \frac{K}{r \sin \theta} \cdot r \sin \theta d \phi d r
$$

$$
=\pi \varepsilon K R=Q
$$

The total charge enclosed by the surface. Therefore, the capacitance per unit length is given as
$C=\frac{Q}{V}=\frac{\pi \varepsilon K}{K \ln \left[\cot \left(\frac{\theta}{2}\right)\right]}=\frac{\pi \varepsilon}{\ln \left[\cot \left(\frac{\theta}{2}\right)\right]}$
The integral of $\csc x$ is $-\ln |\csc x+\cot x|$

In spherical coordinates, under the assumption that the potential at a given point is a function of the radial distance from the source, then we have
$\nabla^{2} V=\frac{1}{r^{2}} \frac{\partial}{\partial r}\left(r^{2} \frac{\partial V}{\partial r}\right)=0$
$\Rightarrow\left(r^{2} \frac{\partial V}{\partial r}\right)=K$
$\Rightarrow \frac{\partial V}{\partial r}=\frac{K}{r^{2}}$
$\Rightarrow V=-\frac{K}{r}+C$

The two most important theorems often used to evaluate radiated fields due to various conductor configurations are Gauss's divergence theorem and Stoke's theorem. The former as the name suggests is related to the divergence of a vector and the latter is related to the curl of a vector. The divergence theorem states that
$\oint_{S} \mathbf{D} \cdot d \mathbf{s}=\iiint_{V} \rho d v=\iiint_{V} \nabla \cdot \mathbf{D} d v$
This gives us
$\nabla . \mathbf{D}=\rho$


Fig.1.2 Path difference between a source point and an observation point
The potential $V$ due to a continuous charge distribution $\rho$ contained in the volume $V_{0}$ is expressed as
$V(r)=\frac{1}{4 \pi \varepsilon} \iiint_{V_{0}} \frac{\rho\left(\mathbf{r}^{\prime}\right)}{R} d v^{\prime}$
where $d \nu^{\prime}=d x^{\prime} d y^{\prime} d z^{\prime}$. The distance $R$ is computed as follows:
$R^{2}=\left\lfloor\left(x-x^{\prime}\right)^{2}+\left(y-y^{\prime}\right)^{2}+\left(z-z^{\prime}\right)^{2}\right\rfloor$
$=\left[\left(r-r^{\prime} \cos \varphi\right)^{2}+\left(r^{\prime} \sin \varphi\right)^{2}\right]$
$=\left(r^{2}-2 r r^{\prime} \cos \varphi+r^{\prime 2} \cos ^{2} \varphi+r^{\prime 2} \sin ^{2} \varphi\right)(\mathrm{c})$
$=\left(r^{2}-2 r r^{\prime} \cos \varphi+r^{\prime 2}\right)$
$=r^{2}\left(1-2 \frac{r^{\prime}}{r} \cos \varphi+\frac{r^{\prime 2}}{r^{2}}\right)$
We can simplify the distance by writing
$R^{-1}=\left(r^{2}-2 r r^{\prime} \cos \varphi+r^{\prime 2}\right)^{-\frac{1}{2}}$
$=r^{-1}\left(1-2 \frac{r^{\prime}}{r} \cos \varphi+\frac{r^{\prime 2}}{r^{2}}\right)^{-\frac{1}{2}}$
$=r^{-1}\left[1+2 \frac{r^{\prime}}{r} \cos \varphi+\frac{1}{2}\left(3 \cos ^{2} \varphi-1\right)\left(\frac{r^{\prime}}{r}\right)^{2}+\ldots\right]$
$=r^{-1} \sum_{m=0}^{\infty} P_{m}\left(\frac{r^{\prime}}{r}\right)^{m} \cos \varphi$
We note that the term $r^{\prime} \cos \varphi$ also represents the projection of $r^{\prime}$ onto $r$ and therefore, can be expressed as $\mathbf{r}^{\prime} \cdot \mathbf{a}_{\mathbf{r}}$.


Fig.1.3 An electric dipole that makes use of the principle outlined in Fig.1.2 (An infinitesimal electric dipole)

Two approximations are usually carried out for the far field of the antennas. These are made to the amplitude and phase of the radiated fields. The approximations are:
$R \approx r-r^{\prime} \cos \varphi$ for the phase term and
$R \approx r$ for the amplitude term
The charge dipole, hence may be expressed as
$\rho(\mathbf{r})=q \delta(x) \delta(y)\left[\delta\left(z-\frac{d}{2}\right)-\delta\left(z+\frac{d}{2}\right)\right]$
The far field potential as obtained from (D) may be expressed as
$4 \pi \varepsilon V(\mathbf{r})=\frac{1}{r^{2}} \iiint_{V} \rho\left(\mathbf{r}^{\prime}\right)\left(\mathbf{r}^{\prime} \cdot \mathbf{a}_{\mathbf{r}}\right) d v$
$=\frac{1}{r^{2}} \iiint_{V} q \delta(x) \delta(y)\left[\delta\left(z-\frac{d}{2}\right)-\delta\left(z+\frac{d}{2}\right)\right] z^{\prime} \cos \theta d x^{\prime} d y^{\prime} d z^{\prime}$
$=\frac{q d \cos \theta}{r^{2}}$
This is due to the fact that $\int \delta(x) d x=1$ and
$\int\left[\delta\left(z-\frac{d}{2}\right)-\delta\left(z+\frac{d}{2}\right)\right] d z$
$=\frac{d}{2}-\left(-\frac{d}{2}\right)=d$
This follows from the fact that the delta function $\delta(x)$ assumes a value of unity only at $x=0$ and is zero at all other values of $x$. Further, the area bounded by a dirac delta function is equal to one.
$\int_{-\infty}^{\infty} \delta(x) d x=1$
This result due to an electric dipole consisting of two fixed static charges gives us the potential at a faraway point. We may observe that the negative of the gradient of the potential gives us the electric field evaluated as
$\mathbf{E}=-\nabla V$
$\Rightarrow E_{r}=-\frac{\delta V}{\delta r} a_{r}=-\frac{2 q d \cos \theta}{4 \pi \varepsilon r^{3}}$
$E_{\theta}=-\frac{\delta V}{r \delta \theta}=\frac{q d}{4 \pi \varepsilon r^{3}} \sin \theta$
and $E_{\phi}=-\frac{\delta V}{r \sin \theta \delta \phi}=0$
These fields are static and they are constants both in magnitude as well as in direction at a certain distance $r$ from the source. They also vary inversely as the cube of the distance which means that at really far distances from the source, their magnitude becomes too insignificant to carry energy or power. These cannot act as radiated fields as required in a power transfer by wireless methods. We must have a field that varies at most as the inverse of the squared distance from the source. Moreover, the field needs to be time varying so that we get radiated fields. Also a magnetic field is required to carry energy
from one point to another in a wireless manner. We take up the generation of the electromagnetic fields first to understand this wireless transfer of energy by electric and associated magnetic fields generated due to time varying sources.

Stoke's law as related to the curl of a vector states that
$\oint_{S}(\nabla \times \mathbf{H}) . d \mathbf{s}=\oint \mathbf{H} . d \mathbf{s}$
The standard wave equations are

$$
\nabla^{2} \mathbf{A}-\mu \varepsilon \frac{\partial^{2} \mathbf{A}}{\partial t^{2}}=-\mu \mathbf{J} \text { and } \nabla^{2} V-\mu \varepsilon \frac{\partial^{2} V}{\partial t^{2}}=-\frac{\rho}{\varepsilon}
$$

Under the assumption of sinusoidal steady state, the wave equations become
$\nabla^{2} \mathbf{A}+\mu \varepsilon \omega^{2}=-\mu \mathbf{J}$
$\nabla^{2} V+\omega^{2} \mu \varepsilon=-\frac{\rho}{\varepsilon}$
The magnetic vector potential $\mathbf{A}$ at a given observation point far away from a current source similarly can be evaluated with the aid of Fig. (A) except that the charge as a source should be replaced by a current source that has a current density of $J \mathrm{~A} / \mathrm{m}^{2}$ established and therefore, we can similarly write
$\mathbf{A}(r)=\frac{1}{4 \pi} \iiint_{V_{0}} \frac{\mu \mathbf{J}\left(\mathbf{r}^{\prime}\right)}{R} d v^{\prime}$
As we may observe, the distance of the observation point from the source $R$ is given by (B).

## Transmission Lines

The transit time effect is an important issue in the characterization of high frequency behavior of electric circuits. At low frequencies, the wavelength is long and the time period of a given signal is also long. This means that, if a source in the form of a periodic waveform is present at one particular point of the circuit at a given point in time, it would be available on some different point almost at the same time.


We observe that, the points marked A-G require a finite amount of time $t_{r}$ to reach the destination end, also known as the load. This is known as the retrace time. The time period of the signal is $T$. We may see that, these two times are almost comparable. By the time point A reaches the destination, the source signal changes to a point marked load end. As the voltage goes to different values between the source and the load end, which are some distance apart, we get a phase difference between these two points. A similar situation holds good for all other points. This is due to the fact that the time to reach the load and the time period of the source are comparable. If we increase the frequency of the source, we may anticipate that the phase change would be more. To describe it mathematically, we define the electrical length of the circuit as $L / \lambda$ where $L$ is the physical length between the source and the load and $\lambda$ is the wavelength of the signal (voltage or current) impressed at the source end. The quasi-static regime is defined for $L / \lambda \ll 1$ which is typical of low frequency lumped circuit parameters. As we may see, the wavelength of the signal is much longer than the physical distance between the source and the load. Let us define the transit time as $t_{r}=L / \mathrm{v}$. Hence, in this case,
$t_{r}=L / \mathrm{v}=\frac{L}{\lambda f}=\frac{L}{\lambda} \cdot T$
$\Rightarrow \frac{t_{r}}{T}=\frac{L}{\lambda} \ll 1$
This simply means that the signal impressed at the source end reaches the load in a very small amount of time. In other words, the transit time is such that, during this time, the signal at the source end does not change much as it has a longer time period as compared to the transit time.

However, as the frequency of the source increases, the time period gets correspondingly shorter and by the time one particular point of the source signal reaches the load end, the source signal changes to a different value, The signal amplitudes are at different values between the source and the load end and this gives rise to the resonance region characterized by $\frac{t_{r}}{T}=\frac{L}{\lambda} \sim 1$. The transit time becomes an appreciable fraction of the time period which means that the source and the load ends experience different amplitudes of the signal which gives rise to a phase difference between them and it becomes a function of the physical distance or the electrical length of the circuit. Due to the phase difference between two different points of the circuit (observed at the same time of course), the effect is a distribution of frequency sensitive elements like inductance and capacitance over the length of the circuit. As practical circuits exhibit losses, there are resistances and conductances as well distributed throughout the length of the circuit. This gives rise to the so called distributed circuit analysis.
$\Delta V=-(R \Delta x+j \omega L \Delta x) I$
$\Delta I=-(G \Delta x+j \omega C \Delta x) V$
These equations can be written as
$\frac{\Delta V}{\Delta x}=-(R+j \omega L) I$ and $\frac{\Delta I}{\Delta x}=-(R+j \omega L) V$
For the lumped circuit analysis to hold good, $\Delta x \rightarrow 0$. Doing so, we get
$\lim _{\Delta x \rightarrow 0} \frac{\Delta V}{\Delta x}=\frac{d V}{d x}=-(R+j \omega L) I$ and similarly,
$\lim _{\Delta x \rightarrow 0} \frac{\Delta I}{\Delta x}=\frac{d I}{d x}=-(G+j \omega C) V$
The voltage and the currents on a transmission line are governed by two coupled first order differential equations as given above. To solve those, let us differentiate both w.r.t. $x$ so that we have
$\frac{d^{2} V}{d x^{2}}=-(R+j \omega L) \frac{d}{d x} I$ and $\frac{d^{2} I}{d x^{2}}=-(G+j \omega C) \frac{d}{d x} V$
Substituting the expressions for the first order differentials in these,
$\frac{d^{2} V}{d x^{2}}=(R+j \omega L)(G+j \omega C) V$
$\frac{d^{2} I}{d x^{2}}=(R+j \omega L)(G+j \omega C) I$
Let us write
$\gamma^{2}=(R+j \omega L)(G+j \omega C)$
So that, now the two equations become
$\frac{d^{2} V}{d x^{2}}=\gamma^{2} V$
and
$\frac{d^{2} I}{d x^{2}}=\gamma^{2} I$

Both the equations are governed by the same second order differential equations. As we might see, the constant $\gamma$ is a constant at a given frequency. The solutions to the voltage and current along the line are:
$V=V^{+} \exp (-\gamma x)+V^{-} \exp (\gamma x)$
$I=I^{+} \exp (-\gamma x)+I^{-} \exp (\gamma x)$
The $V^{+}, V^{-}, I^{+}, I^{-}$are arbitrary constants that are to be evaluated by appropriate boundary conditions. These constants are in general complex and their phases represent temporal phases with respect to some reference time. As we may see, in these expressions, the time harmonic function is implicit. The total solution, is therefore,
$V=\left[V^{+} \exp (-\gamma x)+V^{-} \exp (\gamma x)\right] \exp (j \omega t)$
$I=\left[I^{+} \exp (-\gamma x)+I^{-} \exp (\gamma x)\right] \exp (j \omega t)$
From these two expressions we see that, they represent a standing wave on the line; one goes along the positive $x$-axis given by $\exp (-\gamma x)$ with an amplitude of $V^{+}$and the other along the negative $x$-axis given by $\exp (\gamma x)$ having a amplitude of $V^{-}$. A similar condition holds for the current flowing in the line. What we see is that instead of talking about voltage and current a given point on the line, now we have two waves in two opposite directions with two different peak amplitudes. The most important observation is that, both voltage and current are functions of the given point in space; $x$-axis. This is unlike the low frequency characterization of voltage and current where both are functions of time only. They are temporal functions. However, for the transmission lines, the voltage and currents are also spatial functions. Therefore, we may guess that, the impedance is also point specific; it is a function of space.

Let us write
$\gamma=\alpha+j \beta$
The constant $\alpha$ is known as the attenuation constant of the line; as it attenuates the wave as it propagates along the line. The unit of $\alpha$ is neper $/ \mathrm{m}$. This means that if the voltage travels a length of 1 m from the source, it decays to a value of $e^{-1}$ of its amplitude at the source. The attenuation, in dB is, therefore, $20 \log \left(e^{-1}\right)=-8.68 \mathrm{~dB}$.
To evaluate the four constants, we proceed as
$-\gamma V^{+} \exp (-\gamma x)+\gamma V^{-} \exp (\gamma x)=-(R+j \omega L)\left[I^{+} \exp (-\gamma x)+I^{-} \exp (\gamma x)\right]$
Equating respective terms on both sides, we have, for the coefficient of $\exp (-\chi x)$
$\gamma V^{+}=(R+j \omega L) I^{+}$and similarly, for the coefficient of $\exp (\gamma x)$
$-\gamma V^{+}=-(R+j \omega L) I^{-}$
$\frac{V^{+}}{I^{+}}=\frac{(R+j \omega L)}{\gamma}=\frac{(R+j \omega L)}{\sqrt{(R+j \omega L)(G+j \omega C)}}=\sqrt{\frac{R+j \omega L}{G+j \omega C}}$ and
$\frac{V^{-}}{I^{-}}=-\frac{(R+j \omega L)}{\gamma}=\sqrt{\frac{R+j \omega L}{G+j \omega C}}$
The quantity $\sqrt{\frac{R+j \omega L}{G+j \omega C}}$ has the dimensions of impedance denoted as
$Z_{0}=\sqrt{\frac{R+j \omega L}{G+j \omega C}}$
This $Z_{0}$ is known as the characteristic impedance. The parameters $\gamma$ and $Z_{0}$ are known as the secondary parameters of the transmission line.

## MODULE-II

## 2. GENERATION OF ELECTROMAGNETIC WAVES

An antenna is a structure usually made from a good conducting material that has been designed to have a shape and size such that it will radiate EM power in an efficient manner. Time-varying currents radiate EM waves. Thus an antenna is a structure on which time-varying currents can be excited with a relatively large amplitude when the antenna is connected to a suitable source usually by means of a transmission line or waveguide. In order to radiate efficiently, the minimum size of the antenna must be comparable to the wavelength. A generic structure for the generation of the EM waves is shown in Fig.2.1.


Fig.2.1 A Generic Block Diagram of generation of EM waves through low frequency to high frequency conversion

At frequencies above 40 MHz , communication is essentially limited to line-of-sight paths. A typical LOS link is that used for TV broadcasting. Another example is the LOS microwave link used in the telephone service. In order for an antenna to radiate into a small angular region and thereby provide a higher concentration of power at the receiving site, it must be physically large in terms of wavelength. In the microwave band where the wavelength is in the range of 3 to 30 cm , large reflector antennas with gains as large as 40 to 50 dB are quite common. With a large available gain, the transmitter power can be reduced accordingly. It is not unusual to use transmitter powers of a few watts or even as
low as a few milliwatts in the microwave band. There is also less atmospheric noise at the higher frequencies so smaller signal levels can be used.

The waveguide is a structure that guides electromagnetic waves from a source to a destination. They transport electromagnetic energy over long distances with a minimum amount of signal loss. The waveguides basically are of two types: metal based and dielectric based. The frequency range of operation of the metallic waveguides range from a few tens of kHz to a few tens of GHz . Beyond these frequencies, these waveguides have excessive losses and become inefficient in the transportation of energy. Dielectric waveguides, on the other hand, beyond millimeter wavelength operation, are used to transport electromagnetic energy, however, in the form of optical signals or light waves. We have either slab or cylindrical waveguides. The slab waveguides are used in thin film applications and integrated optical devices. The optical fibers are cylindrical waveguides that have been widely used in both communication and instrumentation applications due to a number of advantages. The interested reader can refer to standard texts on optical fiber based systems for an in depth knowledge about the working and applications of the optical fibers.

### 2.1 Vector Potential approach in forming a field expression

The vector potential set up by the infinitesimal current element at a distance $r$ from the origin of the coordinates system under consideration is
$A(P)=\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l) a_{z}$
We may observe from (2.1) that $\beta$ represents the wave number given as $\beta=\frac{2 \pi}{\lambda}$.


Fig.2.2 Relationship between the unit vectors $a_{r}, a_{z}$ and $a_{\theta}$
From Fig.2.1, in spherical coordinates, $a_{z}=a_{r} \cos \theta-a_{\theta} \sin \theta$
Substitution of (2.2) in (2.1) we have,

$$
\begin{equation*}
A(P)=\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l)\left[a_{r} \cos \theta-a_{\theta} \sin \theta\right] \tag{2.3}
\end{equation*}
$$

the vector magnetic potential as defined above gives a corresponding magnetic field in the region outside of the current source as
$H=\frac{1}{\mu} \Delta \times A$
which gives the $(\Delta \times A)_{\phi}=\frac{1}{r}\left[\frac{\partial}{\partial r}\left(r A_{\theta}\right)-\frac{\partial A_{r}}{\partial \theta}\right]$
The validity of (2.5) follows from the fact that a current flowing through a conductor along the positive $z$-axis produces a magnetic field in the form of loops contained in the $x y$-plane and has a $\phi$ component only. We, therefore have, from elementary magnetostatics,
$H_{\phi}=\frac{I}{2 \pi r} a_{\phi}$
From (2. 3), we note that,
$A_{\theta}=-\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l) \sin \theta a_{\theta}$ and
$A_{r}=\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l) \cos \theta$
Further, $r A_{\theta}=-\frac{\mu e^{-j \beta r}}{4 \pi}(I d l) \sin \theta a_{\theta}$
Thus, $\frac{\partial}{\partial r}\left(r A_{\theta}\right)=\frac{\mu j \beta e^{-j \beta r}}{4 \pi}(I d l) \sin \theta a_{\theta}$
and $-\frac{\partial}{\partial \theta}\left(A_{r}\right)=\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l) \sin \theta$
Use of (2.9) gives (2.5) as

$$
\begin{equation*}
(\Delta \times A)_{\phi}=\frac{1}{r}\left[\frac{\mu j \beta e^{-j \beta r}}{4 \pi}(I d l) \sin \theta+\frac{\mu e^{-j \beta r}}{4 \pi r}(I d l) \sin \theta\right]=\mu \frac{I d l}{4 \pi} \beta^{2} \sin \theta e^{-j \beta r}\left[\frac{j}{\beta r}+\frac{1}{\beta^{2} r^{2}}\right] \tag{2.10}
\end{equation*}
$$

If current flows through the positive-z direction, we note that, $A_{\theta}=\cos \theta \cos \phi a_{x}+\cos \theta \sin \phi a_{y}-\sin \theta a_{z}$
Under the assumption that $\frac{\partial}{\partial \phi}=0$, we write (2.11) as
$A_{\theta}=-\sin \theta a_{z}$
Thus, $(\Delta \times A)_{\phi}=-\frac{I d l}{r^{2}} \sin \theta a_{\phi}$
If $A=\frac{I d l}{4 \pi r} a_{z}$, then $\frac{\partial A_{z}}{\partial r}=-\frac{I d l}{4 \pi r^{2}}$
This gives us
$(\Delta \times A)_{\phi}=-\sin \theta \frac{\partial A_{z}}{\partial r}$
The result given in (2.14) gives the far field distribution for most of the practical antennas.

The elementary magnetic field due to an elementary current $I d l$ flowing in the positive $z$-direction at a distance $r$ from the origin is given as
$d H=\frac{I d l \times a_{R}}{4 \pi R^{2}}$
As we note from above, for a current in the positive z -direction, this becomes
$d H=\frac{I d l \times a_{R}}{4 \pi R^{2}}=\frac{I d z a_{z} \times a_{R}}{4 \pi R^{2}}$
Substitution of (2.2) in (2.16) gives us
$d H=\frac{I d z a_{z} \times a_{r}}{4 \pi r^{2}}=\frac{I d z\left(a_{r} \cos \theta-a_{\theta} \sin \theta\right) \times a_{r}}{4 \pi r^{2}}=\frac{I d z(\sin \theta)\left(-a_{\phi}\right)}{4 \pi r^{2}}=\frac{I d l(\sin \theta) a_{\phi}}{4 \pi r^{2}}$
as for the current flow such as this, $d z=d l$
(2.17) also gives the noted Biot-Savart law in the spherical coordinate.

For a current element $I d l$, the phasor magnetic vector potential, from (2.13) is

$$
\begin{equation*}
A_{z}=\frac{\mu I d l}{4 \pi r} e^{-j \beta r} \tag{2.18}
\end{equation*}
$$

Differentiating (2.18) w.r.t. $r$, we obtain
$\frac{\partial A_{z}}{\partial r}=-j \beta \frac{\mu I d l}{4 \pi r} e^{-j \beta r}+\frac{\mu I d l}{4 \pi r} e^{-j \beta r}\left(-\frac{1}{r^{2}}\right)=-\frac{\mu I d l}{4 \pi r} e^{-j \beta r}\left(j \beta+\frac{1}{r^{2}}\right)$
Hence, the magnetic field becomes
$H_{\phi}=\frac{\mu I d l}{4 \pi r} \sin \theta e^{-j \beta r}\left(j \beta+\frac{1}{r^{2}}\right)$
We note from (2.20) that, the magnetic field has a $\phi$ component due to a current flow in the z -direction. The associated electric field is given as
$E=\frac{1}{\varepsilon} \int \nabla \times H d t$
$(\nabla \times H)_{\phi}=\frac{1}{r}\left[\frac{\partial}{\partial r}\left(r H_{\phi}\right)-\frac{\partial H_{r}}{\partial \theta}\right]$
From (2.22), we understand that, the electric field has an $r$ and a $\theta$ component. The $r$ component is obtained from the first term in the RHS by taking its curl and subsequent integration of it with respect to time. This becomes then,

$$
\begin{align*}
\frac{\partial H_{\phi}}{\partial r} & =\frac{I d l \sin \theta}{4 \pi}\left[\frac{\omega \sin \omega\left(t-\frac{r}{v}\right)}{r^{2} v}+\frac{\omega^{2} \cos \omega\left(t-\frac{r}{v}\right)}{r v^{2}}-2 \frac{\cos \omega\left(t-\frac{r}{v}\right)}{r^{3}}+\frac{\omega \sin \omega\left(t-\frac{r}{v}\right)}{r^{2} v}\right] \\
& =\frac{I d l \sin \theta}{4 \pi}\left[2 \frac{\omega \sin \omega\left(t-\frac{r}{v}\right)}{r^{2} v}+\frac{\omega^{2} \cos \omega\left(t-\frac{r}{v}\right)}{r v^{2}}-2 \frac{\cos \omega\left(t-\frac{r}{v}\right)}{r^{3}}\right] \tag{2.23}
\end{align*}
$$

The $\theta$ component of the electric field is obtained as the time integral of (2.23). Thus,
$E_{\theta}=\frac{I d l \sin \theta}{4 \pi \varepsilon} \int-\frac{\partial H_{\phi}}{\partial r} d t$
This is

$$
\begin{equation*}
E_{\theta}=\frac{I d l \sin \theta}{4 \pi \varepsilon}\left[2 \frac{\cos \omega\left(t-\frac{r}{v}\right)}{r^{2} v}-\frac{\omega \sin \omega\left(t-\frac{r}{v}\right)}{r v^{2}}+2 \frac{\sin \omega\left(t-\frac{r}{v}\right)}{\omega r^{3}}\right] \tag{2.25}
\end{equation*}
$$

Let $t-\frac{r}{v}=t^{\prime}$
Then (2.25) is expressed as
$H_{\phi}=\frac{I d l \sin \theta}{4 \pi}\left(\frac{-\omega \sin \omega t^{\prime}}{r v}+\frac{\cos \omega t^{\prime}}{r^{2}}\right)$
Similarly, $E_{r}=\frac{1}{\varepsilon} \int-\frac{1}{r} \cdot \frac{\partial H_{\phi}}{\partial \theta} d t=\frac{1}{\varepsilon} \int-\frac{1}{r} \cdot \frac{I d l \cos \theta}{4 \pi}\left(\frac{-\omega \sin \omega t^{\prime}}{r v}+\frac{\cos \omega t^{\prime}}{r^{2}}\right) d t$
This becomes

$$
\begin{equation*}
E_{r}=\frac{1}{\varepsilon} \int-\frac{1}{r} \cdot \frac{I d l \cos \theta}{4 \pi \varepsilon}\left(\frac{-\omega \sin \omega t^{\prime}}{r v}+\frac{\cos \omega t^{\prime}}{r^{2}}\right) d t=\frac{I d l \cos \theta}{4 \pi \varepsilon}\left(\frac{\cos \omega t^{\prime}}{r^{2} v}+\frac{\sin \omega t^{\prime}}{\omega r^{3}}\right) \tag{2.29}
\end{equation*}
$$



Fig.2.3 Schematic illustration of formation of waves
The electric field lines between two conductors show a pattern similar to that of the voltage source connected across them. For example, if the voltage is sinusoidal, then the lines of force would exhibit a similar sinusoidal pattern in the space between the two conductors and this is illustrated in Fig.2.3. The repetition of the positive and negative half cycles of the electric field take place at a rate equal to the frequency of the applied source. The relative strength of the field is indicated by the closeness or the farness of the lines in Fig.2.1 and the direction is shown by the arrows. An upward going arrow indicates the positive half cycle and a negative going arrow indicates the negative half cycle of the source. The bunching together of the lines at certain points show the occurrence of positive and negative peaks. As the electric field varies in time, it also generates an associated magnetic field as indicated by Maxwell's law. We, therefore, have both electric and magnetic fields inside the space between the two conductors, At the flared end of the conductors, the waves see a change in impedance as a result of which some of them are reflected and travel toward the source. The interference of the forward traveling wave and the reflected wave would produce a standing wave pattern which is not desirable from an antenna point of view. This is simply due to the factt that we would like the antenna or the system of conductors to radiate waves or make possible only a forward flow of energy.

From all the above discussions, we understand that, for a current flow in the positive $z$ direction, the magnetic field has a $\phi$ component whereas the electric field has components both along the $r$ and the $\theta$ direction. This implies that, the electric field is contained in the $r-\theta$ plane which is perpendicular to the magnetic field oriented completely along the $\phi$ direction. One field gives rise to the other and hence they exist simultaneously. The $H_{\phi}$ field as given in (2.20) has two components; i.e. the first term being proportional to inverse of the distance $r$ from the origin and the second term being proportional to the square of the inverse distance. For small $r$ close to the current carrying conductor, the second term predominates and this is called induction field. This field is also the magnetic field that would be produced by a current of $I \cos \omega t$ due to the application of Biot-Savart law except for the factor of $t^{\prime}$. This field is a function of $t^{\prime}$ is explained by the fact that, a finite time for propagation is required on the part of the field. This causes an energy that is stored in the field during one quarter of a cycle and returned to the circuit during the next.

The first term is known as radiation or distant field and dominates for large; i.e. at greater distances from the current element. This field is not present for steady currents. This results from the finite time of propagation which is of no concern in the steady-field case. This term is responsible for a flow of energy away from the source. The two fields have equal amplitudes at that value of $r$ which makes
$\frac{1}{r^{2}}=\frac{\omega}{r v}$
which implies that, $r=\frac{v}{\omega}=\frac{v}{2 \pi f}=\frac{\lambda}{2 \pi} \approx \frac{\lambda}{6}$

The amplitudes of the radiation fields of an electric current element Idl are, from the second term of (2.25)
$E_{\theta}=\frac{\omega I d l \sin \theta}{4 \pi \varepsilon v^{2} r}=\frac{\eta I d l \sin \theta}{2 \lambda r}=\frac{60 \pi I d l \sin \theta}{r \lambda}$
The first term of (2.27) gives us the amplitude of the magnetic field as
and $H_{\phi}=\frac{\omega I d l \sin \theta}{4 \pi v r}=\frac{I d l \sin \theta}{2 \lambda r}$
These two relations in(2.32) and (2.33) show that the amplitudes of the electric field and the magnetic field are in time phase but space quadrature. They are related by
$\frac{E_{\theta}}{H_{\phi}}=\eta$
which is a constant known as the intrinsic impedance of the medium. For example, the intrinsic impedance $\eta$ corresponding to free space is found to be $120 \pi \Omega$. Knowing one field will help in computing the other.

From the above it is understood that, a magnetic field is produced only when the current is changing and these two fields exist simultaneously. The two fields are in space quadrature. A static charge produces a steady electric field which does not change with respect to time and hence cannot generate an induced magnetic field. Similarly, a dc current flowing in and around a conductor produces a steady magnetic field that remains constant in magnitude and direction at a given distance from the conductor. This kind of steady magnetic field cannot induce an electric field. We, therefore, see that steady electric and magnetic fields are independent of each other and do not contribute to producing a wave of any kind.

### 2.2 Proof of the outward energy flow due to the radiation term

The instantaneous power flow per unit area at the point P is given by the Poynting vector at that point. The radial Poynting vector is expressed as
$P_{r}=E_{\theta} H_{\phi}=\frac{I^{2} d l^{2} \sin ^{2} \theta}{16 \pi^{2} \varepsilon}\binom{\frac{\sin \omega t^{\prime} \cos \omega t^{\prime}}{\omega r^{5}}+\frac{\cos ^{2} \omega t^{\prime}}{r^{4} c}-\frac{\omega \sin \omega t^{\prime} \cos \omega t^{\prime}}{r^{3} c^{2}}}{-\frac{\sin ^{2} \omega t^{\prime}}{r^{4} c}-\frac{\omega \sin \omega t^{\prime} \cos \omega t^{\prime}}{r^{3} c^{2}}+\frac{\omega^{2} \sin ^{2} \omega t^{\prime}}{r^{2} c^{3}}}$
$P_{r}=\frac{I^{2} d l^{2} \sin ^{2} \theta}{16 \pi^{2} \varepsilon}\binom{\frac{\sin 2 \omega t^{\prime}}{2 \omega r^{5}}+\frac{\cos 2 \omega t^{\prime}}{r^{4} c}-\frac{\omega \sin 2 \omega t^{\prime}}{r^{3} c^{2}}}{+\frac{\omega^{2}\left(1-\cos 2 \omega t^{\prime}\right)}{2 r^{2} c^{3}}}$
Hence, the average radial power flow is given by the integral of the instantaneous value over one complete cycle. Doing so, we understand from basic theory that, the first, second, third and the fifth term will yield zero. Hence, we obtain

$$
\begin{equation*}
P_{r(a v)}=\frac{1}{T} \cdot \frac{I^{2} d l^{2} \sin ^{2} \theta}{16 \pi^{2} \varepsilon} \cdot \frac{\omega^{2}}{2 r^{2} c^{3}} \cdot T=\frac{\omega^{2} I^{2} d l^{2} \sin ^{2} \theta}{32 \pi^{2} r^{2} c^{3} \varepsilon} \tag{2.36}
\end{equation*}
$$

where $T$ is the time period of oscillation of the current element and $T=\frac{2 \pi}{\omega}$. The elementary area in spherical coordinates is $2 \pi r^{2} \sin \theta d \theta$, the total power radiated becomes
power $=\oint_{\text {sufface }} P_{r(a v)} d a=\int_{0}^{\pi / 2} \frac{\eta}{2}\left(\frac{\omega I d l \sin \theta}{4 \pi r c}\right)^{2} 2 \pi r^{2} \sin \theta d \theta=\frac{\eta \omega^{2} I^{2} d l^{2}}{16 \pi c^{2}} \int_{0}^{\pi} \sin ^{3} \theta d \theta$

$$
\begin{equation*}
=\frac{\eta \omega^{2} I^{2} d l^{2}}{16 \pi c^{2}}\left[\frac{-\cos \theta}{3}\left(2+\sin ^{2} \theta\right)\right]_{0}^{\pi}=\frac{\eta \omega^{2} I^{2} d l^{2}}{12 \pi c^{2}} \tag{2.37}
\end{equation*}
$$

We note from (2.37) that, the radiated power is inversely proportional to the square of operating frequency and current. If the carrier frequency increases, the power radiated into space also increases as its square. An increase in current gives rise to stronger magnetic and electric fields as is evident from the expressions for $H_{\phi}$ and $E_{r}, E_{\theta}$. Hence, the radiation field is stronger for higher currents. In the above expression, the peak value of current has been used. The effective or the RMS current is, for a sinusoidal distribution
$I_{r m s}=\frac{I}{\sqrt{2}}$
Substitution of (2.38) in (2.37) yields
Power $=\frac{\eta \omega^{2} I_{r m s}^{2} d l^{2}}{6 \pi c^{2}}=80 \pi^{2}\left(\frac{d l}{\lambda}\right)^{2} I_{r m s}^{2}=788.768\left(\frac{d l}{\lambda}\right)^{2} I_{r m s}^{2}$
From (2.39), it appears that, the term $80 \pi^{2}\left(\frac{d l}{\lambda}\right)^{2}$ has the appearance of a resistance as power is $I^{2} R$. This resistance is defined as the radiation resistance and for a current element,

$$
\begin{equation*}
R_{r a d}=80 \pi^{2}\left(\frac{d l}{\lambda}\right)^{2} \tag{2.40}
\end{equation*}
$$

In (2.40), the dipole current is uniform. However, with no end loading, the current must be zero at the ends and if the dipole is short, the current tapers almost linearly from a maximum at the center to zero at the ends with an average value of $\frac{1}{2}$ of the maximum. We also note that, for lengths small compared to a wavelength, the radiation resistance is very small. For example, if $\frac{d l}{\lambda}=0.1$, then $R_{r a d}=0.08 \pi^{2}$. The value of the radiation resistance is an indicator of the power radiated by the antenna into space. Thus, this type of antenna is not a good radiator of EM power into space. A practical antenna such as an elementary dipole is a center-fed antenna. Its length is very short as compared to a wavelength at the carrier frequency. For the same current $I$ at the terminals the practical dipole of length $l$ radiates only one-quarter as much power as the current element of the same length which has the current $I$ throughout its length. The radiation resistance of a practical short dipole is one-quarter that of the current element of the same length. That is
$R_{\text {rad }}($ short dipole $)=20 \pi^{2}\left(\frac{l}{\lambda}\right)^{2} \approx 200\left(\frac{l}{\lambda}\right)^{2}$

### 2.3 Radiation Resistance of a Loop Antenna

The field radiated by a small magnetic dipole is the dual of that radiated by a small electric current dipole short current filament. The source point (the elementary current length) is assumed to be in the $x y$-plane having a coordinate of ( $\left.x^{\prime}, y^{\prime}, 0\right)$ and the observation point is assumed to be at $(x, y, z)$. The contribution of the current filament of strength $\operatorname{Iad} \phi$ to the total vector potential is found by the use of the equation

$$
\begin{equation*}
d A=\frac{\mu_{0} I d l}{4 \pi r} a_{z} \tag{2.42}
\end{equation*}
$$



Fig.2.4 A loop in the $x y$ - plane carrying a current $I$ that produces a field at $(x, y, z)$
(2.42) takes the form of
$d A=\frac{\mu_{0} I}{4 \pi R} \exp (-j \beta R)\left(d \rho a_{\rho}+a d \phi^{\prime} a_{\phi}\right)$
This is due to the reason that the elementary current length is in the $x y$-plane. However, we may note immediately from Fig.2.2 that, the contribution to the field due to the $\rho$ component will be canceled by a diametrically opposite current element. Therefore, (2.43) is simplified to
$d A=\frac{\mu_{0} I}{4 \pi R} \exp (-j \beta R)\left(a d \phi^{\prime} a_{\phi}\right)$
However, $a_{x}=-\sin \phi^{\prime} a_{\phi}$ and $a_{y}=\cos \phi^{\prime} a_{\phi}$

Substitution of (2.44) into (2.43) gives us
$d A=\frac{\mu_{0} I a\left(-a_{x} \sin \phi^{\prime}+a_{y} \cos \phi^{\prime}\right) d \phi^{\prime} e^{-j \beta R}}{4 \pi R}$
where $R=\left[\left(x-a \cos \phi^{\prime}\right)^{2}+\left(y-a \sin \phi^{\prime}\right)^{2}\right]^{1 / 2}$

## Method-I

Thus, $A=\frac{\mu_{0} I a}{4 \pi} \exp (-j \beta R) \int_{0}^{2 \pi} \frac{\left(-a_{x} \sin \phi^{\prime}+a_{y} \cos \phi^{\prime}\right)}{R} d \phi^{\prime}$
We are primarily interested in the far field so that $r \gg a$. It is further assumed that $a \ll \lambda$ so that the loop may be treated as a point source. In the spherical coordinate system,
$R=\left[\left(x-x^{\prime}\right)^{2}+\left(y-y^{\prime}\right)^{2}+\left(z-z^{\prime}\right)^{2}\right]$
$=\left[r^{2}+a^{2}-2 r a \sin \theta\left(\cos \phi \cos \phi^{\prime}+\sin \phi \sin \phi^{\prime}\right)\right]^{1 / 2}$
$=r\left[1+\left(\frac{a}{r}\right)^{2}-2 \frac{a}{r} \sin \theta \cos \left(\phi-\phi^{\prime}\right)\right]^{\frac{1}{2}}$
$\cong r\left[1-\frac{a}{r} \sin \theta \cos \left(\phi-\phi^{\prime}\right)+\frac{1}{2}\left(\frac{a}{r}\right)^{2}+\ldots\right]$
$=r\left[1-\frac{a}{r} \sin \theta \cos \left(\phi-\phi^{\prime}\right)\right]$
as $x=r \sin \theta \cos \phi, y=r \sin \theta \sin \phi$ and $z=r \cos \theta, x^{\prime}=a \cos \phi$ and $y^{\prime}=a \sin \phi$
(2.46) is approximated as
$R \approx r-a \sin \theta\left(\cos \phi \cos \phi^{\prime}+\sin \phi \sin \phi^{\prime}\right) \approx r-a \sin \theta \cos \left(\phi-\phi^{\prime}\right)$
as $a^{2} \ll r^{2}$
Hence, substitution of (2.49) into (2.48) gives us

$$
\begin{align*}
& A=\frac{\mu_{0} I a}{4 \pi r} e^{-j \beta r} \int_{0}^{2 \pi} e^{j \beta a \sin \theta \cos \left(\phi-\phi^{\prime}\right)}\left(-a_{x} \sin \phi^{\prime}+a_{y} \cos \phi^{\prime}\right) d \phi^{\prime}  \tag{2.48}\\
& A=\frac{\mu_{0} I a}{4 \pi r} e^{-j \beta r} \int_{0}^{2 \pi} e^{j \beta a \sin \theta \cos \left(\phi-\phi^{\prime}\right)} \cos \left(\phi-\phi^{\prime}\right) a_{\phi} d \phi^{\prime} \tag{2.49}
\end{align*}
$$

We further make an assumption that $\beta a \ll 1$. Hence, the exponential term may be expanded in a series and the higher orders in the expansion may be neglected. We obtain,

$$
\begin{align*}
& A=\frac{\mu_{0} I a}{4 \pi r} e^{-j \beta r} \int_{0}^{2 \pi}\left(1+j \beta a \sin \theta \cos \left(\phi-\phi^{\prime}\right)\right) \cos \left(\phi-\phi^{\prime}\right) a_{\phi} d \phi^{\prime}  \tag{2.50}\\
& =\frac{\mu_{0} I a}{4 \pi r} e^{-j \beta r} \int_{0}^{2 \pi} \cos \left(\phi-\phi^{\prime}\right)+j \beta a \sin \theta \cos ^{2}\left(\phi-\phi^{\prime}\right) a_{\phi} d \phi^{\prime}
\end{align*}
$$

We note that, the integration of the first term in (2.50) gives us zero and that of the second term $j \beta \pi a^{2} \sin \theta$. Hence, we obtain, the $\phi$ component of the vector potential as $A=j \beta \pi \frac{\mu_{0} I a^{2}}{4 \pi R} e^{-j \beta r} \sin \theta a_{\phi}$
The quantity $\pi a^{2} I$ is the product of the area of the circular loop of radius $a$ and the current flowing through is and is known as the dipole moment of the small circular loop.

## Method-II

The same expression could have been derived by another way in the following manner. As before, under the assumption of $\beta a \ll 1$ and using the series expansion of the exponential term, we obtain,
$A=\frac{\mu_{0} I a}{4 \pi R} e^{-j \beta r} \int_{0}^{2 \pi}\left(1+j \beta a \sin \theta \cos \left(\phi-\phi^{\prime}\right)\right)\left(-a_{x} \sin \phi^{\prime}+a_{y} \cos \phi^{\prime}\right) d \phi^{\prime}$
The integration of the first two terms in the above expression gives us zero.
Using $\cos \left(\phi-\phi^{\prime}\right)=\cos \phi \cos \phi^{\prime}+\sin \phi \sin \phi^{\prime}$ and using in the above expression for the second term, we obtain,
$\int_{0}^{2 \pi}\left(\cos \phi \cos \phi^{\prime}+\sin \phi \sin \phi\right)\left(-a_{x} \sin \phi^{\prime}+a_{y} \cos \phi^{\prime}\right) d \phi^{\prime}=\pi\left(\cos \phi a_{y}-\sin \phi a_{x}\right)=\pi a_{\phi}$
Hence, $A=\frac{\mu_{0} I a}{4 \pi R} e^{-j \beta r} j \beta a \sin \theta \pi a_{\phi}=j \beta \frac{\mu_{0} I \pi a^{2}}{4 \pi R} e^{-j \beta r} \sin \theta a_{\phi}$
We also write, $R \approx r-a \sin \theta \cos \left(\phi-\phi^{\prime}\right)$
Having obtained the vector potential, we can obtain the magnetic field and the corresponding electric field.
$H=\nabla \times A_{\phi}=-\frac{1}{\mu_{0} r} \frac{\partial}{\partial r}\left(r A_{\phi}\right) a_{\phi}=-\frac{\pi a^{2} I \beta^{2}}{4 \pi r} e^{-j \beta r} \sin \theta a_{\phi}=-\frac{M \beta^{2}}{4 \pi r} e^{-j \beta r} \sin \theta a_{\phi}$
The corresponding electric field is given as
$E=-\eta a_{r} \times H_{\phi}=\frac{\eta M \beta^{2}}{4 \pi r} e^{-j \beta a} a_{\theta}$
The radiated power is
$P_{r}=\frac{1}{2}\left(E \times H^{*}\right)=\frac{1}{2} \operatorname{Re} \int_{0}^{2 \pi} \int_{0}^{\pi} E_{\theta} H_{\phi} r^{2} \sin \theta d \theta d \phi=\frac{\eta M^{2} \beta^{4}}{16 \pi^{2}} \int_{0}^{2 \pi} \int_{0}^{\pi} \sin ^{3} \theta d \theta d \phi=\frac{\eta M^{2} \beta^{4}}{12 \pi}$
The radiation resistance of the loop may be found by equating $\frac{1}{2}|I|^{2} R_{a}$ to $P_{r}$. So, we get $\frac{1}{2}|I|^{2} R_{a}=\frac{\eta M^{2} \beta^{4}}{12 \pi}=\frac{120 \pi\left(\pi a^{2} I\right)^{2}}{12 \pi} \beta^{4}=10\left(\pi a^{2}\right)\left(\frac{2 \pi}{\lambda}\right)^{4} I^{2}$
Hence, $R_{a}=20 \pi^{6}\left(\frac{a}{\lambda}\right)^{4}$
Let us consider a loop with a radius of 1 cm operating at a frequency of 1 MHz , and substitution of these values into the above expression gives us a radiation resistance of
$R_{a} \approx 3.8 \times 10^{-9} \Omega$. So a small loop antenna is a very poor radiator. If $N$ turns of wire are used, the radiation resistance is increased by a factor of $N^{2}$. Small loop antennas are often used as receiving antennas for portable radios. Although they are very inefficient, they do give an acceptable performance because of the large available signal level. At low frequencies atmospheric noise is often the limiting factor, so a more efficient antenna does not necessarily give better reception. Of course, a small loop antenna would not be used for transmitting purposes unless very short distances were involved and the poor gain could be tolerated. The gain of a small loop antenna is very low because the ohmic resistance of the wire is generally much greater than the radiation resistance.

Ex. 2.1: Two $z$ directed Hertzian dipoles are in phase and a distance of $d$ apart. The electric field intensity is given by

$$
E_{\theta}=\frac{j \beta \eta(I d l)}{2 \pi r} e^{-j \beta r} \sin \theta \cos \left(\beta \frac{d}{2} \sin \theta \sin \phi\right)
$$

In deriving this expression, use has been made of
$y=r \cos \alpha=r \sin \theta \sin \phi$
The corresponding magnetic field strength is given as

$$
H_{\phi}=\frac{j \eta(I d l)}{2 \pi r} e^{-j \beta r} \sin \theta \cos \left(\beta \frac{d}{2} \sin \theta \sin \phi\right)
$$

Thus, the power radiated becomes

$$
P_{\text {avg }}=\frac{1}{2} \operatorname{Re}\left(E \times H^{*}\right)=\frac{1}{2}\left(\frac{\beta \eta I d l}{2 \pi r} \cdot \frac{\beta I d l}{2 \pi r}\right) \sin \theta \cdot \sin \theta
$$

The radiated power is obtained by taking the integral of the above over an elementary surface.

$$
P_{r a d}=\int_{0}^{2 \pi \pi} \int_{0}^{1} \frac{1}{2}\left(\frac{\beta \eta I d l}{2 \pi r} \cdot \frac{\beta I d l}{2 \pi r}\right) \sin ^{2} \theta r^{2} \sin \theta d \theta d \phi=\frac{\beta^{2} \eta}{8 \pi^{2}} \cdot(I d l)^{2} \int_{0}^{2 \pi \pi} \int_{0}^{2} \sin ^{3} \theta d \theta d \phi
$$

Making use of the integral

$$
\begin{aligned}
& \int \sin ^{n} a x d x=-\frac{\sin a x \cos a x}{n a}+\frac{n-1}{n} \int \sin ^{n-2} a x d x, \text { we note that } \\
& \int_{0}^{\pi} \sin ^{3} \theta d \theta=\left|-\frac{\sin ^{2} \theta \cos \theta}{3}\right|_{0}^{\pi}+\frac{2}{3} \int_{0}^{\pi} \sin \theta d \theta=\left|-\frac{\sin ^{2} \theta \cos \theta}{3}\right|_{0}^{\pi}-\frac{2}{3}|\cos \theta|_{0}^{\pi}=\frac{4}{3} \\
& P_{r a d}=\frac{\beta^{2} \eta}{8 \pi^{2}} \cdot(I d l)^{2} \cdot \frac{4}{3} \int_{0}^{2 \pi} d \phi=\frac{\beta^{2} \eta}{8 \pi^{2}} \cdot(I d l)^{2} \cdot \frac{4}{3} \cdot 2 \pi=\frac{\beta^{2} \eta}{3 \pi}(I d l)^{2}
\end{aligned}
$$

We make use of $\beta=\frac{2 \pi}{\lambda}$ in the above to get

$$
P_{\mathrm{rad}}=\left(\frac{2 \pi}{\lambda}\right)^{2} \cdot \frac{\eta(I d l)^{2}}{3 \pi}=\frac{4 \pi \eta}{3}\left(\frac{I d l}{\lambda}\right)^{2}
$$

### 2.4 DIRECTIVITY

The ability of an antenna or an array of antennas to concentrate the radiated power in a given direction, or conversely to absorb effectively incident power from that direction is specified variously in terms of its gain, power gain, directive gain or directivity.
The magnetic field is obtained from the magnetic vector potential by taking its curl. In spherical coordinates, it is expressed as

$$
\begin{align*}
& \nabla \times A=\frac{1}{r^{2} \sin \theta}\left[\begin{array}{ccc}
a_{r} & r a_{\theta} & r \sin \theta a_{\phi} \\
\frac{\partial}{\partial r} & \frac{\partial}{\partial \theta} & \frac{\partial}{\partial \phi} \\
A_{r} & r A_{\theta} & r \sin \theta A_{\phi}
\end{array}\right] \\
& =\frac{1}{r^{2} \sin \theta}\left[\left(r \sin \theta \frac{\partial A_{\phi}}{\partial \theta}-r \frac{\partial A_{\theta}}{\partial \phi}\right) a_{r}-r\left(r \sin \theta \frac{\partial A_{\phi}}{\partial r}-\frac{\partial A_{r}}{\partial \phi}\right) a_{\theta}+r \sin \theta\left(\frac{r \partial A_{\theta}}{\partial r}-\frac{\partial A_{r}}{\partial \theta}\right) a_{\phi}\right] \tag{2.60}
\end{align*}
$$

Under the assumption of the wave propagating along the radial direction from the current source, we note that the electric and the magnetic fields would lie in the $\theta-\phi$ plane. Therefore, (2.60) reduces to
$\nabla \times A=\frac{1}{r^{2} \sin \theta}\left[r \sin \theta\left(\frac{r \partial A_{\theta}}{\partial r}-\frac{\partial A_{r}}{\partial \theta}\right) a_{\phi}-r\left(r \sin \theta \frac{\partial A_{\phi}}{\partial r}-\frac{\partial A_{r}}{\partial \phi}\right) a_{\theta}\right]$
$=\frac{1}{r}\left(\frac{r \partial A_{\theta}}{\partial r}-\frac{\partial A_{r}}{\partial \theta}\right) a_{\phi}-\left(\frac{\partial A_{\phi}}{\partial r}-\frac{1}{r \sin \theta} \frac{\partial A_{r}}{\partial \phi}\right) a_{\theta}$
$=\frac{\partial A_{\theta}}{\partial r} a_{\phi}-\frac{\partial A_{\phi}}{\partial r} a_{\theta}$
This is due to the fact that the radial component $A_{r}$ does not exist or $A_{r}=0$. As the magnetic vector potential has components constrained to lie in the $\theta-\phi$ plane, we can also express (2.61) as the following.
Let us define a vector potential
$A_{T}=A_{\theta}(r, \theta, \phi) a_{\theta}+A_{\phi}(r, \theta, \phi) a_{\phi}$
Using the fact that $a_{\theta} \times a_{r}=a_{\phi}, a_{r} \times a_{\phi}=a_{\theta}$ and $a_{\phi} \times a_{\theta}=a_{r}$
We may observe that,
$a_{r} \times A_{T}=a_{r} \times\left\lfloor A_{\theta}(r, \theta, \phi) a_{\theta}+A_{\phi}(r, \theta, \phi) a_{\phi}\right\rfloor$
$=-A_{\theta}(r, \theta, \phi) a_{\phi}+A_{\phi}(r, \theta, \phi) a_{\theta}$
$\nabla \times A=\frac{\partial A_{\theta}}{\partial r} a_{\phi}-\frac{\partial A_{\phi}}{\partial r} a_{\theta}$
$=-a_{r} \times A_{T}$
Let us consider very generic expressions for the two components of the magnetic field vector as the following:

$$
\begin{equation*}
A_{\theta}(r, \theta, \phi)=\frac{1}{r} A_{\theta 0}(\theta, \phi) \exp (-j \beta r) \tag{2.66a}
\end{equation*}
$$

And similarly,

$$
\begin{equation*}
A_{\phi}(r, \theta, \phi)=\frac{1}{r} A_{\phi 0}(\theta, \phi) \exp (-j \beta r) \tag{2.66b}
\end{equation*}
$$

Therefore, the first term of (2.61) becomes
$r \frac{\partial A_{\theta}}{\partial r}=r\left[-\frac{1}{r^{2}} A_{\theta 0}(\theta, \phi) \exp (-j \beta r)-j \frac{\beta}{r} A_{\theta 0}(\theta, \phi) \exp (-j \beta r)\right]$
$=-\left[A_{\theta 0}(\theta, \phi) \exp (-j \beta r)\right]\left(\frac{1}{r}+j \beta\right)$
Similarly,

$$
\begin{align*}
& \frac{\partial A_{\phi}}{\partial r}=-\frac{1}{r^{2}} A_{\phi 0}(\theta, \phi) \exp (-j \beta r)-j \frac{\beta}{r} A_{\phi 0}(\theta, \phi) \exp (-j \beta r)  \tag{2.67b}\\
& =-\frac{1}{r}\left[A_{\phi 0}(\theta, \phi) \exp (-j \beta r)\right]\left(\frac{1}{r}+j \beta\right)
\end{align*}
$$

The radiated field components correspond to the terms containing $\frac{1}{r}$ only. Hence, we retain the term $-j \beta\left[A_{\theta 0}(\theta, \phi) \exp (-j \beta r)\right]$ from (2.67a) and $-\frac{j \beta}{r}\left[A_{\phi 0}(\theta, \phi) \exp (-j \beta r)\right]$ from (2.67b). The total contribution to the magnetic vector potential is,
$\nabla \times A=-\frac{j \beta}{r}\left[A_{\theta 0}(\theta, \phi) \exp (-j \beta r) a_{\phi}-A_{\phi 0}(\theta, \phi) \exp (-j \beta r) a_{\theta}\right]$
$=-j \beta\left[A_{\theta}(r, \theta, \phi) a_{\phi}-A_{\phi}(r, \theta, \phi) a_{\theta}\right]$
$=j \beta\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]=B$
We have made use of (2.67) in deriving (2.68). We, therefore, obtain the magnetic field as
$H=\frac{B}{\mu}=\frac{1}{\mu} \nabla \times A=j \frac{\beta}{\mu}\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]$
$=j \frac{2 \pi}{\lambda \mu}\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]=j \frac{2 \pi}{(c / f) \mu}\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]$
$=-\frac{j \omega}{c \mu} a_{r} \times A_{T}$
where we have made use of (2.65).
Further, $c \mu=\frac{\mu}{\sqrt{\mu \varepsilon}}=\sqrt{\frac{\mu}{\varepsilon}}=\frac{1}{\eta}$
Hence, the magnetic field can also be expressed as

$$
\begin{equation*}
H=-\frac{j \omega}{c \mu} a_{r} \times A_{T}=-j \frac{\omega}{\eta} a_{r} \times A_{T} \tag{2.71}
\end{equation*}
$$

The electric field associated with (2.71) is evaluated as
$E=\frac{1}{j \omega \varepsilon}(\nabla \times H)$
$=\frac{1}{j \omega \varepsilon}\left\{\nabla \times \frac{j \omega}{c \mu}\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]\right\}$
$=c .\left\{\nabla \times\left[A_{\phi}(r, \theta, \phi) a_{\theta}-A_{\theta}(r, \theta, \phi) a_{\phi}\right]\right\}$
To evaluate (2.72) we note that, a similar expression appears in (2.61). Here, the radial component is nonexistent. Hence,

$$
\begin{align*}
& \nabla \times H=\frac{1}{r^{2} \sin \theta}\left[\begin{array}{ccc}
a_{r} & r a_{\theta} & r \sin \theta a_{\phi} \\
\frac{\partial}{\partial r} & \frac{\partial}{\partial \theta} & \frac{\partial}{\partial \phi} \\
0 & r A_{\phi} & -r \sin \theta A_{\theta}
\end{array}\right] \\
& =\frac{1}{r^{2} \sin \theta}\left[\left(-r \sin \theta \frac{\partial A_{\phi}}{\partial \theta}-\frac{\partial}{\partial \phi} r A_{\theta}\right) a_{r}-r\left(-r \sin \theta \frac{\partial A_{\theta}}{\partial r}\right) a_{\theta}+r \sin \theta\left(\frac{r \partial A_{\phi}}{\partial r}\right) a_{\phi}\right]  \tag{2.73}\\
& =\frac{1}{r^{2} \sin \theta}\left[r^{2} \sin \theta\left(\frac{\partial A_{\theta}}{\partial r}\right) a_{\theta}+r^{2} \sin \theta \frac{\partial A_{\phi}}{\partial r} a_{\phi}\right] \\
& =\frac{\partial A_{\theta}}{\partial r} a_{\theta}+\frac{\partial A_{\phi}}{\partial r} a_{\phi}
\end{align*}
$$

We may note that,

$$
\begin{align*}
& \frac{\partial A_{\theta}}{\partial r}=-\frac{1}{r^{2}} A_{\theta 0}(r, \theta, \phi) \exp (-j \beta r)-\frac{j \beta}{r} A_{\theta 0}(r, \theta, \phi) \exp (-j \beta r) \\
& =-\frac{1}{r} A_{\theta 0}(r, \theta, \phi) \exp (-j \beta r)\left(\frac{1}{r}+j \beta\right) \tag{2.74}
\end{align*}
$$

A similar expression may be obtained for $\frac{\partial A_{\phi}}{\partial r}$. Combining these two together, we have an expression for the radiated electric field as

$$
\begin{align*}
& E=c\left[-\frac{j \beta}{r} A_{\theta 0}(r, \theta, \phi) \exp (-j \beta r) a_{\theta}-\frac{j \beta}{r} A_{\phi 0}(r, \theta, \phi) \exp (-j \beta r) a_{\phi}\right]  \tag{2.75}\\
& =-j \beta c\left[A_{\theta}(r, \theta, \phi) a_{\theta}+A_{\phi}(r, \theta, \phi) a_{\phi}\right]=-j \omega A_{T}
\end{align*}
$$

The generic expressions for the radiated electric field and the magnetic field are given by (2.75) and (2.71) respectively when the wave propagates along a radial direction from the source. The power radiated per unit area in any direction is given by the Poynting vector $P$. For the distant or radiation field for which the electric field and the magnetic field are orthogonal in a plane perpendicular to the radius vector, and for which $E=\eta_{v} H$, the power flow per unit area is given by

$$
\begin{equation*}
P=\frac{E^{2}}{\eta_{v}}=\eta_{v} H=E \times H \tag{2.76}
\end{equation*}
$$

The radiation intensity $\Phi(\theta, \phi)$ in a given direction is defined as the power per unit solid angle in that direction. This takes the form of
$\Phi(\theta, \phi)=r^{2} P=\frac{r^{2} E^{2}}{\eta_{v}}$ watts/unit solid angle
The radiation intensity is independent of the radial distance. The total power radiated is $W_{r}=\int \Phi d \Omega$ watts
Since there are $4 \pi$ sertaradians in the total solid angle, the average power radiated per unit solid angle is $\frac{W_{t}}{4 \pi}$.
The directive gain or the directivity is defined as the ratio of the radiation intensity in that direction to the average radiated power. That is,
$G(\theta, \phi)=\frac{\Phi(\theta, \phi)}{\Phi_{a v}}=\frac{\Phi(\theta, \phi)}{W_{t} / 4 \pi}=\frac{4 \pi \Phi(\theta, \phi)}{W_{t}}=\frac{4 \pi \Phi(\theta, \phi)}{\int \Phi d \Omega}$
For a current element $I d l$, the distant field in the direction of maximum radiation is, from (2.32)
$E=\frac{60 \pi}{r} I\left(\frac{d l}{\lambda}\right)$
The current required to radiate 1 watt is, therefore, from (2.39)
$I=\frac{\lambda}{\sqrt{80} \pi d l} \mathrm{~A}$
Substitution of (2.81) in (2.80) gives us a corresponding field strength in the direction of maximum radiation as
$E=\frac{60 \pi}{r} \cdot \frac{\lambda}{\sqrt{80} \pi d l} \cdot \frac{d l}{\lambda}=\frac{60 \pi}{r \sqrt{80}} \mathrm{~V} / \mathrm{m}$
The radiation intensity is
$\Phi=\frac{r^{2} E^{2}}{\eta}=\frac{60^{2}}{80 \times 120 \pi}=\frac{3}{8 \pi}$
so that the directivity or maximum gain of the current element is
$g_{d}(\max )=4 \pi \Phi=1.5$
The above expression states that for a current element the computed directivity is 1.5 . Next, let us evaluate the directivity of a half wave dipole by considering its radiated field expressions. The Poynting vector power density for the half-wave dipole is maximum at $\theta=\frac{\pi}{2}$ and hence the effective or the RMS value of the electric field is
$E_{\theta}(r m s)=\frac{60 I_{r m s}}{r}$
The radiated power is
$P_{r}=\frac{E_{\theta}(r m s)^{2}}{\eta}=\frac{1}{120 \pi}\left(\frac{60 I_{r m s}}{r}\right)^{2}$
$=\frac{30 I_{r m s}^{2}}{\pi r^{2}}$
The power input to the dipole is $R_{r a d} I_{r m s}^{2}=73 I_{r m s}^{2}$

From (2.87) we observe that, the radiation resistance of a half wave dipole is $73 \Omega$.
Hence, the directive gain of the half wave dipole is
$G=\frac{4 \pi r^{2} \cdot \frac{30 I_{r m s}^{2}}{\pi r^{2}}}{73 I_{r m s}^{2}}=\frac{120}{73}=1.64$
We can also evaluate the directivity of a half wave dipole in the following manner. For a half-wave dipole, the maximum field strength is
$E=\frac{60 I_{m}}{r} \mathrm{~V} / \mathrm{m}$
The effective or the RMS value of the electric field is, therefore
$E_{r m s}=\frac{60 I_{r m s}}{r}$
The current required to radiate 1 watt is $\frac{1}{\sqrt{73}} \mathrm{amps}$. This can be verified from (2.87).
The corresponding field strength in the direction of maximum radiation is obtained by substituting this value in (2.90) and this is, therefore,
$E=\frac{60}{r} \cdot \sqrt{\frac{1}{73}}$
The radiation intensity is
$\Phi=\frac{r^{2} E^{2}}{\eta}=\frac{r^{2} .60^{2} \frac{1}{73}}{r^{2} \eta}=60^{2} \frac{1}{73 \times 120 \pi}$
Hence its directivity is
$G(\max )=4 \pi \Phi=60^{2} \frac{1}{73 \times 120 \pi} .4 \pi=\frac{60 \times 60}{73 \times 30}=\frac{120}{73}=1.64$
We next relate the electric field strength and the effective radiated power. As the power radiated is proportional to the square of the RMS value of the current,
Thus, $I_{r m s}=\sqrt{\frac{W}{36.5}}$
Substitution of the above value of the RMS current in the field strength expression gives us
$E=\frac{60}{r} \cdot \sqrt{\frac{W}{36.5}}=\frac{9.9313}{r} \sqrt{W} \mathrm{~V} / \mathrm{m}$
One mile $=1609.344$ meters $=1.609344 \mathrm{Km}$
The above RMS field strength for a distance of 1 mile is
$\frac{9.9313}{r \times 1609.344} \sqrt{W}=\frac{6.14}{r} \sqrt{W} \mathrm{mv} / \mathrm{m}$
Next we set up the vector potential due to a traveling wave current distribution in the $z$ direction given by $I(z)=I_{m} e^{j \beta z}$

## MODULE-III

### 3.1 LINEAR ANTENNA

A linear antenna is assumed to be made up of a large number of very short conductors connected in series. A conductor having a length of $\frac{\lambda}{2}$ is shown in Fig.1. The two ends of the conductor are at opposite voltages. The RF energy from the transmitter is fed at its center because the dipole antenna is a symmetrical antenna in which the two ends are at equal potentials relative to the mid-point. At the open-ends, the current is zero and the voltage is maximum. The radiation pattern represents the field strength in various directions of an open ended half-wave antenna. For the opened out half wave conductor, the magnetic field will be maximum along a line extending from the its center and electric fields will be perpendicular to it.

The half-wave dipole (Fig to be inserted) derives its name from the fact that its length is half a wavelength $l=\frac{\lambda}{2}$. It consists of a thin wire fed or excited at its midpoint by a voltage source connected to the antenna through a transmission line. The field due to the dipole may be considered to be consisting of a chain of Hertzian dipoles. The magnetic vector potential at a point P due to a differential length $d l=d z$ of the dipole carrying a phasor current $I_{s}=I_{0} \cos \beta z$ is
$d A=\frac{\mu I_{0} \cos \beta z}{4 \pi r^{\prime}} e^{-j \beta r r^{\prime}}$
A sinusoidal current distribution has been assumed here for two reasons. First it is due to the transmission line model of the dipole. Second, the current must be zero at the ends of the dipole. Thus one may consider a triangular distribution of the current also. However, it is likely to yield a less accurate result. We write
$r=r^{\prime}-z \cos \theta$
or $r^{\prime}=r-z \cos \theta$


Fig.3.1 An electric dipole
The difference between $\beta r$ and $\beta r^{\prime}$ is quite significant. Hence, the above magnetic vector potential is written as
$d A=\frac{\mu I_{0} \cos \beta z}{4 \pi r} e^{-j \beta(r-z \cos \theta)}$
The total magnetic vector potential becomes
$A=\frac{\mu I_{0} e^{-j \beta r}}{4 \pi r} \int_{-\frac{\lambda}{4}}^{\frac{\lambda}{4}} e^{j \beta z \cos \theta} \cos \beta z d z$
We understand that
$\int e^{a z} \cos b z d z=\frac{e^{a z}(a \cos b z+b \sin b z)}{a^{2}+b^{2}}+c$
Substitution of (3.5) in the total magnetic vector potential given as in (3.4) gives

$$
\begin{equation*}
A_{z}=\left.\frac{\mu I_{0} e^{-j \beta r} e^{j \beta z} \cos \theta}{4 \pi r} \frac{(j \beta \cos \theta+\beta \sin b z)}{-\beta^{2} \cos ^{2} \theta+\beta^{2}}\right|_{-\frac{\lambda}{4}} ^{\frac{\lambda}{4}} \tag{3.6}
\end{equation*}
$$

Further, since $\beta=\frac{2 \pi}{\lambda}$ or $\frac{\beta \lambda}{4}=\frac{\pi}{2}$ and $1-\cos ^{2} \theta=\sin ^{2} \theta$, then

$$
\begin{align*}
A_{z} & =\frac{\mu I_{0} e^{-j \beta r}}{4 \pi \beta^{2} \sin ^{2} \theta}\left[e^{j\left(\frac{\pi}{2}\right) \cos \theta}(0+\beta)-e^{-j\left(\frac{\pi}{2}\right) \cos \theta}(0-\beta)\right] \\
& =\frac{\mu I_{0} e^{-j \beta r} \cos \left(\frac{\pi}{2} \cos \theta\right)}{2 \pi r \beta \sin ^{2} \theta} \tag{3.7}
\end{align*}
$$

In this section, we will derive a general expression for the distant electric field of a dipole antenna of any half-length $H$. The total vector potential at point P is

$$
\begin{equation*}
A_{z}=\frac{\mu}{4 \pi} \int_{-H}^{0} \frac{I_{m} \sin \beta(H+z) e^{-j \beta r}}{R} d z+\frac{\mu}{4 \pi} \int_{0}^{H} \frac{I_{m} \sin \beta(H-z) e^{-j \beta r}}{R} d z \tag{3.8}
\end{equation*}
$$

Because of the approximation $R \approx r$, we write (2.104) as

$$
\begin{equation*}
A_{z}=\frac{\mu I_{m}}{4 \pi}\left[\int_{-H}^{0} \frac{\sin \beta(H+z) e^{-j \beta z \cos \theta}}{r} d z+\int_{0}^{H} \frac{\sin \beta(H-z) e^{-j \beta z \cos \theta}}{r} d z\right] \tag{3.9}
\end{equation*}
$$

We make use of the following integral to evaluate the above.
$\int e^{a x} \sin b x d x=\frac{e^{a x}}{a^{2}+b^{2}}(a \sin b x-b \cos b x)$
In the first term in the RHS of (3.9), it is noted that, $a=j \beta \cos \theta$ and $b=\beta$
Thus, with this the first term in the RHS of (3.9) becomes.

$$
\begin{aligned}
& \frac{\mu I_{m}}{4 \pi} \int_{-H}^{0} \frac{\sin \beta(H+z) e^{-j \beta z \cos \theta}}{r} d z=\frac{\mu I_{m}}{4 \pi r}\left|\frac{e^{j \beta \cos \theta}}{\beta^{2}\left(1-\cos ^{2} \theta\right)}[j \beta \cos \theta \sin \beta(H+z)-\beta \cos \beta(H+z)]\right|_{-H}^{0} \\
& =\frac{\mu I_{m}}{4 \pi r \beta^{2} \sin ^{2} \theta}\left[j \beta \cos \theta \sin \beta H-\beta \cos \beta H-e^{-j \beta H \cos \theta}(-\beta)\right]
\end{aligned}
$$

Similarly, the second term in the RHS of (3.9) becomes
$a=j \beta \cos \theta$ and $b=-\beta$
$\frac{\mu I_{m}}{4 \pi} \int_{0}^{H} \frac{\sin \beta(H-z) e^{-j \beta z \cos \theta}}{r} d z=\frac{\mu I_{m}}{4 \pi} \left\lvert\, \frac{e^{j \beta \cos \theta}}{\beta^{2}\left(1-\cos ^{2} \theta\right)}[j \beta \cos \theta \sin \beta(H-z)+\beta \cos \beta(H-z)]_{0}^{H}\right.$
which is equal to

$$
\frac{\mu I_{m}}{4 \pi r \beta^{2} \sin ^{2} \theta}\left[-j \beta \cos \theta \sin \beta H-\beta \cos \beta H-e^{-j \beta H \cos \theta}(-\beta)\right]
$$

Combining the two, we obtain

$$
\begin{equation*}
A_{z}=\frac{\mu I_{m}}{4 \pi r \beta^{2} \sin ^{2} \theta}[2 \beta \cos (\beta H \cos \theta)-2 \beta \cos \beta H]=\frac{\mu I_{m} e^{-j \beta r}}{2 \pi r \beta \sin ^{2} \theta}(\cos (\beta H \cos \theta)-\cos \beta H) \tag{3.10}
\end{equation*}
$$

We recall that when the current is entirely in the $z$ direction,
$\mu H_{\phi}=-\sin \theta \frac{\partial A_{z}}{\partial r}$
Differentiation of (3.10) w.r.t. the radial distance $r$ gives us the expression for the magnetic field strength at a distant point as

$$
\begin{equation*}
H_{\phi}=\frac{j I_{m} e^{-j \beta r}(\cos (\beta H \cos \theta)-\cos \beta H)}{2 \pi r \sin \theta} \tag{3.11}
\end{equation*}
$$

where only the inverse distance term has been retained. The electric field strength corresponding to the radiation field will be

$$
\begin{equation*}
E_{\theta}=\frac{j 60 I_{m} e^{-j \beta r}(\cos (\beta H \cos \theta)-\cos \beta H)}{r \sin \theta} \tag{3.12}
\end{equation*}
$$

From this, we note that, the electric field and the magnetic field are in time phase. The power density is expressed as

$$
\begin{equation*}
P=E_{\theta} \times H_{\phi}=\frac{30 I_{m}^{2}}{\pi r^{2}}\left[\frac{\cos (\beta H \cos \theta)-\cos \beta H}{\sin \theta}\right]^{2} \tag{3.13}
\end{equation*}
$$

A resonant antenna corresponds to a resonant transmission line and the dipole antenna is a resonant antenna. Such an antenna may be viewed as an open-circuited transmission line, open-circuited at the far end of the resonant length, i.e. a multiple of quarterwavelength so that the length of the antenna is a multiple of half-wavelengths. The radiation pattern is a line drawn to join points in space which have equal field intensity due to the source.

### 3.2 NONRESONANT ANTENNA

A nonresonant antenna is like a nonresonant transmission line on which there are no standing waves. These are suppressed by the use of a correct termination to ensure that no power is reflected thus ensuring the presence of a traveling wave only. In a correctly matched transmission line, all the transmitted power is dissipated in the terminating resistance. When an antenna is terminated similar to a transmission line, about two-thirds of the input power is radiated, the remaining one third is dissipated in the antenna and none is reflected to the input. An example is a Rhombic antenna that is used for point-topoint working in the HF range spanning a frequency range of 3 -to $30-\mathrm{MHz}$. It is a broadband antenna. For the terminated antennas having only a traveling wave distribution of current, we derive next the pattern of these next.

For a distant point P located at a distance of $R$ from the origin, we note that, the elementary vector potential is expressed as
$d A_{z}=\frac{\mu I_{m}}{4 \pi R} e^{-j \beta r R} d z$
$R=r-z \cos \theta$
For the distant field calculation,
$R \approx r$
Hence, $d A_{z}=\frac{\mu I_{m} e^{-j \beta z}}{4 \pi r} e^{-j \beta(r-z \cos \theta)} d z$
Thus, $A_{z}=\int_{0}^{L} d A_{z}=\frac{\mu I_{m} e^{-j \beta r}}{4 \pi r} \int_{0}^{L} e^{-j \beta z(1-\cos \theta)} d z=\frac{\mu I_{m} e^{-j \beta r}}{4 \pi r} \cdot-\frac{1}{j \beta(1-\cos \theta)}\left|e^{-j \beta z(1-\cos \theta)}\right|_{0}^{L}$
Evaluating the integral, we obtain

$$
A_{z}=\frac{\mu I_{m} e^{-j \beta r}}{4 \pi r} \cdot-\frac{1}{j \beta(1-\cos \theta)}\left[-1+e^{j \beta(1-\cos \theta) L}\right]=\frac{\mu I_{m} e^{-j \beta r}}{4 \pi r} \cdot \frac{1}{j \beta(1-\cos \theta)}\left[1-e^{j \beta(1-\cos \theta) L}\right]
$$

We recall that when the current is entirely in the $z$ direction,
$\mu H_{\phi}=-\sin \theta \frac{\partial A_{z}}{\partial r}$
Hence, the magnetic field strength in the perpendicular direction is
$H_{\phi}=-\sin \theta \frac{I_{m} e^{-j \beta r}}{4 \pi r} \cdot \frac{1}{j \beta(1-\cos \theta)}\left[1-e^{j \beta(1-\cos \theta) L}\right]=-\sin \theta \cdot-j \beta \cdot \frac{I_{m}}{4 \pi r \cdot j \beta(1-\cos \theta)}\left[1-e^{j \beta(1-\cos \theta) L}\right]$
This becomes, after simplifying,

$$
H_{\phi}=\frac{I_{m} \sin \theta \cdot e^{-j \beta r}}{4 \pi r(1-\cos \theta)}\left[1-e^{-j \beta L(1-\cos \theta)}\right]
$$

The electric field intensity is expressed as

$$
E_{\theta}=120 \pi H_{\phi}=\frac{30 I_{m} \cdot e^{-j \beta r}}{4 \pi r(1-\cos \theta)}[1-\cos \beta L(1-\cos \theta)+j \sin \beta L(1-\cos \theta)]
$$

Thus, the magnitude of the electric field is
$E_{\theta}=\frac{30 I_{m}}{r(1-\cos \theta)}[2-2 \cos \beta L(1-\cos \theta)]^{\frac{1}{2}}$
It is observed from the above expression that with a traveling wave the pattern is no longer symmetrical about the $\theta=\frac{\pi}{2}$ degrees plane, but instead the radiation tends to lean in the direction of the current wave. The angle $\theta$ between the axis of the antenna and the direction of maximum radiation becomes smaller as the antenna and the direction of maximum radiation becomes smaller as the antenna becomes longer. For the standing wave current distribution, the pattern is always symmetrical about the $\theta=\frac{\pi}{2}$ degrees plane.

### 3.3 LOOP AND FERRITE ROD RECEIVING ANTENNAS

The loop antenna is made up of one or more turns of wire on a frame, which may be rectangular or circular and is very much smaller than one wavelength across. The antenna is popular for two reasons: (1) it is relatively compact, lending itself to use with portable receivers; and (2) it is quite directive, lending itself to use with direction-finding equipment. A loop antenna made of several turns of wire around a rectangular frame were popular for earlier model broadcast receivers, with the loop being mounted in the back of the cabinet. Recently, these have been replaced by ferrite-rod antennas. When the loop is aligned for maximum signal strength, the magnetic flux linkages are $B A N$ where $B$ is the rms magnetic flux density in Tesla, $A$ is the physical loop area in square meter and $N$ is the number of turns. The induced emf is given by
$V_{S}=\omega N B A$
When the loop is tuned by means of an external capacitor to the received frequency, the voltage at the capacitor terminals is magnified by the quality factor $Q$ of the circuit. Hence, the capacitor voltage becomes
$V_{\text {max }}=V_{S} Q=Q \omega N B A$
Since the loop is usually much smaller than the received wavelength, the induced voltage may be quite small. It may be increased by increasing any one of the factors as shown above. The $Q$ is determined by the desired selectivity. The area must be kept small; increasing the number of turns increases the coil inductance and changes $Q$, and even changing the flux density affects the $Q$. However, changing the flux density by using a
magnetic core can be achieved with a minimal change of $Q$ using ferrite cores. This has been preferred now. The loop antenna is usually used as a direction finding device.

The ferrite-rod antenna is made by winding a coil of wire on a ferrite rod similar to the one shown in Fig. Ferrites are materials that exhibit the properties of ferromagnetism. The materials exhibit a high relative permeability in the same manner as magnetic materials do, but unlike the ferromagnetic metals, they also have a high bulk resistivity. This means that at high frequencies, eddy currents induced within the materials are practically nonexistent and high-Q coils can be used. A high length-to-diameter ratio for the rod gives a high permeability, which is desirable.

The size of the coil is a compromise among several factors. If the coil is too long compared to the rod length, the change of permeability with temperature will cause a noticeable change in the inductance. If it is too short, the Q will be low. Positioning the coil on the core is critical as well, since the effective permeability is a function of position of the rod, changing from maximum at the center to a minimum at either end. The coil is usually placed near the quarter-point, allowing adjustment in either direction to trim the coil inductance. When more than one coil is mounted on the same rod, they must be placed at opposite ends to minimize interaction between them.

The coil of wire on the ferrite rod is basically a modified loop antenna, so the induced maximum emf appearing at its terminals is given by

$$
V_{s}=\omega B A N F \mu_{r}
$$

where $F$ is the modifying factor accounting for coil length, ranging from unity for short coils to about 0.7 for one that extends the full length of the rod, $\mu_{r}$ is the effective relative permeability of the rod, as measured for the actual coil position and $A$ is the rod crosssectional area

Since the voltage appearing at the terminals is of more importance in a receiving antenna, the factor $Q l_{\text {eff }}$ is often given as a figure of merit for rod antennas. The directional properties of the ferrite-rod antenna are similar to those of the loop antenna, although the null may not be quite so pronounced.
A term that has special significance for the receiving antennas is its effective area (sometimes also called the effective aperture). It is defined in terms of the directive gain of the antenna through the relation

$$
A=\frac{\lambda^{2}}{4 \pi} g_{d}
$$

The effective area is thus the ratio of power available at the antenna terminals to the power per unit area of the appropriately polarized incident wave. That is $W_{R}=P A$
where $W_{R}$ is the received power and $P$ is the power flow per square meter for the incident wave. When the directivity is used in the expression of the effective area it is assumed that all of the available power is delivered to the load. This is the case for a 100 per cent efficient correctly matched receiving antenna with the proper polarization
characteristics. For an effective field strength $E$ parallel to the antenna the power per square meter in the linearly polarized received wave is

$$
P=\frac{E^{2}}{\eta}=\frac{E^{2}}{120 \pi} \text { watts/sq m }
$$

The power is absorbed in a properly matched load connected to the antenna would be

$$
W_{R}=\frac{V_{o c}^{2}}{4 R_{r a d}}=\frac{E^{2} l_{e f f}^{2}}{4 R_{r a d}}
$$

For a current element $l_{\text {eff }}=d l$ and hence

$$
W_{R}=\frac{E^{2} d l^{2}}{4 R_{r a d}}
$$

Substitution of the value of the radiation resistance in the above gives us

$$
W_{R}=\frac{E^{2} d l^{2}}{4 R_{r a d}}=\frac{E^{2} d l^{2}}{4.80 \pi^{2}} \cdot\left(\frac{\lambda}{d l}\right)^{2}=\frac{E^{2} \lambda^{2}}{320 \pi^{2}}
$$

Hence the maximum effective area is

$$
A=\frac{W_{R}}{P}=\frac{E^{2}}{120 \pi} \div \frac{E^{2} \lambda^{2}}{320 \pi^{2}}=1.5 \frac{\lambda^{2}}{4 \pi}
$$

### 3.4 TELEVISION ANTENNA

The basic antenna for TV transmission and reception that make use of the VHF band of frequencies $(30-300 \mathrm{MHz})$ is the half-wave dipole antenna. This is called a resonant antenna as it gives out its best performance only at a particular frequency relative to its length. For antennas close to earth, vertically polarized EM waves yield better SNR. However, when the antenna is several wavelengths above the ground, horizontally polarized waves yield better SNR. The TV signals are transmitted by space wave propagation and hence the antenna height must be as high as possible in order to increase the line-of-sight distance. Horizontal polarization is the standard for TV broadcasting. Horizontal polarization indicates the plane of the electric field. Such a polarization is preferred because of the availability of good signal-to-noise ratio (SNR) when antennas are placed quite high above the surface of earth. For maximum signal pick-up, the receiving antenna should have the same polarization as that of the transmitted signal and this is the reason why the TV receiving antennas are aligned horizontally.

## LONG WIRE ANTENNAS

In the frequency band from 2 to 30 MHz long wires (several wavelengths long) supported by suitable towers may be used as efficient antennas. The best known types are the horizontal V antenna, the horizontal rhombic antenna, the vertical V and sloping rhombic antennas, the vertical inverted V or half rhombic antenna and the single horizontal-wire antenna. Most long wire antennas can be operated as resonant antennas, in which case the current on the wire will be a standing wave with the sinusoidal variation. These antennas usually operate satisfactorily only at a particular frequency and harmonics of this frequency. The input impedance will be highly frequency selective, so only narrow-band
operation is possible. Most long-wire antennas can also be operated as traveling wave or nonresonant structures by terminating the far end of the wire in a suitable resistance having a value equal to the characteristic impedance of the antenna viewed as a transmission line. In this mode of operation the useful frequency band can be quite large, with am acceptable impedance match over the whole range of frequencies.

Various types of long-wire antennas are used for commercial shortwave transmission in the frequency range from 2 to 30 MHz when propagation is by means of inonospheric reflection. For these applications the optimum angle of radiation is usually from 10 to $30^{\circ}$ relative to the horizontal line in the direction of receiving station.

Since the long-wire antennas are located in the presence of ground, the latter has an important effect on the radiation pattern and must be taken into account in the design of antenna configuration. In general, the design problem is one of obtaining a directive beam at the desired angle relative to the ground for optimum long-distance communication via reflection from the ionosphere, along with acceptable input impedance characteristics that will facilitate matching the antenna to its feed line.

A monopole antenna consists of one-half of a dipole antenna mounted above the earth or a ground plane. It is normally one-quarter-wavelength long, except with space restrictions or other factors dictate a shorter length. The vertical monopole antenna is used extensively for commercial AM broadcasting $(550-1500 \mathrm{KHz})$ in part because it is the shortest efficient antenna to use at these long wavelengths ( 200 to 600 m ) and also because vertical polarization suffers less propagation loss than horizontal polarization does at these frequencies. The monopole antenna is also widely used for the land mobilecommunication service. Quarter wavelength antennas are widely used in mobile communications with the vehicle itself providing the required ground plane. In the 27MHz citizen band, a quarter-wavelength monopole antenna is 2.77 m long. Many CB band radio users find an antenna of this length undesirable. Consequently antennas for the CB radio are often only 1 to $1,5 \mathrm{~m}$ long and use either base loading or center loading to tune the antenna to resonance. The overall efficiency will not be as great as for the fulllength antenna, since the radiation resistance is reduced quite markedly, and the unavoidable dissipative losses in the tuning coil, ground screen, and the antenna itself will consume a significant fraction of the input power.

Propagation below 2 MHz is made possible by the surface waves for which the horizontally polarized waves are attenuated much more rapidly than vertically polarized fields. For this reason horizontally oriented long wire antennas are normally not used below 2 MHz . In the shortwave band from 2 to 30 MHz , where propagation is via ionospheric reflection, long-wire antennas are effective and because of their simple structures are commonly used. Rhombic and V antennas also find some applications at frequencies from 30 to 60 MHz .

At frequencies above about 300 MHz slot antennas cut in a metallic surface, such as the skin of an aircraft or the wall of a waveguide often prove to be convenient radiators. The
slot may be fed by a generator or transmission line connected across it or in the case of the waveguide, by a guided wave incident upon the slot.
At very high frequencies $(3-30 \mathrm{GHz})$, the size of the radiating elements become very small. It is then convenient to use the concept of " current sheet radiators" such as slot antennas, horns and paraboloid antennas. The radiation from these kind of current sheet radiators can be evaluated from the individual current elements exactly as we have done for the linear radiators provided the current distribution is known or can be estimated. However, in many cases, the current distribution is neither known nor can be easily estimated. In that case, a method often used to compute the radiated fields due to continuous current distributions is the " aperture" of the antenna.

The discone antenna, the helical antenna are used in the VHF-UHF range. The former is used to radiate an omnidirectional pattern with vertical polarization. It is a broadband antenna with usable characteristics over a frequency range of nearly 10:1. It is usually designed to be fed directly from a $50-\Omega$ coaxial line and is mounted directly on the end of that line. This type of antenna is ideal for base-station operation for urban mobile communication systems, since it gives a good omnidirectional pattern, is physically very compact and rugged, and is quite inexpensive to construct. Its directional gain along the horizontal plane is comparable to that of dipole antenna.

The $\log$ periodic antenna is basically an array of dipoles, fed with alternating phase, lined up along the axis of radiation. The element lengths and their spacing all conform to a ratio, given as

$$
\tau=\frac{L_{n+1}}{L_{n}}=\frac{X_{n+1}}{X_{n}}
$$

The angle of divergence is given as

$$
\alpha=\tan ^{-1}\left(\frac{L_{n}}{X_{n}}\right)
$$

The open-end length $L$ must be larger than $\frac{\lambda}{2}$ if high efficiency is to be obtained. The impedance of this antenna is a periodic function of the logarithm of the frequency-hence its name. The antenna characteristics are broadband and it has the directional characteristics of a dipole array. This type of antenna is often used for mobile-basestation operations, where many channels must be handled over a single antenna system with good directive characteristics.

### 3.5 APERTURE ANTENNAS

The literal meaning of an aperture is an opening or a slot in a closed surface. An aperture antenna therefore, is made out of a closed surface by making a small opening or a slot that is made to carry a time-varying current and hence to radiate. Computation of the fields is not as straightforward as for the simple geometry antennas that we covered previously. The electromagnetic fields in a source free lossless region are completely specified by the tangential components of the electric and the magnetic fields on the surface enclosing the region. As the region is considered to be source free, the tangential
fields on the surface and the fields inside the region are produced by sources external to the region. As we may observe later, the aperture antenna is analyzed by making use of the tangential components of the electric field and the magnetic field. We know that, across the boundary of two media, the tangential components of the electric field $\mathbf{E}$ and $\mathbf{H}$. If the fields in a plane aperture are $\mathbf{E}_{a}$ and $\mathbf{H}_{a}$, then
$\mathbf{J}_{S}=\hat{n} \times \mathbf{H}_{a}$ and
$\mathbf{M}_{S}=-\hat{n} \times \mathbf{E}_{a}$
where $\mathbf{J}_{S}$ is the surface current density and $\mathbf{M}_{S}$ is the magnetic current density. The concept of magnetic current density may sound absurd in the first place as it would require the flow of magnetic charges in a closed path and this is not possible because of the fact that magnetic charges or monopoles do not exist. However, the concept of one would help analyzing the behavior of structures that radiate from apertures. As we have seen previously, the wire currents are suitably excited by high frequency currents to produce radiated fields. However, we cannot apply the same principle to an aperture simply because of the fact that it is an opening or a cavity (shape does not matter now) in an otherwise closed surface and conventional electric current cannot flow in such an opened out structure. We, therefore need a different approach to evaluate the radiated fields due to an aperture like structure made in a metallic surface. This approach makes use of Huygen's secondary wave principle following which the two expressions are written. A large square surface of a plane wave front acts like a rectangular array of Huygen's sources, all fed in phase. The radiation pattern of the array is, therefore, obtained by multiplying the unit pattern of the element by the array factor. The magnetic vector potential is expressed as

$$
\mathbf{A}(r, \theta, \phi)=\frac{\mu}{4 \pi} \iiint_{V^{\prime}} \mathbf{J}_{S}\left(x^{\prime}, y^{\prime}, z^{\prime}\right) \frac{\exp (-j \beta R)}{R} d v^{\prime}
$$

We have previously seen from (2.75) that the electric field can also be evaluated in terms of the magnetic vector potential. Combined with the gradient of the scalar potential, we have

$$
\begin{aligned}
& \mathbf{E}=-j \omega \mathbf{A}-\nabla V=-j \omega \mathbf{A}-\nabla(\nabla \cdot \mathbf{A}) \\
& \mathbf{H}=\frac{1}{\mu} \nabla \times \mathbf{A}
\end{aligned}
$$

Similarly, we may define an electric vector potential due to the magnetic current flow as the following:

$$
\mathbf{F}(r, \theta, \phi)=\frac{\varepsilon}{4 \pi} \iiint_{V^{\prime}} \mathbf{M}_{S}\left(x^{\prime}, y^{\prime}, z^{\prime}\right) \frac{\exp (-j \beta R)}{R} d v^{\prime}
$$

Here, $V^{\prime}$ is the volume that contains the magnetic source current $M_{S}$. In similar lines, we may write

$$
\begin{aligned}
& \mathbf{E}=-\frac{1}{\varepsilon} \nabla \times \mathbf{F} \\
& \text { Making use of } \\
& \nabla \times \mathbf{F}=\frac{\partial F_{\theta}}{\partial r} a_{\phi}-\frac{\partial F_{\phi}}{\partial r} a_{\theta}
\end{aligned}
$$

The electric field in terms of the electric vector potential is expressed as

$$
\begin{aligned}
& \mathbf{E}=-\frac{1}{\varepsilon}\left(\frac{\partial F_{\theta}}{\partial r} a_{\phi}-\frac{\partial F_{\phi}}{\partial r} a_{\theta}\right) \\
& \mathbf{E}=-\frac{1}{\varepsilon}\left[-\frac{j \beta}{r} F_{\theta 0}(r, \theta, \phi) \exp (-j \beta r) a_{\phi}+\frac{j \beta}{r} F_{\phi 0}(r, \theta, \phi) \exp (-j \beta r) a_{\theta}\right] \\
& \\
& =j \frac{\beta}{\varepsilon}\left[F_{\theta}(r, \theta, \phi) a_{\phi}-F_{\phi}(r, \theta, \phi) a_{\theta}\right] \\
& \\
& =j \omega \eta\left[F_{\theta}(r, \theta, \phi) a_{\phi}-F_{\phi}(r, \theta, \phi) a_{\theta}\right]
\end{aligned}
$$

and the corresponding magnetic field as
$\mathbf{H}=-j \omega \mathbf{F}-\nabla(\nabla . \mathbf{F})$
The total electric field due to the magnetic vector potential, electric vector potential, magnetic scalar potential and electric scalar potential is expressed as
$\mathbf{E}=-\frac{1}{\varepsilon} \nabla \times \mathbf{F}-j \omega \mathbf{A}-\frac{j}{\omega \mu \varepsilon} \nabla(\nabla . \mathbf{A})$
and similarly,
$\mathbf{H}=-\frac{1}{\mu} \nabla \times \mathbf{A}-j \omega \mathbf{F}-\frac{j}{\omega \mu \varepsilon} \nabla(\nabla \cdot \mathbf{F})$
Now, we can express the radiated field component due to all the factors combined together as
$\left.\mathbf{E}=j \omega \eta \mid F_{\theta}(r, \theta, \phi) a_{\phi}-F_{\phi}(r, \theta, \phi) a_{\theta}\right]-j \omega\left(A_{\theta} a_{\theta}+A_{\phi} a_{\phi}\right)$
Separating the $\theta$ - and the $\phi$-components, we have
$E_{\theta} \approx-j \omega\left\lfloor\eta F_{\phi}(r, \theta, \phi)+A_{\theta}\right\rfloor$ and
$E_{\phi} \approx-j \omega\left[A_{\phi}-\eta F_{\theta}(r, \theta, \phi)-\right]$
The corresponding magnetic fields are expressed as
$H_{\theta} \approx-\frac{E_{\phi}}{\eta}$
$H_{\phi} \approx \frac{E_{\theta}}{\eta}$
Next, we wish to evaluate the radiated fields due to a rectangular aperture of size $(a \times b)$ placed in the $x y$-plane and there is an electric field along the $x$ axis and a magnetic field along the $y$ - axis. Further, these fields are assumed to be constants over the aperture. Let $\mathbf{E}_{a}=E_{0} a_{y}$ and $\mathbf{H}_{a}=-H_{0} a_{x}$. The magnetic surface current through the aperture is
$\mathbf{M}_{S}=-\hat{n} \times \mathbf{E}=-a_{z} \times E_{0} a_{y}=E_{0} a_{x}$
Similarly, the electric current density through the aperture is
$\mathbf{J}_{S}=\hat{n} \times \mathbf{H}_{a}=a_{z} \times-H_{0} a_{x}=-H_{0} a_{y}$
The electric vector potential is

$$
\begin{align*}
& \mathbf{F}(r, \theta, \phi)=\frac{\varepsilon}{4 \pi} \iiint_{V^{\prime}} \mathbf{M}_{S}\left(x^{\prime}, y^{\prime}, z^{\prime}\right) \frac{\exp (-j \beta R)}{R} d v^{\prime} \\
& =\frac{\varepsilon}{4 \pi} \iiint_{V^{\prime}} E_{0} a_{x} \frac{\exp (-j \beta R)}{R} d v^{\prime}  \tag{A}\\
& =\frac{\varepsilon}{4 \pi} \cdot E_{0} a_{x} \iiint_{V^{\prime}} \frac{\exp (-j \beta R)}{R} d x^{\prime} d y^{\prime}
\end{align*}
$$

Hence, the integral reduces to computing the exponential term. We find that, a little manipulation of (2.49) results in the following simplified expression for the distant point located $R$ units away from the origin. Therefore,

$$
\begin{aligned}
& R=\left[\left(x-x^{\prime}\right)^{2}+\left(y-y^{\prime}\right)^{2}+\left(z-z^{\prime}\right)^{2}\right] \\
& =\left[r^{2}+\left(x^{\prime 2}+y^{\prime 2}\right)-2 r\left(x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right)\right]^{1 / 2} \\
& =\left[r^{2}+r^{\prime 2}-2 r\left(x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right)\right]^{\frac{1}{2}} \\
& =r\left[1+\left(\frac{r^{\prime}}{r}\right)^{2}-\frac{2}{r}\left(x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right)\right]^{\frac{1}{2}} \\
& \cong r\left[1-\frac{1}{r}\left(x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right)+\frac{1}{2}\left(\frac{r^{\prime}}{r}\right)^{2}+\ldots\right] \\
& =r\left[1-\frac{1}{r}\left(x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right)\right]
\end{aligned}
$$

Substituting this in (A), we have

$$
\begin{aligned}
& \mathbf{F}(r, \theta, \phi)=\frac{\varepsilon}{4 \pi} \cdot-E_{0} a_{x} \iiint_{S} \frac{\exp (-j \beta R)}{R} d x^{\prime} d y^{\prime} \\
& =\frac{\varepsilon}{4 \pi r} \cdot-E_{0} a_{x} \iiint_{S} \exp \left(-j \beta\left[r-x^{\prime} \sin \theta \cos \phi-y^{\prime} \sin \theta \sin \phi\right]\right) d x^{\prime} d y^{\prime} \\
& =-\frac{\varepsilon E_{0}}{4 \pi r} \exp (-j \beta r) \iiint_{S} \exp \left(j \beta x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right) d x^{\prime} d y^{\prime} \\
& =\varepsilon E_{0} G I
\end{aligned}
$$

where $I$ represents the integral over the aperture surface and $G$ is the familiar scalar spherical wave function given as

$$
G=\frac{\exp (-j \beta r)}{4 \pi r}
$$

The integral $I$ is evaluated as

$$
\begin{aligned}
& \int_{-a / 2}^{a / 2} \exp \left(j \beta x^{\prime} \sin \theta \cos \phi\right) d x^{\prime} \\
& =\left.\frac{1}{j \beta \sin \theta \cos \phi} \exp \left(j \beta x^{\prime} \sin \theta \cos \phi\right)\right|_{-a / 2} ^{a / 2} \\
& =\frac{1}{j \beta \sin \theta \cos \phi}\left[\exp \left(j \beta \frac{a}{2} \sin \theta \cos \phi\right)-\exp \left(-j \beta \frac{a}{2} \sin \theta \cos \phi\right)\right] \\
& =\frac{2 j \sin \left(\beta \frac{a}{2} \sin \theta \cos \phi\right)}{j \beta \sin \theta \cos \phi}=\frac{a \sin \left(\beta \frac{a}{2} \sin \theta \cos \phi\right)}{(a / 2) \beta \sin \theta \cos \phi} \\
& =a \sin c\left(\beta \frac{a}{2} \sin \theta \cos \phi\right)
\end{aligned}
$$

Therefore,
$\iiint_{S} \exp \left(j \beta x^{\prime} \sin \theta \cos \phi+y^{\prime} \sin \theta \sin \phi\right) d x^{\prime} d y^{\prime}$
$=a b S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
where $S a(x)=\frac{\sin x}{x}$.The electric vector potential has a component along the $x$-axis only. In a similar manner, we can evaluate the magnetic vector potential also by writing the following:

$$
\begin{aligned}
& \mathbf{A}(r, \theta, \phi)=\frac{\mu}{4 \pi} \iiint_{V^{\prime}} \mathbf{J}_{S}\left(x^{\prime}, y^{\prime}, z^{\prime}\right) \frac{\exp (-j \beta R)}{R} d v^{\prime} \\
& =\frac{\mu}{4 \pi} \iiint_{V^{\prime}}-H_{0} a_{y} \frac{\exp (-j \beta R)}{R} d v^{\prime} \\
& =\frac{\mu}{4 \pi} \cdot-H_{0} a_{y} \iiint_{S} \frac{\exp (-j \beta R)}{R} d x^{\prime} d y^{\prime}
\end{aligned}
$$

This turns out to be
$\mathbf{A}(r, \theta, \phi)=\frac{\mu}{4 \pi} .-H_{0} a_{y} \iiint_{S} \frac{\exp (-j \beta R)}{R} d x^{\prime} d y^{\prime}$
$=-\mu H_{0} G I$
We know from elementary coordinate geometry that
$A_{\theta}=A_{x} \cos \theta \cos \phi, \quad A_{\phi}=-A_{x} \sin \phi$ and
$F_{\theta}=F_{y} \cos \theta \sin \phi, \quad F_{\phi}=F_{y} \cos \phi$
From all these above discussions, it is clear that the magnetic vector potential has component along the $y$-axis only. We now combine these two results and write the overall radiated fields as

Therefore,
$E_{\theta}=-j \omega\left\lfloor\eta F_{\phi}(r, \theta, \phi)+A_{\theta}\right\rfloor$
$=-j \omega\left[-\eta F_{x} \sin \phi+A_{y} \cos \theta \sin \phi\right]$
$=-j \omega\left[-\eta \varepsilon E_{0} G I \sin \phi-\mu H_{0} G I \cos \theta \sin \phi\right]$
$=j \omega a b G\left(\eta \varepsilon E_{0}+\mu H_{0} \cos \theta\right) \sin \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
$=j \omega a b G E_{0}\left(\eta \varepsilon+\frac{\mu}{\eta} \cos \theta\right) \sin \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
This is due to the fact that $\mu H_{0}=\frac{\mu}{\eta} E_{0} \quad$ and $\quad \eta \varepsilon=\sqrt{\frac{\mu}{\varepsilon}} . \varepsilon=\sqrt{\mu \varepsilon}=\frac{1}{c} \quad$ and $\frac{\mu}{\eta}=\mu \cdot \sqrt{\frac{\varepsilon}{\mu}}=\sqrt{\mu \varepsilon}=\frac{1}{c}$ and further, $\frac{\omega}{c}=\frac{2 \pi}{\lambda}=\beta$.This is equal to, after the substitution of the necessary expressions that
$E_{\theta}=j \omega a b G E_{0}(\eta \varepsilon+\cos \theta) \sin \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
$=j \beta a b E_{0}(1+\cos \theta) \sin \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
Similarly,
$E_{\phi} \approx-j \omega\left[A_{\phi}-\eta F_{\theta}(r, \theta, \phi)-\right]$
$=-j \omega\left[A_{y} \cos \phi-\eta F_{x} \cos \theta \cos \phi\right]$
$=-j \omega\left[-\mu H_{0} G I \cos \phi-\eta \varepsilon E_{0} G I \cos \theta \cos \phi\right]$
$=j \omega G\left(\mu H_{0}+\eta \varepsilon E_{0} \cos \theta\right) \cos \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
$=j \beta G a b E_{0}(1+\cos \theta) \cos \phi S a\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$
From the expressions for the electric field in the $\theta-\phi$ plane, we observe that the maximum value of $S a(x)$ is obtained for $x=0$. This corresponds to the case of $\theta=0$. The associated magnetic fields are given as (B). In the far field region, the radiation intensity is $r^{2}$ times the time-averaged power density and this is expressed as
$U(\theta, \phi)=\frac{r^{2}}{2 \eta}\left[\left|E_{\theta}\right|^{2}+\left|E_{\phi}\right|^{2}\right]$ ©
Substitution of the necessary expressions in © gives us
$U(\theta, \phi)=\frac{r^{2}}{2 \eta}\left[\left|E_{\theta}\right|^{2}+\left|E_{\phi}\right|^{2}\right]$
$=\frac{\beta^{2}}{2 \eta(4 \pi)^{2}}(a b)^{2}\left|E_{0}\right|^{2}(1+\cos \theta)^{2} S a^{2}\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a^{2}\left(\frac{\beta b}{2} \sin \theta \sin \phi\right)$

Therefore, the maximum radiation intensity is obtained by substituting $\theta=0$ in (D) and hence we obtain

$$
\begin{aligned}
& U(\theta, \phi)=\frac{r^{2}}{2 \eta}\left[\left|E_{\theta}\right|^{2}+\left|E_{\phi}\right|^{2}\right] \\
& =\frac{\beta^{2}}{2 \eta(4 \pi)^{2}}(a b)^{2}\left|E_{0}\right|^{2}(1+\cos \theta)^{2} S a^{2}\left(\frac{\beta a}{2} \sin \theta \cos \phi\right) S a^{2}\left(\frac{\beta b}{2} \sin \theta \sin \phi\right) \\
& =\beta^{2} \frac{(a b)^{2}\left|E_{0}\right|^{2}}{2 \eta(4 \pi)^{2}} \cdot 4 \\
& =\frac{\left|E_{0}\right|^{2}}{2 \eta}\left(\frac{2 \pi}{\lambda}\right)^{2}\left(\frac{a b}{2 \pi}\right)^{2} \\
& =\frac{\left|E_{0}\right|^{2}}{2 \eta}\left(\frac{a b}{\lambda}\right)^{2}
\end{aligned}
$$

The radiated power passing through the aperture area is found out by evaluating the Poynting vector and integrating it over the aperture. Doing this, we obtain
$\int_{\Omega} U(\theta, \phi) d \Omega=\oint_{S}\left(\mathbf{E}_{a} \times \mathbf{H}_{a}\right) \cdot d \mathbf{s}$
As $\mathbf{E}_{a}=E_{0} a_{y}$ and $\mathbf{H}_{a}=-H_{0} a_{x}$, the power radiated through the aperture becomes

$$
\begin{aligned}
& \frac{1}{2} \oint_{S}\left(\mathbf{E}_{a} \times \mathbf{H}_{a}\right) \cdot d \mathbf{s}=\frac{1}{2} \oint_{S}\left(E_{0} a_{y} \times-H_{0} a_{x}\right) \cdot d \mathbf{s} \\
& =\frac{1}{2}\left|E_{0}\right| \cdot \frac{\left|E_{0}\right|}{\eta} a b=\frac{a b}{2 \eta}\left|E_{0}\right|^{2}
\end{aligned}
$$

The directivity reduces to
$D_{\text {max }}=\frac{4 \pi\left(\frac{a b}{\lambda}\right)^{2}}{a b \frac{\left|E_{0}\right|^{2}}{2 \eta}} \frac{\left|E_{0}\right|^{2}}{2 \eta}=4 \pi\left(\frac{a b}{\lambda}\right)^{2}$
The expression for directivity shows that ratio of the directivity to the aperture area is a constant which is equal to $\left(\frac{4 \pi}{\lambda}\right)^{2}$ and is independent of the antenna size. This depends only on the wavelength or frequency of operation. Less the wavelength, higher is the value. However, if the current distribution is not uniform, then an equivalent uniform aperture area is defined as the maximum effective aperture area and then the expression for directivity gets modified as

$$
D_{\max }=\frac{4 \pi}{\lambda^{2}} A_{e}
$$

where $A_{e}$ is the effective aperture area. For a uniform current distribution, the effective area is the same as that of the actual area of the aperture.

### 3.6 Rectangular Waveguides

A rectangular waveguide is used to carry energy from a source to a destination at microwave and optical frequencies. This is a hollow metallic pipe with a rectangular cross section. It is assumed that the walls of the waveguide have infinite conductivity and the medium filing up the space is an ideal dielectric with a given permeability, permittivity and no conductivity. The waveguide is assumed to have a rectangular cross section of dimension $a \times b$ and is infinitely long. We may assume here that $a \geq b$ and the broader dimension; $a$ is oriented along the $x$-axis and the shorter dimension $b$ is aligned along the $y$-axis. The length of the waveguide is aligned along the $z$-axis. We further assume wave propagation to take place along the positive $z$-axis.


A hollow rectangular waveguide with dimension $a \times b$
We may note that the wave propagation inside a hollow structure such as the one shown in Fig. takes place in the form of modes. It is no longer the uniform plane waves or the TEM mode that is possible in an unbounded homogeneous medium. As we may see, the walls of the waveguide are made of a good conductor while the hollow space is filled with some kind of dielectric. Hence, any wave propagating inside the space has to be governed by boundary conditions along the two dimensions; $x$ - and the $y$-axes. As we shall see, application of the boundary conditions at the cross section gives rise to modes for the waves. A mode is characterized by an integer number of half sinusoids along the particular dimension under consideration. The simplest being just half a sinusoid typically known as the dominant mode. The waves have to be transverse electric (TE), transverse magnetic ${ }^{\text {TM }}$ waves or some definite combination of these two known as hybrid modes. As we know, the longitudinal electric field is zero while there exists a nonzero longitudinal magnetic field for the TE waves. Let us analyze the TE waves first.

## Transverse Electric (TE) Waves

As the magnetic field has a nonzero component along the $z$-axis, the wave equation becomes
$\nabla^{2} H_{z}+\omega^{2} \mu \varepsilon H_{z}=0$
Expansion of this gives us
$\left(\frac{\partial^{2}}{\partial x^{2}}+\frac{\partial^{2}}{\partial y^{2}}+\frac{\partial^{2}}{\partial z^{2}}\right) H_{z}+\omega^{2} \mu \varepsilon H_{z}=0$
Let $H_{z}(x, y, z, t)=X(x) Y(y) Z(z) \exp (j \omega t)$
Performing the double differentiation, we obtain
$Y Z \frac{d^{2} X}{d x^{2}}+Z X \frac{d^{2} Y}{d y^{2}}+X Y \frac{d^{2} Z}{d z^{2}}+\omega^{2} \mu \varepsilon X Y Z=0$
$\Rightarrow \frac{1}{X} \frac{d^{2} X}{d x^{2}}+\frac{1}{Y} \frac{d^{2} Y}{d y^{2}}+\frac{1}{Z} \frac{d^{2} Z}{d z^{2}}+\omega^{2} \mu \varepsilon=0$
Dividing both sides by $X Y Z$
Now, we can equate each of these terms to a constant as shown below
$\frac{1}{X} \frac{d^{2} X}{d x^{2}}=-A^{2}, \frac{1}{Y} \frac{d^{2} Y}{d y^{2}}=-B^{2}$ and $\frac{1}{Z} \frac{d^{2} Z}{d z^{2}}=-\beta^{2}$
The differential equation becomes
$\frac{1}{X} \frac{d^{2} X}{d x^{2}}+\frac{1}{Y} \frac{d^{2} Y}{d y^{2}}-\beta^{2}+\omega^{2} \mu \varepsilon=0$
$\Rightarrow-A^{2}-B^{2}+h^{2}=0$
$\Rightarrow h^{2}=A^{2}+B^{2}$
In the above equation, we have made use of

$$
\omega^{2} \mu \varepsilon-\beta^{2}=h^{2}
$$

We have, therefore
$\frac{d^{2} X}{d x^{2}}=-A^{2} X$
$\Rightarrow X=C_{1} \sin A x+C_{2} \cos A x$
Similarly,

$$
\frac{d^{2} Y}{d y^{2}}=-B^{2} Y
$$

$\Rightarrow Y=C_{3} \sin B y+C_{4} \cos B y$
The complete solution of the longitudinal magnetic field, therefore becomes $H_{z}(x, y, z, t)=\left(C_{1} \sin A x+C_{2} \cos A x\right)\left(C_{3} \sin B y+C_{4} \cos B y\right) \exp (-j \beta z) \exp (j \omega t)$
We know that the tangential electric fields are $E_{x}$ and $E_{y}$ which are related to the magnetic field $H_{z}$ in the following manner:
$E_{x}=-\frac{j \omega \mu}{h^{2}} \frac{\partial H_{z}}{\partial y}$
$\Rightarrow E_{x}=-\frac{j \omega \mu}{h^{2}}\left(C_{1} \sin A x+C_{2} \cos A x\right)\left(C_{3} B \cos B y-C_{4} B \sin B y\right) \exp (-j \beta z) \exp (j \omega t)$
And similarly,
$E_{y}=\frac{j \omega \mu}{h^{2}} \frac{\partial H_{z}}{\partial x}$
$\Rightarrow E_{y}=\frac{j \omega \mu}{h^{2}}\left(C_{1} A \cos A x-C_{2} A \sin A x\right)\left(C_{3} \sin B y+C_{4} \cos B y\right) \exp (-j \beta z) \exp (j \omega t)$
The waveguide walls are metallic surfaces bounded by $x=0$ to $x=a$ on the broader side and $y=0$ to $y=b$ on the narrower side. The tangential components of the electric fields are zero at the walls. Therefore, both $E_{x}$ and $E_{y}$ are zero, however, at different surfaces. The $E_{x}$ field is zero at $y=0$ and $y=b$ while the $E_{y}$ is zero at $x=0$ and $x=a$, Substituting these into the two tangential electric fields, we have $C_{3}=0$ obtained for $E_{x}$ and $C_{1}=0$ for $E_{y}$. Similarly, at the ends of the cross section, we get

$$
\begin{aligned}
& E_{x}=-\frac{j \omega \mu}{h^{2}}\left(C_{1} \sin A x+C_{2} \cos A x\right)\left(-C_{4} B \sin B b\right) \exp (-j \beta z) \exp (j \omega t)=0 \\
& \Rightarrow B=\frac{n \pi}{b}
\end{aligned}
$$

Similarly,
$E_{y}=\frac{j \omega \mu}{h^{2}}\left(-C_{2} A \sin A x\right)\left(C_{3} \sin B y+C_{4} \cos B y\right) \exp (-j \beta z) \exp (j \omega t)=0$
$\Rightarrow A=\frac{m \pi}{a}$
The resultant solution, therefore becomes

$$
\begin{aligned}
& E_{x}=\frac{j \omega \mu}{h^{2}}\left(\frac{n \pi}{b}\right)\left(C_{2} \cos \frac{m \pi}{a} x\right)\left(C_{4} B \sin \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \\
& =\frac{j \omega \mu}{h^{2}} K\left(\frac{n \pi}{b}\right)\left(\cos \frac{m \pi}{a} x\right)\left(\sin \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)
\end{aligned}
$$

We have expressed $C_{2} C_{4}=K$
Similarly,

$$
\begin{aligned}
& E_{y}=\frac{j \omega \mu}{h^{2}}\left(\frac{m \pi}{a}\right)\left(-C_{2} A \sin \frac{m \pi}{a} x\right)\left(C_{4} \cos \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \\
& =-\frac{j \omega \mu}{h^{2}}\left(\frac{m \pi}{a}\right) K\left(\sin \frac{m \pi}{a} x\right)\left(\cos \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)
\end{aligned}
$$

The tangential magnetic fields are expressed as

$$
\begin{aligned}
& H_{x}=\frac{j \omega \varepsilon}{h^{2}} \frac{\partial E_{z}}{\partial y}-\frac{j \beta}{h^{2}} \frac{\partial H_{z}}{\partial x} \\
& =-\frac{j \beta}{h^{2}}\left(\frac{m \pi}{a}\right) K\left(\sin \frac{m \pi}{a} x\right)\left(-\cos \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \Theta E_{z}=0 \\
& =\frac{j \beta}{h^{2}}\left(\frac{m \pi}{a}\right) K\left(\sin \frac{m \pi}{a} x\right)\left(\cos \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \\
& H_{y}=-\frac{j \omega \varepsilon}{h^{2}} \frac{\partial E_{z}}{\partial x}-\frac{j \beta}{h^{2}} \frac{\partial H_{z}}{\partial y} \\
& =\frac{j \beta}{h^{2}} K\left(\frac{n \pi}{b}\right)\left(\cos \frac{m \pi}{a} x\right)\left(\sin \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \Theta E_{z}=0
\end{aligned}
$$

and finally,

$$
\begin{aligned}
& H_{z}(x, y, z, t)=\left(C_{2} \cos \frac{m \pi}{a} x\right)\left(C_{4} \cos \frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t) \\
& =K \cos \left(\frac{m \pi}{a}\right) x \cos \left(\frac{n \pi}{b}\right) y \exp (-j \beta z) \exp (j \omega t)
\end{aligned}
$$

We may further verify that,
$h^{2}=\omega^{2} \mu \varepsilon-\beta^{2}=$
$A^{2}+B^{2}=\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}$
The integers $m$ and $n$ together define the order of the mode the wave takes on to propagate down the walls of the waveguide. The mode is designated as $T E_{m n}$ mode. We observe from the expressions for the transverse fields that, either of the integers may be zero for these to be nonzero as each of the four fields contain a half cosine term given by $\cos \left(\frac{m \pi}{a} x\right)$ or $\cos \left(\frac{n \pi}{b} y\right)$. If both of the integers are zero, then the longitudinal magnetic field $H_{z}$ becomes non zero, however, all the transverse fields become zero. As we know, a magnetic field variation in space is not possible without a corresponding electric field and vice versa. Thus, there can be no $T E_{00}$ mode. We can have, as the lowest order modes corresponding to either $m=0$ giving rise to $T E_{01}$ mode or a $T E_{10}$ mode corresponding to $n=0$.
Prob: Establish the field relations from the four basic Maxwell's equations.
Prob: Establish the dimensional of $h^{2}$. What is its significance?
Prob: Sketch $\cos \left(\frac{m \pi}{a} x\right)$ and $\sin \left(\frac{n \pi}{b} y\right)$ mark your observations.

## Transverse Magnetic (TM) Waves

Similar to the TE waves, the TM waves are characterized by a longitudinal electric field; $E_{z} \neq 0$ and the magnetic field $H_{z}=0$. Therefore, the equation governing the TM mode is expressed as
$\nabla^{2} E_{z}+\omega^{2} \mu \varepsilon E_{z}=0$
Expansion of this gives us
$\left(\frac{\partial^{2}}{\partial x^{2}}+\frac{\partial^{2}}{\partial y^{2}}+\frac{\partial^{2}}{\partial z^{2}}\right) E_{z}+\omega^{2} \mu \varepsilon E_{z}=0$
Let $E_{z}(x, y, z, t)=X(x) Y(y) Z(z) \exp (j \omega t)$
Performing the double differentiation, we obtain
$Y Z \frac{d^{2} X}{d x^{2}}+Z X \frac{d^{2} Y}{d y^{2}}+X Y \frac{d^{2} Z}{d z^{2}}+\omega^{2} \mu \varepsilon X Y Z=0$
$\Rightarrow \frac{1}{X} \frac{d^{2} X}{d x^{2}}+\frac{1}{Y} \frac{d^{2} Y}{d y^{2}}+\frac{1}{Z} \frac{d^{2} Z}{d z^{2}}+\omega^{2} \mu \varepsilon=0$
Dividing both sides by $X Y Z$
Now, we can equate each of these terms to a constant as shown below
$\frac{1}{X} \frac{d^{2} X}{d x^{2}}=-A^{2}, \frac{1}{Y} \frac{d^{2} Y}{d y^{2}}=-B^{2}$ and $\frac{1}{Z} \frac{d^{2} Z}{d z^{2}}=-\beta^{2}$
The differential equation becomes
$\frac{1}{X} \frac{d^{2} X}{d x^{2}}+\frac{1}{Y} \frac{d^{2} Y}{d y^{2}}-\beta^{2}+\omega^{2} \mu \varepsilon=0$
$\Rightarrow-A^{2}-B^{2}+h^{2}=0$
In the above equation, we have made use of
$\omega^{2} \mu \varepsilon-\beta^{2}=h^{2}$
We have, therefore
$\frac{d^{2} X}{d x^{2}}=-A^{2} X$
$\Rightarrow X=C_{1} \sin A x+C_{2} \cos A x$
Similarly,
$\frac{d^{2} Y}{d y^{2}}=-B^{2} Y$
$\Rightarrow Y=C_{3} \sin B y+C_{4} \cos B y$
The complete solution of the longitudinal electric field, therefore becomes
$E_{z}(x, y, z, t)=\left(C_{1} \sin A x+C_{2} \cos A x\right)\left(C_{3} \sin B y+C_{4} \cos B y\right) \exp (-j \beta z) \exp (j \omega t)$ (A)
The $E_{z}$ field is a tangential field along the $z$-axis which is assumed to be infinite in extent. Also, we note that
$E_{z}=0$ at $x=0$ and $x=a$ and likewise $E_{z}=0$ at $y=0$ and $y=b$
Substitution of these in (A) gives us $C_{2}=0$ and further, $C_{4}=0$. Again, we have
$A=\frac{m \pi}{a}$ and $B=\frac{n \pi}{b}$ at the other ends of the rectangular wall.
The complete solution to the longitudinal field then is,
$E_{z}(x, y, z, t)=C_{1} \sin \left(\frac{m \pi}{a} x\right) C_{3} \sin \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)$
$=K \sin \left(\frac{m \pi}{a} x\right) \sin \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)$
In the above, we have expressed $C_{1} C_{3}=K$
The other transverse fields are obtained as follows:
$E_{x}=-\frac{j \beta}{h^{2}} \frac{\partial E_{z}}{\partial x}$
$\Rightarrow E_{x}=-\frac{j \beta}{h^{2}}\left(\frac{m \pi}{a}\right) K \cos \left(\frac{m \pi}{a} x\right) \sin \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)$
$E_{y}=-\frac{j \beta}{h^{2}} \frac{\partial E_{z}}{\partial y}$
$\Rightarrow E_{x}=-\frac{j \beta}{h^{2}}\left(\frac{n \pi}{b}\right) K \sin \left(\frac{m \pi}{a} x\right) \cos \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)$
$H_{x}=\frac{j \omega \varepsilon}{h^{2}} \frac{\partial E_{z}}{\partial y}$
$\Rightarrow H_{x}=\frac{j \omega \varepsilon}{h^{2}}\left(\frac{n \pi}{b}\right) K \sin \left(\frac{m \pi}{a} x\right) \cos \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)$

$$
H_{y}=-\frac{j \omega \varepsilon}{h^{2}} \frac{\partial E_{z}}{\partial x}
$$

and,

$$
\Rightarrow E_{x}=-\frac{j \omega \varepsilon}{h^{2}}\left(\frac{m \pi}{a}\right) K \cos \left(\frac{m \pi}{a} x\right) \sin \left(\frac{n \pi}{b} y\right) \exp (-j \beta z) \exp (j \omega t)
$$

As before, we note that
$h^{2}=\omega^{2} \omega \varepsilon-\beta^{2}=A^{2}+B^{2}=\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}$
The integers $m$ and $n$ together define the order of the mode the wave takes on to propagate down the walls of the waveguide. The mode is designated as $T M_{m n}$ mode. We observe from the expressions for the transverse fields that, if either of the integers is zero, then the fields become zero as each of the four fields contain a half sine term given by $\cos \left(\frac{m \pi}{a} x\right)$ or $\cos \left(\frac{n \pi}{b} y\right)$. Thus, there can be no $T M_{00}$ mode. Similarly, as the lowest order modes corresponding to either $m=0$ giving rise to $T M_{01}$ mode or a $T M_{10}$ mode corresponding to $n=0$ are also not possible. The lowest order mode, therefore is attained only when both the integers are at least equal to one, $m=1=n$. This is the lowest order or the fundamental mode of a TM wave known as $T M_{11}$ mode.

## MODULE-IV

Exercise: Enumerate the fundamental differences between a TE wave and a TM wave.
We observe from all the above discussions that for both the waves, fields exist in discrete patterns of half sinusoids or cosinusiods corresponding to only integer values of $m$ and $n$. Therefore, $m$ and $n$ can take on only discrete values and not a continuum of values. This makes the field pattern for each of the sets of $m$ and $n$ unique and also the fundamental mode of propagation. We can also see that, for the guided waves, other than the fundamental mode, all other modes if allowed to exist, are the higher order modes. This essentially means an integer number of half cycles contained along the appropriate dimension. For example, if a mode is given as $T E_{22}$, it simply means that there is one full cycle of field variation (two half cycles of alternating polarity) each along the $x$-axis and the $y$-axis.

At this point, we would like to find out which modes can be supported by a given waveguide. In order to answer that, we note that

$$
\beta^{2}=\omega^{2} \mu \varepsilon-\left(\frac{m \pi}{a}\right)^{2}-\left(\frac{n \pi}{b}\right)^{2}
$$

Therefore,
$\beta=\sqrt{\omega^{2} \mu \varepsilon-\left(\frac{m \pi}{a}\right)^{2}-\left(\frac{n \pi}{b}\right)^{2}}$
We note that, the $\omega^{2} \mu \varepsilon>\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}$ in order for $\beta$ to be real. And for a give mode to exist in the waveguide, the value of $\beta$ must be real. The frequency at which $\beta$ changes from real to imaginary is known as the cut off frequency, given as

$$
\begin{align*}
& \omega_{c}=\frac{1}{\sqrt{\mu \varepsilon}} \sqrt{\left[\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}\right]} \\
& \Rightarrow f_{c}=\frac{1}{2 \pi \sqrt{\mu \varepsilon}} \sqrt{\left[\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}\right]} \tag{D}
\end{align*}
$$

It follows from (D) that, the fundamental mode for a TE wave has the following cut off frequencies corresponding to $T E_{10}$ as
$f_{c}\left(T E_{10}\right)=\frac{1}{2 \pi \sqrt{\mu \varepsilon}}\left(\frac{\pi}{a}\right)=\frac{1}{2 a \sqrt{\mu \varepsilon}}$
Similarly,
$f_{c}\left(T E_{01}\right)=\frac{1}{2 \pi \sqrt{\mu \varepsilon}}\left(\frac{\pi}{b}\right)=\frac{1}{2 b \sqrt{\mu \varepsilon}}$
And for the TM wave, the fundamental mode has a cut off frequency given as

$$
\begin{aligned}
& f_{c}\left(T M_{11}\right)=\frac{1}{2 \pi \sqrt{\mu \varepsilon}} \sqrt{\left[\left(\frac{\pi}{a}\right)^{2}+\left(\frac{\pi}{b}\right)^{2}\right]} \\
& =\frac{1}{2 \sqrt{\mu \varepsilon}} \sqrt{\left[\left(\frac{1}{a}\right)^{2}+\left(\frac{1}{b}\right)^{2}\right]}
\end{aligned}
$$

As $a>b$, therefore, we have
$f_{c}\left(T E_{10}\right)<f_{c}\left(T E_{01}\right)<f_{c}\left(T M_{11}\right)$
We, therefore see that the lowest frequency that can propagate inside a rectangular waveguide corresponds to the $T E_{10}$ mode. No power transfer is possible corresponding to a frequency lower than the cutoff frequency. This is simply due to the fact that at frequencies lower than the cut off frequency, the propagation constant $\beta$ becomes imaginary. Thus, the term $\exp (-j \beta z)=\exp (\beta z)$ does not represent a wave. In this aspect then, a waveguide can be thought of as a high pass filter allowing frequencies above the cut off frequency. We can further see that, as the order of the mode increases, the cutoff frequency similarly increases. The $T E_{10}$ is known as the dominant mode of a rectangular waveguide.
Similar to the cutoff frequency, the cut off wavelength for a given mode is given as

$$
\lambda_{c}=\frac{v}{f_{c}}=\frac{v}{\frac{v}{2 \pi} \sqrt{\left[\left(\frac{m \pi}{a}\right)^{2}+\left(\frac{n \pi}{b}\right)^{2}\right]}}=\frac{2}{\sqrt{\left[\left(\frac{m}{a}\right)^{2}+\left(\frac{n}{b}\right)^{2}\right]}}
$$

This is due to the fact that the intrinsic velocity of a wave in a waveguide is given as
$v=\frac{1}{\sqrt{\mu \varepsilon}}$.
It follows from the above discussion that, the cutoff wavelength corresponding to the dominant $T E_{10}$ mode is
$\lambda_{c}\left(T E_{10}\right)=\frac{2}{\sqrt{\left[\left(\frac{1}{a}\right)^{2}\right]}}=2 a$
$\Rightarrow a=\frac{\lambda_{c}\left(T E_{10}\right)}{2}$
The longest wavelength that can propagate inside a rectangular waveguide corresponds to the dominant mode and is equal to twice the size of its broader dimension.
The Biconical Antenna

This antenna has broadband Chanoctenistict. It is
analogous to an infinite un form tracumistion line TK
acts as a guide for a sphencat wave in the same way that a uniform thantmibion tine acts of a guide for a plane wove tine proceed now to derive the.
Charactenstic impedance of this antenna Let a generates be connected to the terminal of an inficike biconical antenna The gerenates causer wove s with iphencal phase front e to travel radially outward from the terminals The waver produce wronents on the cones and a voltage between them Lex $v$ be the voltage between perse on the upper ind lowe cones a dibtance from the terming as. Let I be the fokal curvent on the outface of one of the conte at a dixtarice or from the terminals


As on an ordinary transmission line, the ratio $\frac{V}{I}$ is the Chanaclem tic impedance of the arena. Fox the characteristic impedance to be uniform. IN it recellary that the ratio $\frac{V}{I}$ be independent of A .
Before $V$ and I can be Calculated We mut determine the nature of the electric and magnetic ficidu existing in the space between the conducting canes.
Although the biconical Prommustion line can iuppont an infinite number of tranimionios
 Nranmistion mode. 13 present rae the TEM mode, bolt $E$ and $H$ are entirely Aransveases they hove no radial component. The $E$ tines are along greatGrates pasting through the polar axis. this satisfies the boundary conditions since $E$ is normal to the surface of the cones. The $H$ lines are circles lying in planes noxmat to the polar axis

Maxwells equation t from Faraday's las for hamorucats Varying fields is

$$
\begin{equation*}
\nabla \times E=-j \omega \mu \mu \tag{1}
\end{equation*}
$$

Expanding (1) in spherical coosclinates, we have $\nabla \times E=\frac{a}{s^{2} \sin \theta}\left[\frac{\partial\left(N \sin \theta E_{\phi}\right)}{\partial \theta}-\frac{\partial\left(r t_{\theta}\right)}{\partial \epsilon}\right]$

$$
+\frac{a_{\theta}}{x \sin \theta}\left[\frac{\partial E_{x}}{\partial \phi}-\frac{\partial\left(n \sin \theta c_{\rho}\right)}{\partial x}\right]+\frac{a \beta}{\pi}\left[\frac{\partial\left(x E_{\theta}\right)}{\partial x}-\frac{\partial t_{s}}{\partial \theta}\right]-(2)
$$

E has only a $\theta$ smponeal, which by symmetry it independent of $q$, only the fifth team of (2) doer not Vanish Thus,

$$
\begin{equation*}
\nabla \times E=\frac{a_{p}}{n} \frac{\partial\left(\lambda \varepsilon_{\theta}\right)}{\partial \lambda} \tag{3}
\end{equation*}
$$

Expanding the aught side of (1) in aphoucar coordinates.

$$
\text { Since } 14 \text { has only a } \phi \text { component, (4) became }
$$

$$
-j \omega \mu H=-j \omega \mu H \not \omega_{p} \quad \ldots(b)
$$

Now equating (3) and (5) we have

$$
\begin{equation*}
\frac{1}{x} \frac{\partial\left(\lambda E_{0}\right)}{\partial x}=-J_{0} p H_{\phi} \tag{6}
\end{equation*}
$$

Paxinetis equation from Anperse law for fawoncally varying fibldu in a noncenducting medium is

$$
\begin{equation*}
\nabla \times H=j \omega \varepsilon E \tag{7}
\end{equation*}
$$

It has anly a of component and $t$ only a $\theta$ Component since $E_{x}=0$ it follows that

$$
\begin{equation*}
\frac{\partial(\sin \theta 11 \theta)}{\partial \theta}=0 \tag{8}
\end{equation*}
$$

which when redured besomu

$$
\begin{equation*}
\frac{\partial(x+y)}{\partial x}=-j \omega t\left(x E_{0}\right) \tag{9}
\end{equation*}
$$

Nows defferentiating (q) bi.k.k a micl intraducing (6). we obkoin a wave equation in $(x H p)^{\prime}$. Thas

$$
\frac{\partial^{2}(x+14)}{\partial x^{2}}=-\alpha^{2} \mu \varepsilon(x \| \phi)
$$

The condetion of (8) sequitra that Ho Vary inverily as the sine of $\theta$.

$$
A \phi \quad \alpha \frac{1}{\sin \theta}
$$

Hence, a solution of $(10)$ which als fuxfils
(il) is

$$
H \frac{i b}{4}=\frac{1}{n \sin \theta} 4=e^{-i \beta n}
$$

Where $\beta=\omega \sqrt{\mu \varepsilon}=\frac{2 \pi}{\lambda}$
This solution sepresents an awgoing havelling wave on the antenina:

$$
\begin{equation*}
L_{B}=Z_{0} H \phi=\frac{\angle 0}{x \sin \theta} 1+e^{-1} \int \pi x \tag{13}
\end{equation*}
$$

Where $z_{2}$ is the intrirasc impedance of the medianEquations (12) and (13) give the variation of the magnetic and electric fields of a TEM outgoing wave In the space between the cased of a toicontcal antenna as function of $\theta$ and $n$, The fields are independent of $\phi$.

The voltage vic) between points I and a on the Cones at a distance $n$ frore the terminal can now be obtained by toking the line integrand of $E_{0}$ axing a great circle between the two points. Thew

The tatar Gurvent $I(n)$ on the lane at $a$ distance o from the temininale $C$ an be obtained by apptining Amperes low. Thar

$$
I(x)=\int_{0}^{2 \pi}+\phi \times \sin \theta d y-2 \pi s+\operatorname{Hos} \sin \theta \quad \text { (17) }
$$

Now substituting is Ho from (12) in 17) give

$$
\begin{equation*}
I(n)=\text { ont He } e^{-i \beta x} \tag{487}
\end{equation*}
$$

The chovackeristic impedance $z_{k}$ af the biconvicat antenna is

$$
Z_{k}=\frac{v(x)}{I(x)}=\frac{Z_{0}}{\pi} \ln \cot \frac{\theta_{n c}}{2}=120 \ln \cot \frac{\theta_{h o}(\Omega)}{2}(19)
$$

$$
2 k=126 \ln \operatorname{cok}\left(\frac{\theta_{n}}{2}\right)
$$

$$
\begin{aligned}
& V(x)=\int_{\theta_{h}}^{\pi-\theta_{\theta}} E_{\theta} n d \theta \ldots(4)
\end{aligned}
$$

$$
\begin{align*}
& \text { or } v(*)=2 z_{0}+t_{0} n-j \beta \frac{\beta n}{}\left(\cot \theta_{n c} / 2\right) \tag{16}
\end{align*}
$$

Trays of point sources

Case J. Two isotropic point sources of same?
amplitude and phase




AE
a distant
the fiend from source 1 is retarded by $\phi$ $\frac{1}{2}$ dicosp, white the freer from sowice is is advanced by $\frac{1}{2}$ oh $\cos \phi$, where $\alpha_{i} 5_{0}$ the distance between the somites expressed in malians that is $\alpha x=\left(\frac{2 \pi}{7}\right) \alpha=\beta \alpha \alpha$ (pals difference)
The total field at a large distance $A$ in 'the direction $\phi$ is then

$$
\begin{equation*}
E=i_{0} e^{-i \psi / a}+E_{0} e^{j x i 2} \tag{1}
\end{equation*}
$$

Whee $y=d_{1} \cos \phi$ and the amplitude af the E.

Equation (1) may be written a

$$
E=2 e_{0} \cdot \frac{e^{j y / 2}+e^{-i \psi / 2}}{2}
$$

$$
=2 E_{1} \cos \frac{y_{0}}{}-2 \varepsilon_{0} \cos \left(\frac{\alpha}{2} \cos \phi\right) \text { (3) }
$$

We note that the phase of the lookout freed toe not change is a function of $\varphi$.
If $2 E_{0}=1, \quad t=\frac{\lambda}{2}$ then $\alpha_{2}=0$ and (3) becomes
E. $\cos \left(\frac{\pi}{2} \cos \phi\right) \quad \cdots$ (4)


The field from sow se I Ts taken of reference The firs from source 2 in the direction $p$ is advanced by dscosp. Thus. the total fred d $E$ at a distance $x$ is the vector sam of the two foetors from the two sources as given by

$$
E=E_{E}+E \cdot e^{+j \psi}
$$

where $x=d_{n} \cos \phi$

$$
\therefore \quad E=2 t_{0} \cos \frac{\psi}{a}=2 c_{0} \cos \left(\frac{b \cdot \cos \phi}{2}\right) \cdots(6)
$$

The prose of the total field $E$ is. not constant in this case but $\frac{p}{2}$
as $E \times E \cdot\left(1+e^{j \psi}\right)=2 E \cdot e^{j x / 2} \frac{\cos x}{2}:$ (7)
Case II. Two Isotropic point sources of same Amplitude but opposite phase
The lolat field at a large olithance $x$ in the direction of $\phi$ is given os

$$
\begin{align*}
& E=E_{0} e^{j \mu / 2}-E_{n} e^{j \psi / a}  \tag{8}\\
& =2 j t_{0} \operatorname{bin} \frac{y}{2}=2 j \epsilon_{9} \sin \left(\frac{d n}{2} \cos \phi\right)  \tag{9}\\
& \text { If } 2 L_{0} \rightarrow 1 \text { and } d=\frac{\lambda}{2} \text {, } \\
& \text { then } E=\sin \left(\frac{\pi}{2} 6 \pi p\right) \cdot 2^{2},(10)
\end{align*}
$$

The directions $\sigma_{m}$ of maximum fiesta are obtained by setting the argument of (la) equal to $\pm(2 k+1) \frac{\pi}{-2}$ Thus.

$$
\frac{\pi}{2} \cos \phi_{m}=t(2 k+1) \frac{\pi}{2}
$$

Where $k=0,1,2,3 \ldots$ for $k=0, \cos \beta_{m}= \pm 1$ and $\phi_{m}=o^{\circ}$ and $180^{\circ}$

The ruts diredions $p$, Ane given as

$$
\begin{align*}
& \frac{3}{2} \cos \phi_{0}=+k_{\text {g }}  \tag{1}\\
& \text { For } K=0, \quad \phi_{0}= \pm q_{0}^{*}
\end{align*}
$$

The halp-power points (directions) are given by

(Relative fix patten fox a riolrepic Point Solace of the same completude and opposite phase, tpaced $\frac{\lambda}{2}$ apart)
Case IV. Two Idtoknopic point sources of the Siume amplituole but in phase quadrature

Les the two point sources be lrcateol os in Fig. 1. Taking the origin of the condinater of the reference for phase let he urge 1 bo retarded by $46^{\circ}$ and source a be ablvanted by $45^{\circ}$ then the total field in the $\phi$ direction at a large distance o th given by

$$
\mathrm{E}=\mathrm{E}_{a} \exp \left[+j\left(\frac{\cos \cos \phi}{2}+\frac{\pi}{4}\right)\right]+\mathrm{L}_{2} \exp \left[-j\left(\frac{\sin \cos \hat{4}}{2}+\frac{\pi}{y}\right)\right]
$$

$$
\begin{equation*}
=2 E \cdot \cos \left(\frac{v}{4}+\frac{d n}{2} \cos \phi\right) \tag{13}
\end{equation*}
$$

If $2 E_{0}=1$ and $d=\frac{7}{2}$,
then

$$
E=\cos \left(\frac{\pi}{4}+\frac{\pi}{2} \cos \phi\right) \quad(14)
$$



The directions $\phi_{m}$ of maximum field are obtained by setting the argument of (14) equal to kT, where $k$ - $0,1,2,3$... So we obtain

$$
\begin{equation*}
\frac{\pi}{4}+\frac{\pi}{2} \cos \rho_{m}=k \pi \tag{15}
\end{equation*}
$$

for $k=0$,

$$
\frac{\pi}{2} \cos \phi_{m}=-\frac{\pi}{4} \quad \cdots(16)
$$

and $\phi_{n}=120^{\circ}$ and $2 y^{\circ}$
If $\alpha=\frac{\lambda}{4}$, then

$$
\begin{equation*}
E=\cos \left(\frac{\pi}{4}+\frac{\pi}{4} \cos \phi\right) \tag{18}
\end{equation*}
$$

It is a Cardioide-shaped, unidirectional patken with maximum field in the negative $x$ direction The space pattern as a figure of revolution of the $60^{\circ}$ pattern around the $x$ axis


phases diagrams we yer.

$$
\begin{aligned}
& E=E \cdot \sqrt{(1+a \cos x)^{2}+a^{2} \sin ^{2} x} \\
& \text { phase angle }=\operatorname{Lan}^{-1}\left(\frac{a \sin x}{1+a \cos y}\right)
\end{aligned}
$$

Linear Arrays of in Lxokxapit point sources of Equal Amplituole and spacing


Arrangement of bingen Atony of $n$ Iselsepic point sources
The tokal field at a large distance in the direction $\phi$ is given by

$$
E=1+e^{j \psi}+e^{j 2 \psi}+e^{j 3^{\psi}}+\cdots+e^{j(n-1)} \psi \cdots \text { (s) }
$$

where $\psi$ th the total phase difference of the fields from adjacent sources as given by

$$
\begin{equation*}
\psi=\frac{2 \pi}{\lambda} d \cos \phi+\delta>\alpha \cdot \cos \phi+\delta \tag{2}
\end{equation*}
$$

whence $\delta$ is the phase difference of adjacent sources, ie. sownce \& w.a.k. I. 3 w.s.k a, etc.

The amplitudes of the fret od from the sources are all equal and taken as unity. source if it the phase reference. Thus, at a distant point in the direction $\phi$ the field from source \& is advanced in phase w.s.t rounce I by $y$, the field from source 3 is advanced in phase wi ant bounce i by dy etc

Malt plying (1) by $\epsilon^{5 y}$, we get,

$$
\begin{equation*}
E_{e}^{j \psi}=e^{j y}+e^{j 2 \psi}+e^{j} \partial y+\cdots+e^{j n \psi} . \tag{3}
\end{equation*}
$$

Now butatracking (3) from (1) and dividing by 1-ej\%, we gat

$$
\begin{aligned}
E & =\frac{1-e^{j n y}}{1-e^{j \psi}} \cdot(4) \\
& =\frac{e^{j / \% / 2}}{e^{j y / 2}}\left(\frac{e^{j n y / 2}-e^{-j n / / 2}}{e^{j / 2 / 2}-e^{-j \psi / 2}}\right)
\end{aligned}
$$

from which

$$
E=e^{j *} \frac{\left(\sin \frac{n y}{2}\right)}{\left(\sin \frac{\psi}{2}\right)}=\frac{\sin \left(\frac{n y}{2}\right)<3 .(6)}{\sin \left(\frac{y}{2}\right)}
$$

Where is is referred to the field from source t. The value of $\hat{3}$ it Given by

$$
\begin{equation*}
\delta=\frac{n-1}{2} \psi \tag{7}
\end{equation*}
$$

If the phase is referee to the centexpaint of the array, (6) becomes

$$
\begin{equation*}
E=\frac{\sin (x y / 2)}{\sin (y / 2)} \tag{8}
\end{equation*}
$$

In this case tho phase patten is a step function as given by the sign of (z) The Phase of the fred is constant wherever $E$ has a value but Changes inge when $E$ goes though Zero.
when $y=0,(6)$ on (8) is indeterminate so thatfor this case E must be obtained os the limit of (8) as $y$ approaches zero. Thus for $y=a$ see haw the relation that

$$
\begin{equation*}
E-n \tag{9}
\end{equation*}
$$

This is the maximum value of that $E$ con attain Hence the value of the kos tor field. for fax $=n$ is

$$
\begin{equation*}
E=\frac{1}{n} \frac{\sin (n / / 2)}{\sin (y / 2)} \tag{10}
\end{equation*}
$$

The freud os given by ilo will be deferred to as - array factor:
ease 1. Broadside Array (sources in phase)
The first case is a linear annoy of $n$ isotropic bounces of the some amplitude and phase. Therefone, $\delta=0$ and $\psi=\alpha_{0}$ cis $\quad$ a. (v)

To make $y=0$ requires that $y=(5 k+1) \frac{\pi}{a}$ where $k$ is $0,1,2,3$ The freed is therefore. a maximum when $\phi=\frac{\pi}{2}$ and $\frac{3 \pi}{a} \cdots$ (111a)

That iss the maximum field is in a otrection noma to the array Hence, this Condition. Which is Characterized by in-phase sower $(\delta=0)$ results in a "broadside" type of away.
Case 2. Ordinary End-Tive Array.
Let us now finch the prose angle between adjacent bow ice that is required to make the field a maximum in the direction of the aron y $(\bar{i}=0)$. An anray of this type may be caked an "end-fre" array. For this we substitute the conditions $k=0$ and $\phi=0 \quad 7 \mathrm{c} \circ$ (2), from which

$$
\begin{equation*}
\delta=-d a \tag{12}
\end{equation*}
$$

Hence, for an end-fire array, the phase between bounces to retarded progresivvely by the tame amoursas the spacing between sources in sadiron.
Case 3 End-ifise Arrange with Increased Dinectovity
The situation in case 2, namely. fox $\delta=-\alpha$. produces a maximum fred in the drivection $\phi=0$ but does not give the maximum directivity. A langer directivity is abstained by increasing the prove change berwean sources so that

$$
\begin{equation*}
\delta=-\left(\alpha_{n}+\frac{\pi}{n}\right) \tag{13}
\end{equation*}
$$

This condition WEll be refereed to os the Condition for Brinareased deredivity. This for the pharedifievenc.

If the fields at a lane distance we have

$$
\begin{equation*}
\psi=d x(\cos \phi-1)-\frac{\pi}{n} \tag{19}
\end{equation*}
$$

To realize the directivity increase afforded by the additional phase difference requires that $|y|$ be restricted in its range to a value of $\frac{\pi}{n}$ at $\phi=0$ and a value in the vicinity of II at $\phi=180^{\circ}$ This can be fulfilled if the spacing is reduced for example

The maximum of the field patten occur at $\phi=t$ and $p=\frac{-\pi}{n}$. In general, any in aheabed dirgectivity end-fire array, with maximum at $y=\frac{-\pi}{n}$, has a normalized fielol pattern given by e

$$
E=\sin \left(\frac{\pi}{2 n}\right) \frac{\sin (n x / 2)}{\sin (x / 2)} \cdots(15)
$$

Eave 4. Array with Maximum Fielder in ad
Abiknary Direction Scanning ANoa
Let us Consider the case of an array with a field pattern having a maximum in tome? arbitrary otivection of, not equal to $\frac{k w}{4}$, where $k=a$ $1,2,3$. Then (2) becomes

$$
\begin{equation*}
0=d_{n} \cos +\phi_{1}+\delta \tag{16}
\end{equation*}
$$

By specifying the spacing, dols, the required Phase difference \& is then determined by (16) cinveruety, by changing $\delta$ the beam direction $\phi$, Can be shifted or scanned.

Nux Diveckion For Arvays of in Isokropic paide Sourice of Equat Amplituole inot spouing

The num directions for on axbyy of in ssotage point sounce of equal ampirkude and ipacting occon When $E=0$ ow. provided that the denominatox of is G\% not Zeno, where

$$
\begin{equation*}
e^{\operatorname{jin} \psi}=1 \tag{1}
\end{equation*}
$$

Equation (1) kequitro that

$$
\begin{equation*}
n p= \pm 2 k \pi \tag{2}
\end{equation*}
$$

where $k-1,2,3 \ldots$
Sabstituting the value of $\psi$ in (2) to (2), we get

$$
\begin{align*}
& \psi=\omega_{2} \cos \phi+\delta= \pm \frac{2 k \pi}{n}  \tag{3}\\
\therefore & \phi 0=\cos ^{-1}\left[\left(\frac{1}{n} \frac{2 k \pi}{n}-\delta\right) \cdot \frac{1}{d r}\right] \tag{4}
\end{align*}
$$

where $p_{0}$ gives the disection of the pattern nush whe note that values of $k$ must be excludeof far which $k=m$ men, where $n=1,2,2 \ldots$. Thas, ifi $k=$ mon, 2) Neduccs to $\psi- \pm 2 m \overline{4}$ and the denomsinatar of $[4]$ equas zero th that the nul direction of (1) that the numenaton of $\frac{1-\varepsilon \text { finy be zere in insufficier }}{1-e j y}$ by

Irr- a broadside array $\delta=0$, do that (y) be comes

$$
\begin{equation*}
\phi=\cos ^{-1}\left( \pm \frac{2 k \pi}{n d r}\right)=\cos ^{-1}\left(\frac{1 \cdot \lambda}{n d}\right) \tag{5}
\end{equation*}
$$

If $\&$ in (3) is replocid by sks complementan angle for then ( 5 ) becomes

$$
\gamma_{0}=\sin ^{-1}\left(\frac{+k \lambda}{n d}\right) \quad \ldots(b)
$$

If the arity is long, 10 lhat nis $\gg k x$,

$$
f_{0}=\frac{k \lambda}{n d} \quad \therefore(7)
$$

The first nus u either side of the maximum occur for $k=1$. These angle i will be designated for. Thus,

$$
\begin{equation*}
f_{0}= \pm \frac{\lambda}{n d} \tag{8}
\end{equation*}
$$

and. the total beam wiolst of the main lobe between first nus for a long broadtiole array is then

$$
\begin{equation*}
2 / 01 \approx \frac{2 \lambda}{n \alpha} \tag{9}
\end{equation*}
$$

For ordinary end-fire arrays, $\delta=-\alpha$. Thad.
(3) becomes

$$
\begin{equation*}
\cos k_{0}-1= \pm \frac{2 k \pi}{n d} \tag{19}
\end{equation*}
$$

from which we obtain

$$
\begin{align*}
& \frac{\beta}{2}=\sin ^{-1}\left( \pm \frac{\left.\sqrt{\frac{k \pi}{n d x}}\right)}{\ldots}\right. \text {.. (1) } \\
& \text { ar } A=2 \sin ^{-1}\left( \pm \sqrt{\frac{k \pi}{n \alpha_{x}}}\right) \cdots(12 \tag{12}
\end{align*}
$$

If the away is long, to that nd $\gg k \lambda$. (ia) becomes

$$
\begin{equation*}
\phi .= \pm \sqrt{\frac{2 k \pi}{n d}} \tag{13}
\end{equation*}
$$

The first nus either side of the main lobe socle for $k=1$. There angles will be designated $\phi_{1}$. ThaN.

$$
\begin{equation*}
\phi_{01} \div \pm \sqrt{\frac{2 \lambda}{n d t}} \tag{14}
\end{equation*}
$$

and the total beam width of the main lobe between first nubs fox a Long andinary end- fire array is then

$$
\begin{equation*}
2 \phi_{01} \approx 2 \sqrt{\frac{2 \lambda}{n \alpha}} \tag{15}
\end{equation*}
$$

For end-fire arrays with increased directivity, the condition is that $\delta=-\left(d_{n}+\frac{\pi}{n}\right)$. These for this cone
(3) become l

$$
\text { dx }\left(\cos \phi_{0}-1\right)-\frac{\pi}{n}= \pm 2 \frac{k \pi}{n} \ldots(16)
$$

from which

$$
\frac{\phi_{0}}{2}=\sin ^{-1}\left[ \pm \sqrt{\frac{\pi}{2 \alpha_{x}}(2 k-1)}\right] \ldots(17)
$$

os $\phi_{0}=2 \sin ^{-1}\left[ \pm \sqrt{\frac{\lambda}{4 n \alpha}(2 k-1)}\right] \cdots(18)$
If the array is long, so that nod $\ggg 1 / \mathrm{g})$
becomes

$$
\text { fo } \quad \pm \pm \sqrt{\frac{\lambda}{n d}(2 k-1)} \ldots \text { (19) }
$$

The first nulls either side of the main lobe, $\phi_{0}$. occur for $k=1$. This.

$$
\phi_{01} \approx \pm \sqrt{\frac{\lambda}{n d}} \quad(20)
$$

and the total beam width of the main lobe between first null for a long end-fire array With increased directivity is then

$$
\begin{equation*}
2 \phi_{01} \simeq 2 \sqrt{\frac{\lambda}{n o t}} \tag{21}
\end{equation*}
$$

This width is $\left(\frac{1}{\sqrt{2}}\right)$ os $71 \%$ of the width of the ordinary end-fire array.

The helix is a basic 3-dimensional geometric form 1 helical wise on a uniform cylinder becomes a straight wire when unwound by ailing the cylinder on a flat surface viewed and-on a helix projects as a circle. Thus a helix Combines the geometric form of a starlight line a bride and a cylinder. In addition a felt has handechess; it can be either left an
right handed.

The following symbols will be wed to $\begin{aligned} & \text { describe a helix. } \\ & D=\text { diameter of helix (cento to center) }\end{aligned}$ $C=$ circumference of helix $=\pi D$ $S=$ spacing between turn (center to center) $\alpha$ pitch angle $=\tan ^{-1} \frac{s}{\pi D}$
$l=$ length of 1 then
$n=$ number of turns
$A$ - axial Length -ns
$d$ = diameter of helix conductor
(Helix and associated dimensions)
The equation of a helix is

$$
\begin{aligned}
& x=a \cos t \\
& y=a \sin t \\
& z=b t
\end{aligned}
$$

A subscript $\lambda^{x}$ signifies that she dimension is measured in free-tpace wavelengths for example, $D_{x}$ is the helix
diamexen in free-space wavelungthe
If 1 toun of a criculas hewx is censoked on a flat plane, the relation between the ipacing 6. cerciom fenence $c$ tum kengtri $i$ and pitch angle $\alpha$ soue at tlluster ated by the following triangle.


When $b, a, x=0$ and the fetix becrmes a bopp. an the ather hand, when the oliameter is zere, rti= 90 The helix becames a linear Ginducton.

The temm tmandristion mode it wed to describe the mancies. in which an electromagretic wave is prapagated along an infirite nelix as thowegn the helix Conytituter an infintre tharsmistion line as wavegwide. A. variety of different thantmidion modes are poikible.

The term radiation mode is wsed to olercribe the gervenal fosm of the fan-lidel pattes of a firite hetical ontenna fuhough mary patterne, are posible, two kinde of are of particulan rinterest oneis the axial las beam) mode of madiation ( $R$, mode, berm on the $a \times i$ ) and the okner is the nosmal mode of radiation ( $R$ mode, radialion maximuin perpendicular to a为.

The lowest tranimision mode for a hetis has adjarent regions of poritive and regative change sepanated by mary twon. This mode $\mathrm{F}_{1}$ designed as the To $i$ varumision mode anol the instantaneous Chavge atstribution at thown in the following figuse.
 E \ll $\lambda$
(2)

(c)


The To moote is Tmportont when the lengti af l laint it smale companed to the wavelongth $(1 \ll ~ \%)$ and is The mode sccuring on low- frequency inductancesIt is also the impostant dxangmision mocse in the traveling wave tabe since the adjacent segions of Poxitive and negative change are beparated by an appreciable axiak clistance a subdtential axial Component of the electric field is pruent. and in the lhavelling wave tube this field interacts with the alection stream. if the carterion $L_{x}<\frac{1}{6}$. is arbitrarily belected as a boundary for the To thanmision mode, the region of the heter dimensions for wheci this mode it impostary is within the Tike area.
A helix excited in lhe.: To thamsmiskion mode
may hadiate Let w consider.. the case then the, fielix is Very thont ( $n L \ll \lambda$ ) and the curnentis aslumeol to be of urviforum magnetude and in phate along the Entise hetix. It is thearelically posribie lo approsinas this congetian on a imaut end-loaded hodix
Howerens its sadiation sebiktence is ismapl the, moximum fietd frown the helex is normal to the hetex axis fox au helix dimention provided only that $A L \ll \lambda$. Thut that Gonctition is $E$ alled on "ncomal radiation mode" (Ro). Any cemponext of the fietel, ha a sine variation with 0

## Axidur Motle patkesns and the phoore vekoing of Wrave, Prispagution on tronofilar teltes I:

As a first approximation, a monofilan herical antema sodiating in the axtal mode may be asumed to have a single trave ling woive of uniform amplieude ateng its conctuctan the for fretd patiex of a helex is the produck of The patken for i twan and the pattern foy an soucy of $n$ isotropic point sountes. The nuriten. $n$ equals the number of twom the spating s botwsen sowres is equal to the tian speeing,

When the helix is lang $\left(n s_{2} 71\right), 17$ belay partan is much sharper than the single turn patter and hence largely detamines the shape of the collat fan-fierd patter. Hence, the approximate fow-fierd pattern of a long kelix is given by the arran patters.

$$
\text { The tray proton on array factor } E \text { for on }
$$

$$
\text { array of } n \text { isotropic point sounces arranged as in }
$$

the following figure is given as

$$
E=\frac{\sin (\operatorname{sic} / 2)}{\sin (y / 2)}
$$

$$
:-\frac{4}{4} \mathrm{k}
$$

Where $n=$ number of rounce and

$$
\begin{array}{r}
\varphi=\delta_{x} \cos \phi+\delta \\
\text { Where } s_{k}-\left(\frac{s^{p}}{\gamma}\right) s
\end{array}
$$

for the present case, (2) becomes

$$
\psi=2 \pi\left(s_{x} \cos \phi-\frac{L_{x}}{p}\right)=(+(3)
$$

where $p=\frac{u}{c}$ - relative phase veleciky of wave propagation along the helical conductor, it being the phase velocity along the helical conduction and $c$ being the velocity of hight in free pace If the fields from all sources are in phase at o point on the helix $a x i=$ ( $\phi=0$ ). The radiation will bo in the axial made. For the fiuds bo be in phase (andine End fire rEndition) requires that

$$
\begin{equation*}
y=-2 \pi m \tag{4}
\end{equation*}
$$

$$
\text { where } m=0,1,2,3
$$

That the minces sign in (4) mesults (ram the pack w. r. source 1. Source $3^{\text {is }}$ timifandy retarded to. rat source 2 etc.

$$
\begin{aligned}
& =\text { Not putting } \phi=0 \text { and equating (s) sind (a), } \\
& \text { tue have } \\
& \frac{L_{x}}{\beta}=S_{x}+m \cdots(5)
\end{aligned}
$$

when $m=1$ anot $p=1$, we have the relation

$$
\begin{equation*}
\varepsilon_{x}-s_{\lambda}=1 \quad \text { or } \quad L-\xi=\lambda \tag{6}
\end{equation*}
$$

This is an approximate relation between the kwan length mid pacing requited for a helix thadrating in the axial mode Since for a her ix $\left(2-(i 0)^{2}+s^{2}\right.$, (6) (an ks e tewnetten as

$$
\begin{equation*}
D_{\lambda}=\frac{\sqrt{2 s_{\lambda}+1}}{\pi} \text { ax } c_{\lambda}=\sqrt{2 s_{\lambda}+1} \tag{7}
\end{equation*}
$$

When " $m=1$ and solving (5) for p. we hare

$$
\begin{equation*}
p=\frac{L_{\lambda}}{\delta_{\lambda}+1} \tag{8}
\end{equation*}
$$

(8) Can abs be expressed as

$$
\begin{equation*}
\phi=\frac{1}{\sin \alpha+\left[\cos \alpha / c_{\lambda}\right]} \tag{9}
\end{equation*}
$$

Equation (q) gives the required variation the relative phase velocity $p$ on a function of the circumference $c_{r}$ for in-phase fiats in the axial direction.

If the increased ditectivity condition is Presumed to Exist, (4) becomes

$$
y+=-\left(2 \pi r+\frac{\pi}{n}\right) \cdots(18)
$$

Now Equating (ic) and (3). putting $\phi=0$ and boring for $p$ we have

$$
\begin{equation*}
p=\frac{L_{n}}{S_{n}+m+(1 / 2 n)} \tag{iv}
\end{equation*}
$$

Fax the case of interact $m=1$ and

$$
\begin{equation*}
p=\frac{c_{7}}{s_{n}+[(2 n+1) / s n]} \tag{iv}
\end{equation*}
$$

Fox large values of $n$, (is) reduces 10 (a) Equation (iv) Cans also be
expressed as

$$
p=\frac{1}{\sin \alpha+(2 n+1)\left(\cos \alpha / c_{n}\right)} \ldots(, 13)
$$

The increased directivity Condition ss approximated as a natural condition on helices radiating in the axial mode.

Let is procecol to develop the fas-fietel patterns from a single turn of a monofilas helix Radiating in the axial mode. It is ousumed thou the tingle kwan has a uniform traveling wave along its entire length The product of the single-fium patton and the orroyg factor then giver the total helix patrean.
A Creventan helix may be treated approximately by assuming that it is of square Enos section. The total fred from a ingle turn it then the rowerant of the frelols of four short, linear antennas.


A helix of square cross section. can of course be kreakecl exactly by thin method Measurements indicate thou the difference between helices of eirluta carol square cross section Sb anal


Referring to the figlare below, the for electric pred Components EsT and tor. in the az plane as a function of $\beta$ for a single-twon helix $x$.

Let the area of the square helix be equal so that of the Givactan helix so thou-

$$
g=\frac{\sqrt{7} D}{2} \quad \frac{\pi D^{2}}{4}-g^{2}
$$

Where of is at shown.
The for magnetic field for a linear elementwith a uniform traveling wave is given by

$$
\begin{aligned}
& |H y|=\frac{J_{\Delta p}}{2 \pi r,}\left\{\frac{\sin \phi}{1-p \cos \phi} \text { sin }\left[\frac{\omega b}{2 \rho c}(1-p \cos \phi)\right] 3 \mathrm{t}\right. \\
& \text { Phase angle } \tan ^{-1}\left[\omega^{2}\left(k-\frac{x_{1}}{c}\right)=\frac{\alpha b}{\text { apc }}(1-\text { pest })\right]
\end{aligned}
$$

Let w multiply if by $z$ of free face and put $\gamma=\frac{8 r}{4}+\alpha+\beta, t=0$, and $b=\frac{g}{C \operatorname{cov}}$, io we obtain the experestion for the of component of Est of the far fred in the 22 plane due 10 element of the square helix at follows:

$$
\begin{aligned}
E_{\phi 1} & =k \frac{\sin y}{A} \sin B A<\left(6\left(-\frac{w \pi}{c}-B A\right) \cdots(2)\right. \\
\text { where } k & =\frac{I_{e p r}}{2 \pi r i} \\
A & =1-p \operatorname{cosy} \\
B & =\frac{a y}{2 p c \cos )}
\end{aligned}
$$

The expression for Eta, Eft etc Due to elements 2, 3 and 4 of the square tics are olotaineot in a similar way. once the elements are ats disifmilar sources. the total $\phi$ component. EAt from a single square twin is obtained by adding the frats from the fou elements at each angle $p$ for which the total field is coplculared. the lam of the fields -From the fou clements is then

$$
\begin{aligned}
& E_{\phi T}=k \frac{\sin f}{A} \sin B A /\left(-B A-\frac{\omega r_{1}}{C}\right) \\
& +K \frac{\sin B A^{*} \text { sind } \sin \phi}{A^{\nu}} /\left[-B A^{*}-\frac{2 \operatorname{ta}}{4 p C^{\prime}}+\frac{\omega}{6}\left(\frac{6 \cos \phi}{4}+g \sin \phi-x_{1}\right)\right] \\
& +k \frac{\sin f^{\prime}}{A^{\prime}} \sin B A^{\prime} /\left[-B A^{\prime}-\frac{L \omega}{\alpha p c}+\frac{\omega}{c}\left(\frac{\delta \operatorname{cob} \varphi}{2}+g \sin \phi-\alpha_{1}\right)\right]
\end{aligned}
$$

$$
+k \frac{\sin B A^{\prime \prime} \sin \alpha \sin \phi}{A^{\circ}}\left[\left(C^{\circ} A^{\prime \prime}-\frac{3 L \omega}{4 P^{\circ}}+\frac{\omega}{C}\left(\frac{3 \sin \phi}{4}-x_{1}\right)\right]\right.
$$

Where $\gamma=\frac{8 \pi}{2}+\alpha+\phi \quad \gamma^{\prime}=\frac{\pi}{2}-\alpha+\phi$

$$
\gamma^{\prime \prime}=\cos ^{-1}(\sin \alpha \cos \phi)
$$

$$
\left.A=1-p \cos y, \quad A^{\prime}=1-p \cos \right]^{\prime \prime}, \quad A^{*}=1 \text {-pony }{ }^{*}
$$

When a helix of executor cons section is being calculated $t=\bar{D} / \cos \alpha$ in (3), while for a helix of square cross section $L=46$. A

If the contribution of elements 2 and 4 avo neglected. Which is a good approximation When both $\alpha$ and $\phi$ are small. The expression for E pr is Considerably simplified Making this approximation. Sexing $k-1$ and $r_{1}=$ constant i? we clotain

$$
\begin{align*}
E_{\phi T} & =\frac{\sin f}{A} \sin F A(C-B A) \\
& +\frac{\sin \gamma^{\prime}}{A^{\prime}} \sin B A<\left(C-13 A^{\prime}-2 \sqrt{B i} B+\pi\left(5, \cos \phi+\pi D x^{\sin \theta}\right)\right. \tag{4}
\end{align*}
$$

Equation (4) applies isperificatly ho helices of circular cross lection, no that is in (4) is

$$
\begin{equation*}
B=\frac{D_{\lambda} \pi^{2 / 2}}{2 p \cos \alpha} \tag{5}
\end{equation*}
$$

Equation (4) give the approximate pritern of the $\phi$ component of the far field in the $x z$ plane from a single-twen helix of circular Gob section-

In the case of the e component of the fan feted in the $x z$ plane only elements: i and 4 contribute putting $x=b$, the magnitude of the approximate $e$ parser of the foo foetal of a singlettun helix of eirculas enacts) section is

$$
\begin{aligned}
\left|E_{e r}\right|= & 2 \frac{\sin f^{\prime} \sin B A \cdot \cos \alpha}{A^{4}\left(1-\sin ^{2} \alpha \cos \alpha \phi\right){ }^{1 / 2}} \\
& \left.\sin \frac{1}{2}[2(\delta, \cos \phi-\sqrt{\pi D} D \sin \phi)-2 \sqrt{\pi} B] \cdots \text { ( } 6\right)
\end{aligned}
$$

Let wo proceed to find the cred rarticued by a traveling wave on a thin linear conductors. Let ah consider a: conductors of length $b$ coincidentWith the $z$ axis and with one end at the origin of a. Cylindrical coordinate system $(r, \phi, z)$ as in the following figure. It is assumed that a single, uniform traveling wave is moving to the sightalong the inductor


Since the current is entirely in the $z$ direction. the magnetic field has bu' one component Ho. The $\phi$ dimpction is normal to the page at $p$ and its positive sense is outward from the page. The magnetic field Hg can $\mathrm{te}_{\mathrm{e}}$ obtained from the Hertz Vector $\pi$. Since the current is entirely in the $z$ direction, the Hertz vector has only a $z$ component Then.

$$
H_{\phi}=j \omega \varepsilon(\nabla \times \pi)_{p}=-j \omega \varepsilon \frac{\partial \pi_{2}}{\partial \pi} \cdots(1)
$$

where $\pi_{r}$ is the $z$ component of the retarded
Hertz vector at the print $p$, as given by

$$
\Pi_{2}-\frac{1}{4 \pi j \omega \varepsilon} j_{0}^{b} \frac{[t]}{n} d z_{1}
$$

$$
\begin{equation*}
\text { where }[I]=I \cdot \sin \omega\left(t-\frac{\pi}{c}-\frac{z_{1}}{u}\right) \tag{3}
\end{equation*}
$$

$$
\begin{aligned}
& \text { Where } z_{1} \text { a pine on the conductor } \\
& \text { and } y=p c \text { on } p=\frac{k}{c} \text { (4) }
\end{aligned}
$$

In (4), $p$ is the rater प्य the velocity along the conductor is to the velocity of light c. Flit ratio will ike called the molalive phase velocity.

Thus.

$$
H_{\phi}=\frac{I \cdot c}{4 \pi} \frac{\partial}{\partial-\frac{1}{4}} \int_{h_{1}}^{40} \frac{\sin \omega \omega_{0} \theta}{2-(\hat{y})} d_{i x}
$$

$$
\because 11 \quad \therefore 1+\varnothing \text { and } 1
$$

At arbitrarily large distances, the
instant the ne
and for the. case therese

$$
\text { - } \quad \sin \omega k_{2}+\sin \text { wild, } f 0 \text {, with }
$$

$$
\begin{aligned}
& \text { we howe a. +aus }
\end{aligned}
$$

$$
\begin{aligned}
& \text { where } u_{1}=t-\frac{r_{1}}{c} \text {. } \\
& \text { and } u_{2}=t-\frac{s_{2}}{c}-\frac{b}{v} \rightarrow \text { (q) }
\end{aligned}
$$

$$
\begin{equation*}
H \phi=\frac{I_{0}}{4 \pi=} \frac{\sin \phi \theta}{\cos \theta-(1 / \rho)}\left(\sin \omega u_{2}-\sin \omega u_{1}\right) \tag{13}
\end{equation*}
$$

Where the relations have been introduced for $r \gg b$ so that

$$
\begin{aligned}
& \frac{z}{r}=\cos \theta \quad \text { and } \frac{p}{r}=\sin \theta \ldots(14) \\
H \phi & =\frac{I_{0} p}{2 \pi-1}\left\{\frac{\sin \phi}{1-\mu \cos \theta} \sin \left[\frac{\omega / 1}{2 p c}(1-p \cos \theta)\right]\right\} \\
& {\left[\left[\omega\left(t-\frac{r_{4}}{c}\right)-\frac{\omega \cdot 6}{2 p c}(1-p \cos \theta)\right] \ldots\right.}
\end{aligned}
$$

Fonefilor and Musifilan Noxmat-Mode Telical Antennos

Normat is being whed in the rense of perpendicutar to os at right angles to the helix axis.
witen the thelix uxcumplererce it approxisiately a wavelength the axial made of tadiation is doncunart, but when the tiskempenence is mucl imalla the nesmat mide it domenans

Let in Conider a heix asiented wïth axis Coincident with 2 Gxis




If the shimentions कre sanalu ( $n t\langle i\rangle$ ), the maximum saciation it in the xy plane fou a helix asierted wid in the fgive above with zeve field in the $z$ direction stien the pitch engle is zene, the hetax becomes a keop when the pilch angle is qo". The herix straighrev out into a timeth Gutenna.

The fre field of a herix riony be described by two componene. of the electur fielal. Eq and tp

Let us now dievelop expartuor fon the tarfift patiexre of theie smperenty fas a imans shein belix. The developmeit to facrictared by ascosing that the helix unnists of a nermber of smou loope sund short dipoler connected in zeren. iple The cument is asauing of to be woufowo (y) in magnitude and phose oven the C+T entime length of the heler. sfince the helix is imau. the far field pattexn is independent of the numbet of twors. Hence it sufficer to couculate the far - fictol patsers of a Lingle imau tnop and one shant dipole.

The fac fietd of a imeth toop hai only in Eq Gompenent It Valuze ir given as

$$
\begin{equation*}
E_{\phi}=\frac{120 \pi^{2}[\pi] \xi \sin \frac{A}{\lambda^{2}}}{力} \tag{1}
\end{equation*}
$$

Whene $A=$ area of the krop $=\frac{\pi D^{2}}{4}$
The far field of the tho en oupols has only en Eo component Its value is givan as

$$
\begin{equation*}
E \theta=j \frac{60 \pi[] \sin \theta}{x} \frac{\xi}{\lambda} \tag{2}
\end{equation*}
$$

Equation (1) and (2) Fndicate that Es anet Es are in phose quadratwe Jo oktain the axial sation we wsite

$$
A R-\frac{\left|E_{\theta}\right|}{\left|E_{\beta}\right|}=\frac{5 \lambda}{3 \pi A}=\frac{25 \lambda}{|\| D|^{2}}=\frac{2 \Delta \lambda}{C_{\lambda}^{2}}
$$

A. monofitan noxmal-mode helite fiwtilung the condition; F. E- IEel EAI giving $A R=$, arid rebulling in ercolon potanzar or (zehe frese in the $x$ dinection) is a rebanant narkew-band anterina. *

CONTINUOUS ARRAYS

Let ins ienstien kontinueous anyways of point sower, i.cu arrays of an infinter number of sones separated by infinitesimal diskances. By Huygens principle, a continuous array of Point sower is equivalent to a centinusm Pied dizkrbwion.

We that now develop an expression for the for field of a Continuous array of point Sourer of equal amplitude and of the tare e Dhase let the uresay of length $\Rightarrow$ be porvallet Lo the $y$ axis with ils Cental as the origin as indicated in the pig we


The field dE af a distant print in the direction of cure to the point sources in the infiruterimal length dy at a dibkimce If from the origin is

$$
d E=\frac{A}{x_{1}} e^{j \beta\left[t-\left(\frac{x_{1}}{c}\right)\right]} d y=\frac{A}{x_{1}} e^{J\left\{\omega t-\beta x_{1}\right\}} / y_{y}
$$

where $\beta=\frac{\omega}{c}=\frac{8 \pi}{\gamma}$ and $A$ it a Gratant involving amplituole The total firetat at the distant point vi s then

$$
\begin{aligned}
E & =\int_{-a / 2}^{0 / 2} \frac{A}{x_{1}} e^{j\left(\pi \Delta t-f \Delta x_{1}\right)} d y \\
& =\frac{A e^{j \omega t}}{x_{1}} \int_{-\alpha / 2}^{q / 2} e^{-j / i x_{1}} d y
\end{aligned}
$$

But $x_{1}=x-y \sin \theta$

$$
\begin{aligned}
& E=A^{\prime} \int_{-D / a} e^{x p y \sin \theta} d y \\
& \text { Where } A^{\prime}=\frac{\left.A e^{j(k t}-B / \omega\right)}{r,}
\end{aligned}
$$

Then,

$$
\begin{aligned}
E & =\frac{2 A^{2}}{\beta \sin \theta} e^{i(j a / 2) \sin \theta}-e^{-j \operatorname{cic} a / 2 / \sin \theta} \\
& =\frac{2 A^{\prime}}{\rho \sin \theta} \sin \left(\frac{\beta e}{2} \sin \theta\right)
\end{aligned}
$$

Let $p^{\prime}=\beta a \sin \theta=$ Us $\sin \theta$
Where da $-\hat{p a}=\frac{2 r}{\lambda} \alpha=$ barony Length, tad
Then

$$
E=\frac{2 A^{\prime}}{p_{1} \sin \theta} \sin \frac{\psi^{\prime}}{2}
$$

Put Bine $=\frac{x^{2}}{a}$

$$
E=a^{\prime} \frac{\sin (x / 2)}{x^{\prime} / 2}
$$

Normalizing if gives

$$
\frac{c}{x}=\frac{\sin (\psi / 2)}{\psi^{\prime} / 2}
$$

This equation expressed the for field, os Fraunhofer diffraction pattern of a centinuow broudride arovily of length a hawing heniform cimphikude and phase Font a desoreke equally Mated sower it wa previously shew that the cormatized value of the kotiol fight it

$$
\begin{aligned}
& \qquad E=\frac{\sin h y(2)}{n \sin (y / 2)} \\
& \text { Where } \psi=\frac{\operatorname{des} x p+\delta}{t}
\end{aligned}
$$

Fer in phase rounded, $\delta=0$.
Thus $\quad \phi=\theta+\frac{\pi}{2}$
 zimale value of for which ofcior for smote values of $\theta \alpha$ or both $E$ can be
expressed as

$$
E=\frac{\sin (n y / 2)}{n y / 2}=\frac{\sin [(\beta n d / 2) \sin \theta]}{(\beta n d / 2) \sin \theta}
$$

The length a of the array of discrete sources is

$$
a=\alpha(n-1)
$$

Where $n$ - number of rounce

$$
d=\text { spacing }
$$

If $n \gg 1, a \approx n d$.

$$
E=\frac{\sin [(\beta \alpha / 2) \sin \theta]}{(\beta \alpha / 2) \sin \theta}=\frac{\sin [(\alpha / 2) \sin \theta]}{(\alpha a / \theta) \sin \theta}
$$

Where $a_{x}=\beta a=\frac{2 \pi}{\lambda} \alpha$

$$
a=\frac{\sin \left(x^{\prime} / 2\right)}{x^{\prime / 2}}
$$

The null directions $\theta_{R}$ of the continuow array pattern are given by

$$
\frac{\psi^{\prime}}{2}= \pm k x
$$

where $k=1,2,3$.
Thew,

$$
\theta .=\sin ^{-1} \pm\left(\frac{k x}{a}\right)
$$

For a long array

$$
\theta_{0}=+\frac{k}{a_{7}}(r-a d) \Rightarrow \pm \frac{57.3 k\left(d_{e g}\right)}{a_{7}}
$$

Where $\quad a_{x}=\frac{a}{\lambda}$
The beam width between first numb $(k=1)$ for a long array is then

$$
2 \theta_{21} \approx \frac{2}{a_{\lambda}} \text { (had) } \approx \frac{115}{a_{\lambda}}\left(d_{\text {eg }}\right)
$$

## Long - Wire Antennas

In the frequenty bancl from = \& 30 MHz lang them (3everak wavelengthe length) thpponted by switable fower may be wsed as efficient antening The but known sypes are the hosizontal $V$ ankenna the horizontat Nhambic antenna. the vertical $v$ and sleping rhombic antennas, the vertical invented $v$ ar half. rhambic anterna, and the single novizontal-wise antenna Most long-wire ankennal Can be operatie as relonant Ankenns, in which case the turrent on the wire witl be a standing wave with the Characteriste in insoichal variation These ankennas lemoth perate sotisfactonty only at a panticut as frequenty and harmenilu of this frequenty The inputs impedance will be highty ferquenily-sensitive, so ony nawow.band optoation in poblible-Mosk lang-wise Qrepentar can aso be oproces as kraweling wrive Ot normesonant stmuckues by lismanating the for end of the wirt in a toitable resistence howing a value equal to the characterstic impedance of the antenna viewed as a trensmistion tine In thit mode of epperation the usefwe frequenily band ban be quike lazge, with an accupirble imprdance match ovo the whele range of frequentio.
for Voriout typu of long-wire antennas are wad for Commerseat shostwavt kraramistion in the Frequency sange frem i to $30 \mathrm{M} H 2$ thee prepagate i) by meart of conospheric heflection Fox thase applectation the optimum angle of vacliation is urually from to lo $30^{\circ}$ Melative to the kurrizontal line in the direction of receving 140kion

Since long-wire antennai are tocaked in the presenle of ground the lattis has an impontiani
effect on the xadration patrenn and must be tiaken impo acceunt th the oletign op anderniog Coneigwiation Iv giveral the design preblem is One Uf obtaining a dinective bram at the dibined angie rolative to the guwnd for pptumum long-distante Gmmunnicution via sefrestion fram the ionolphere, along with accoplable iaputimpiedance charactex usics that kill facilitate miatctuing the antenna to its feect lene. Routiation from a Resonant Long-wise Anterm


Thit long wire axtenna is of Length $l=$ $\frac{m h_{0}}{g}$ weth a curbent distr-bw-ion $f(x)=$, sink $x$ It witl by cenverient fo he the argle y os the polon angle rolative to the diredion of the wire and to expmeri the sadrated slectrat fietd as Ep. The fielet may be found in the sourne mannew at wed for the half-bowe dipale arte nina

$$
\left.E(x)=\frac{\int x-z_{v} e^{-1 k_{e} x}}{4 \pi x} \int_{i}\left[\left(x_{x}-a\right) \alpha_{x}-0\right] I 4\right) e^{j h+t u c} d x
$$

The surt vecton a- $a_{x}$

$$
x^{\prime}+x^{\prime} a_{2}
$$

$$
a_{2}=a_{0} \sin \theta \cos \varphi+a_{e} \text { Ecxetosq-aptinf }
$$

$\theta=q_{0}=\psi=\psi$ (foos the present cose)


## Radiation from a $v$ Antenna



The $v$ antenna shown hoe consists of two Atraight-wike artecrnat aweanged io as to 2wbtend an angle $y$, The radiation it the superposition of that from each straight-wire section the wbjective in the deign of $V$ antenna is to choose the angle $y$.
do at to align the two labs produced by each tinaight-wise section if a maximum in the direction of the $v$ arrcenona and in the plane of the $v$ antenna is arrived. then the optimum value fay the angle $\psi$. it, twice the angle than the radiation lobe make with the wine axis fox a thnaight-ase antenna If each arm of the $V$ antenna is $\frac{3 \text { ) }}{2}$ long, this optimum angle it 2 y": Fox ls $2>\mathrm{c}$, the optimum angle is epprexionatly $72^{\circ}$. The currents On the two arms of the antenna che out of phase. but this is jut the condition required for the electric frets ty radiated by each ain to radiation The resultant maximum labe in bath the antenna will have a forward and backward directions. Smaller minor lobe 0.111 occur in between.

For a haxizontar $x$ antenna durgneal to radiale with a maximum at an angle $y$ restive to the horizontal plane, the angle ye mast be seduced to being the lobe inge alignments That this 52 the colt may be sean by recalling that the pattern from a single vice is a ene of -nneadions to that in ode to bring th Prime

The lang -wire antenna may be fed at one end as shown in the figurer below. Howler. because of the unsymmetrical arrangement the currents in the transmission line will not be balanced, and rome radiation will occur from the feed line itself. A more satisfactory feed arrangement is to connect the trangmastion line ar the center of a current loop os Close to the midpoins of the antennas os possible $A \frac{\lambda}{4}$ transformer can be wed to transform the radiation ruistance to the commonly used $600-\Omega$ two wire transmission line The required impedance of the matching section is given by $z_{c}=\sqrt{600 \mathrm{Ra}}$.

Caincrdences the $v$ angle $y_{0}$ must be reduced An alternositive proceolure is to ulope the $v$ antenna upwarde by the requised elevati on engle of


The discustion above applies in genesal to The shombic snterna atho. The dergin objetine is la cheose the angutre orientation of Each straight wire tection to as to Dlign the a adiation lobs frem the fowe indevidual section in one 6 mm in orrection in adoution the length of each aem must be chesen so a to abtain in-prase addition of the fietd radiated by each straight wire aection Rodiation fien a Lang-wite Anterna With I Pavoling wave Cborent.


If the for end of the 3 traight wite sxterina is lemmenated in a matched sesistance, the chorent daxtrubution on the andenina car be approximated as a Araveling wave of the form $I=I \cdot e^{-j k o z}$. The relultant radiation
patter is given by


$$
=\frac{j t_{0} z_{0}}{4 \pi x} e^{-j k_{0}\left[r+\left(\frac{1}{2}\right)(1-\cos y)\right]} \sin x \frac{\sin \pi x_{0}(\sin 2 y]}{\sin ^{2}\left(\frac{y}{2}\right)}
$$

The lobe maxima are detrained by

$$
\frac{\tan \beta}{\beta}=2-\frac{\alpha \beta}{x_{0} \alpha}
$$

Where $\beta=k_{0} \sin ^{2}\left(\frac{y}{2}\right)$.
for sol>> the solution is $\beta=1.16 \mathrm{E}$. For other
value of vol. kay value can be wed foe $\beta$ in the above equation to obtain a corrected value For $\beta$ A somewhat move realistic approximation for the crarent is $t_{0}$ e $j k \cdot x-\alpha x$ where the atkentration Gritant $\alpha$ accounts pie sanitation loss. From the antenna as the wave propagates along the wine The attenuation factor $\alpha$ is imals ind produces arty a small change in the sadsation patters so we are neglecting this effects

The radiation pattern for the travelling waite antenna is shown in the following figlaen for two different antenna lenglks i. The main feature exhibited by these pattern is a mayon cone of radiation in the forward direction and the absence of a major cont of radiation in the backward direction to the lengit of the antenna is increased. the angle of the major cone decrease. If $t$ - $n x$. there will be $x$ total of $2 n$ liber.

The $V$ antenna can be converted to a traveling-wave antenna by terminating each arm in a matched resistance. The opkimuen $v$ angle is chosen to. align the radiation Coma from each arm in the one common deresed direction. The principles involved are the same as those for the resonant $V$ antenna.

The rhombic antenna can also be made into a traveling-wave antenna by inserting a suitable seristance at the vertex farthest away from the feed end The current on each wire in the rhombs will then be a traveling current wave analogow to that On a transmission line terminated in a matched load. The nominal input hesitance of the rhombic antenna is in the range of 700 to $800 \Omega$.

The traveling-wave antennas have the advantage that the input impedance is mostly resistive and relatively independent of frequency. Thus these antennas will operate over a fairly bread frequency band. The limiting factor is primarily the misalignment of the lobes that takes place as the Frequent is changed.

## Ground Interference Effects

Let us consider the long noxiziontras- withe antenna as shewn in the figure below. Fire aacliation at The angle th the field has a Contribution from the diet radiation from the outema and from radiation reflected from the grand at the angle y. The reflected radiation undergoes a propagation phase delay equivalent to that from the image of the andenna in the ground If the field radiated by the antenna in Free space is

$$
E=f(x) \frac{t^{-j k i r}}{4 \pi n}
$$

the total field obtained by taking the reflected field trod account will be

$$
E=f(\psi) \frac{e^{-j k c}}{4 \pi r}\left(1+\rho e^{\left.j \alpha_{0}-j 2 k \sin \psi\right)}\right.
$$

Where $f e^{i x}$ is the reflection Coefficient as the ground and incing is the ext ha propagation offitarice sattroduced then the antenna is at a height ho Thill expression is seen to be of the same form at that occuring in surat theory With the array factor being

Image antenna

The reflection sufficient depends on the Conductivity of the ground, on the grazing angle
$\psi$, and on whet hen the pred is vertically on
homzancally polarised. For there redon no
Siple evaluation of the last equation is not possible. It is often possible to assume that the ground acts a a perfectly conducting surface withow devious error. In this care $\rho=1$ and $\alpha=\pi$ for hasizantal polarization and $\alpha=0$ for vertical polarization with the ideasicel conditions the array factor becomes

$$
\begin{aligned}
& |F(\psi)|=a|\sin (k \cdot h s i n \varphi)| \text { hisezenal pelaizatos } \\
& {[f(x)\}=2\left\{63\left(\mathrm{~K}_{0} \text { hisiny }\right)\right\} \text { vearical palarizatis }}
\end{aligned}
$$

Fo, a horizontal wire the reflected field it out of phase lith the direct satiation, a) shown in the figure in the prexiow pages that is, the image current is out of phase With the antenna current, io the appropriate array factor to we is i $|\sin (k \operatorname{cosin} y)|$, which as approximately equal to $2 /$ Sinkahyl for the range of angles "y of intercut in practice the soong factor is shown in the figure below for wive of $h$ corresperinding to $\frac{\lambda_{0}}{4}, \frac{\lambda}{2}$. and $\lambda_{\text {a }}$ IV is readily seen that the reflection from the ground will reduce the far frets rigrifica wily at low elevation angles antes the antenna height it targe enough so that the array faction itself will exhibit a beebe maximum in the devised direction for example for a maximum of $\psi=20^{\circ}$, we require kory $=90^{\circ}$ as $\mathrm{k}=2.25 \% / \pi=0.72 \pi$. With this height the free-2pace field is doubled in value by the
reflected field adoring in phase with an actual Ground the reflection coefficient o will be less than unity; nevexthetess a rignificart reinforcement of the direct radiation will occur when the antenna height is sappropreately chosen

For surface waver / propagation at frequendia below $211 H z$ therizontany polarized field me attencia ked much moke rapidly than Verkicaury polarized fields. For this reason horizontally oxienteot long-hine antennas are normally not wed below 2binz. In the short wave band from 2 to 80 MHz , where propagation it via torospheme reflection, long-wire cutennas are effective and because of their simple structure ares Commonly wed Rhombic and $v$ antennas apo find tome application at frequencies from so ko bo mut z.

AlL VLF, LF and MF antennas, as well as many HF antennas are made vertically polarized because of the proximity of ground. The se are advantages in swing hoxzontally polarised artien af hughes frequencies, especially since most man-macle noise has vertical potablamzatio

## MODULE-V

## USASE OF EM WAVES IN RADAR

## RADAR SYSTEM

Radar is an electromagnetic system for the detection and location of reflecting objects such as aircraft, ships, spacecraft, vehicles,, people and the natural environment. It operates by radiating energy into space and detecting the echo signal reflected from an object, or target. The reflected energy that is returned to the radar not only indicates the presence of a target, but by comparing the received echo signal with the signal that was transmitted, its location can be determined along with other target-related information. Radar can perform its function at long or short distances and under conditions impervious to optical and infrared sensors. It can operate in darkness, haze, fog, rain and snow. Its ability to measure distance with high accuracy and in all weather is one of its most important attributes.

The basic principle of radar is shown in Fig.1. A transmitter generates an electromagnetic wave (such as a short pulse of sinewave) that is radiated into space by an antenna. A portion of the transmitted energy is intercepted by the target and radiated in many directions. The reradiation directed back towards the radar is collected by the radar antenna, which delivers it to a receiver. The receiver processes it to detect the presence of the target and determine its location. A single antenna is usually used on a time-shared basis for both transmitting and receiving when the radar waveform is a repetitive series of pulses. The range, or distance, to a target is found by measuring the time it takes for the radar signal to travel to the target and return back to the radar. This is called range in radar nomenclature. The target's location in angle can be found from the direction of the narrow-beamwidth radar antenna points when the received echo signal is of maximum amplitude. If the target is in motion, there is a shift in the frequency of the received echo signal due to the Doppler effect. This frequency shift is proportional to the velocity of the target relative to the radar (also called the radial velocity). The Doppler frequency shift is widely used in radar as the basis of separating desired moving targets from fixed unwanted "clutter" echoes reflected from the natural environment such as land, sea or rain. Radar can also provide information about the nature of the target being observed.

The term radar refers to radio detection and ranging. The name reflects the importance placed by the early workers in this field on the need of a device to detect the presence of a target and to measure its range. Although modern radar can extract more information from a target's echo signal than its range, the measurement of range is still one of its most important functions. There are no competitive techniques that can accurately measure long ranges in both clear and adverse weather as well as can radar.

Radars can operate in various modes by radiating different frequencies, with different polarizations. The radar can also employ various waveforms with different pulse widths, pulse repetition frequencies, or other modulations; and different forms of processing different types of clutter, interference, and jamming. The various waveforms and processing need to be selected wisely. A trained operator can fulfill this function, but an
operator can become overloaded. When there are many available system options, the radar can be designed to automatically determine the proper mode of operation and execute what is required to implement it. The mode of radar operation is often changed as a function of the antenna look-direction and/or range, according to the nature of the environments.
Conventional radars generally operate in the microwave region. Operational radars in the past have been at frequencies ranging from about 100 MHz to 36 MHz , which covers more than eight octaves. These are not necessarily the limits. Operational HF over-thehorizon radars operate at frequencies as low as a few megahertz. At the other end of the spectrum, experimental millimeter wave radars have been at frequencies higher than 240 GHz . The various frequency ranges of operation as used in radar systems is shown in Table 1.

Table No. 1 Part of electromagnetic spectrum used in RADAR applications

| Band Designation | Nominal Frequency range | Specific frequency ranges |
| :---: | :---: | :---: |
| HF | $3-30 \mathrm{MHz}$ |  |
| VHF | $30-300 \mathrm{MHz}$ | $138-144 \mathrm{MHz}$ |
|  |  | $216-225 \mathrm{MHz}$ |
| UHF | $300-1000 \mathrm{MHz}$ | $420-450 \mathrm{MHz}$ |
|  |  | $850-942 \mathrm{MHz}$ |
| L | $1-2 \mathrm{GHz}$ | $1215-1400 \mathrm{MHz}$ |
| S | $2-4 \mathrm{GHz}$ | $2300-2500 \mathrm{MHz}$ |
|  |  | $2700-3700 \mathrm{MHz}$ |
| C | $4-8 \mathrm{GHz}$ | $5250-5925 \mathrm{MHz}$ |
| X | $8-12 \mathrm{GHz}$ | $8500-10,680 \mathrm{MHz}$ |
| Ku | $12-18 \mathrm{GHz}$ | $13.4-14.0 \mathrm{GHz}$ |
|  |  | $15.7-17.7 \mathrm{GHz}$ |
| K | $18-27 \mathrm{GHz}$ | $24.05-24.25 \mathrm{GHz}$ |
| Ka | $27-40 \mathrm{GHz}$ | $33.4-36 \mathrm{GHz}$ |
| V | $40-75 \mathrm{GHz}$ | $59-64 \mathrm{GHz}$ |
| W | $75-110 \mathrm{GHz}$ | $76-81 \mathrm{GHz}$ |
|  |  | $92-100 \mathrm{GHz}$ |
| mm | $110-130 \mathrm{GHz}$ | $126-142 \mathrm{GHz}$ |
|  |  | $144-149 \mathrm{GHz}$ |
|  |  | $231-235 \mathrm{GHz}$ |
|  |  | $238-248 \mathrm{GHz}$ |

## RANGE OF A TARGET:

The most common radar signal, or waveform is a series of short duration, somewhat rectangular-shaped pulses modulating a sinewave carrier. (This is sometimes called a pulse train.) The range to a target is determined by the time $t_{r}$ it takes the radar signal to travel to the target and back. As electromagnetic waves travel in free space with the speed
of light, the time for the signal to travel to a target located at range $R$ and return back to the radar is $\frac{2 R}{c}$, the range to a target is

$$
R_{r}=\frac{c t_{r}}{2}
$$

This means that one microsecond of round-trip travel corresponds to a distance of 150 meters. (How to measure this time of return? Will autocorrelation of the two signals is useful?)

Maximum unambiguous range:
Once a signal is radiated into space by a radar, sufficient time must elapse to allow all echo signals to return to the radar before the next pulse is transmitted. The rate at which pulses may be transmitted, therefore is determined by the longest range at which targets are expected. If the time between pulses $t_{p}$ is too short, an echo signal from a long-range target might arrive after the transmission of the next pulse and be mistakenly associated with that pulse rather than the actual pulse transmitted earlier. This can result in an incorrect or ambiguous measurement of the range. Echoes that arrive after the transmission of the next pulse are called second-time-around echoes (or multiple-timearound echoes if from even earlier pulses). Such an echo would appear to be at a closer range than actual and its range measurement could be misleading if it were not known to be a second-time-around echo. The range beyond which targets appear as second-timearound echoes is the maximum unambiguous range $R_{u n}$, and is given by

$$
R_{u n}=\frac{c t_{p}}{2}=\frac{c}{2 f_{p}}
$$

where $t_{p}$ is the pulse repetition period and $f_{p}=1 / t_{p}$ is the pulse repetition frequency (prf), usually given in hertz or pulses per second (pps). A plot of the maximum unambiguous range as a function of the prf is shown in Fig.2. The term pulse repetition rate is also used interchangeably with pulse repetition frequency.

## THE SIMPLE FORM OF RADAR EQUATION

The radar equation relates the range of a radar to the characteristics of the transmitter, receiver, antenna, target, and the environment. It is useful not only for determining the maximum range at which a particular radar can detect a target, but it can serve as a means for understanding the factors affecting radar performance. It is also an important tool to aid in radar system design. In this section, the simple form of the radar range equation is derived.

If the transmitter power $P_{t}$ is radiated by an isotropic antenna, the power density at a distance $R$ from the radar is equal to the radiated power divided by the surface area $4 \pi R^{2}$ of an imaginary surface of radius $R$, or

Power density at range $R$ from an isotropic antenna $=\frac{P_{t}}{4 \pi R^{2}}$
Power density is measured in units of watts per square meter. Radars, however, employ directive antennas (with narrow beamwidths) to concentrate the radiated power $P_{t}$ in a particular direction. The gain of an antenna is a measure of the increased power density radiated in some direction as compared to the power density that would appear in that direction from an isotropic antenna. The power density at the target from a directive antenna with a transmitting gain $G$ is
Power density at range $R$ from a directive antenna $=\frac{P_{t} G}{4 \pi R^{2}}$
The target intercepts a portion of the incident energy and reradiates it in various directions. It is only the power density reradiated in the direction of the radar (the echo signal) that is of interest. The radar cross section of the target determines the power density returned to the radar for a particular power density incident on the target. It is denoted by $\sigma$ and is often called, for short, target cross section, radar cross section, or simply cross section. The radar cross section is defined by the following
Reradiated power density back at the radar $=\frac{P_{t}}{4 \pi R^{2}} \cdot \frac{\sigma}{4 \pi R^{2}}$
The radar cross section has units of area, but it can be misleading to associate the radar cross section directly with the target's physical size. The radar antenna captures a portion of the echo energy incident on it. The power received by the radar is given as the product of the incident power density times the effective area $A_{e}$ of the receiving antenna. The effective area is related to the physical area by $A_{e}=\rho_{a} A$ where $\rho_{a}$ is the antenna aperture efficiency. The received signal power is then

$$
P_{r}=\frac{P_{t} G}{4 \pi R^{2}} \cdot \frac{\sigma}{4 \pi R^{2}} \cdot A_{e}=\frac{P_{t} G A_{e} \sigma}{(4 \pi)^{2} R^{4}}
$$

The maximum range of a radar $R_{\max }$ is the distance beyond which the target cannot be detected. It occurs when the received signal power $P_{r}$ just equals the minimum detectable signal $S_{\text {min }}$. Substituting $S_{\text {min }}=P_{r}$, and rearranging the terms, we obtain

$$
R_{\max }=\left[\frac{P_{t} G A_{e} \sigma}{(4 \pi)^{2} S_{\min }}\right]^{1 / 4}
$$

This is the fundamental form of the radar equation. It is also called radar range equation. The important parameters are the transmitting gain and the receiving effective area. The transmitter power has not been specified as either the average or the peak power. It depends on how $S_{\text {min }}$ is defined. Here, $P_{t}$ denotes the peak power.
If the same antenna is used for transmitting and receiving, as it usually is in radar, the relationship between the transmit gain $G$ and the receiver effective area $A_{e}$ is

$$
G=\frac{4 \pi A_{e}}{\lambda^{2}}=\frac{4 \pi \rho_{a} A}{\lambda^{2}}
$$

Use of these expressions yield two other forms of radar equation.

$$
\begin{gathered}
R_{\max }=\left[\frac{P_{t} G^{2} \lambda^{2} \sigma}{(4 \pi)^{3} S_{\min }}\right]^{1 / 4} \\
R_{\max }=\left[\frac{P_{t} A_{e}^{2} \sigma}{4 \pi \lambda^{2} S_{\min }}\right]
\end{gathered}
$$

A major advance in radar was made with the invention of the high-power microwave cavity magnetron at the University of Birmingham in England early in WW II. The magnetron dramatically changed the nature of radar as it existed up to that time by allowing the development of radars with small antennas that could be carried on ships and aircraft, and by land-mobile systems. Most countries involved in early radar research recognized the importance of obtaining high power at microwave frequencies and tried to push the conventional magnetron upwards in power.

The use of the Doppler effect in the MTI pulse radar was perfected to separate desired aircraft targets from undesired large underground echoes.

High-power stable amplifiers such as klystron, traveling wave tube, and solid state transistor allowed better application of the Doppler effect, use of sophisticated waveforms, and much higher power than could be obtained with the magnetron.

Highly accurate angle tracking of targets became practical with monopulse radar.
Pulse compression allowed the use of long waveforms to obtain high energy and simultaneously achieve the resolution of a short pulse by internal modulation of the long pulse.

The airborne synthetic aperture radar (SAR) provided high resolution map-like imaging of ground scenes.

Airborne radars using Doppler processing methods gave rise to airborne MTI and pulse Doppler radars, which were able to detect aircraft in the midst of heavy ground clutter.

The electronically steered phased array antenna offered rapid beam steering without mechanical movement of the antenna.

HF over-the-horizon radar extended the detection range for aircraft by a factor of ten, to almost 2000 nmi . Radar became more than a blob detector by extracting information from the echo signal to provide target recognition.

Radar has become an important tool for meteorologist and as an aid for safe efficient air travel by observing and measuring precipitation, warning of dangerous wind shear and other hazardous weather conditions, and for providing timely measurements of the vertical profile of wind speed and direction.

The rapid advances in digital technology made theoretical capabilities with digital signal processing and digital data processing.

The ability of a radar receiver to detect a weak signal is limited by the ever-present noise that occupies the same part of the frequency spectrum as the signal. The weakest signal that can just be detected by a receiver is the minimum detectable signal. Use of the minimum detectable signal is not common to detect echo signals from targets.

Detection of a radar signal is based on establishing a threshold at the output of the receiver. If the receiver output is large enough to exceed threshold, a target is said to be present. If the receiver output is not of sufficient amplitude to cross the threshold, only noise is said to be present. This is called threshold detection.

When a large echo signal from a target is present, it can be recognized on the basis of its amplitude relative to the rms noise level. If the threshold level is set properly, the receiver output should not normally exceed the threshold if a strong target echo signal were present along with the noise. If the threshold level were too low, noise might exceed it and be mistaken for a target. This is called a false alarm. If the threshold were set too high, noise might not be large enough to cause false alarms, but the weak target echoes might not exceed the threshold and would not be detected. When this occurs, it is called a missed detection. The threshold is set according to the classical detection theory.

Almost all radars employ matched filters for maximization of signal to noise ratio. A matched filter does not preserve the shape of the waveform.

The selection of a proper threshold is a compromise that depends on how important it is to avoid the mistake of (1) failing to recognize a target signal that is present (missed detection) or (2) falsely indicating the presence of a target signal when none exists (false alarm). The SNR is a better measure of a radar's detection performance than is the minimum detectable signal.

At microwave frequencies, the noise with which the target echo signal competes is usually generated within the receiver itself. If the radar were to operate in a perfectly noise-free environment so that no external sources of noise accompany the target signal, and if the receiver itself were so perfect that it did not generate any excess noise, there would still be noise generated by the thermal agitation of electrons in the ohmic portion of the receiver input stages. This is called thermal noise. Its magnitude is proportional to the bandwidth and the absolute temperature of the ohmic portions of the input circuit. The available thermal noise power generated at the input of a receiver of bandwidth $B_{n}$ at a temperature $T$ is

$$
N=k T B_{n}
$$

where $k=1.38 \times 10^{-23} \mathrm{~J} / \mathrm{deg}$. The term available means that the device is operated with a matched input and a matched load. The bandwidth of a superheterodyne receiver (and almost all radar receivers are of this type) is taken to be that of the IF amplifier (or matched filter).
$\int_{0}^{\infty}|H(f)|^{2} d f$
The bandwidth $B_{n}=0$

This equation states that the noise bandwidth is the bandwidth of the equivalent rectangular filter whose noise power output is the same as the filter with the frequency response function $H(f)$. Noise bandwidth is not the same as the more familiar half power bandwidth. The half power bandwidth is a reasonable approximation for many practical radar receivers. Thus the half power bandwidth $B$ is usually used to approximate the noise bandwidth $B_{n}$.

The noise power in practical receivers is greater than that from thermal noise alone. The measure of the noise out of a real receiver to that from the ideal receiver with only thermal noise is called noise figure and is defined as

$$
F_{n}=\frac{N_{\text {out }}}{k T_{0} B G_{a}}
$$

where $N_{\text {out }}=$ noise out of the receiver, and $G_{a}$ is the available gain. The factor $k T_{0}=4 \times 10^{-21} \mathrm{~W} / \mathrm{Hz}$ at room temperature. The available gain $G_{a}$ is the ratio of the signal out, $S_{\text {out }}$ to the signal $S_{\text {in }}$ with both the output and the input matched to deliver maximum output power. The input noise $N_{\text {in }}$ in an ideal receiver is equal to $k T_{0} B_{n}$. Hence the noise figure can be written as

$$
F_{n}=\frac{S_{\text {in }} / N_{\text {in }}}{S_{\text {out }} / N_{\text {out }}}
$$

This equation shows that the noise figure may be interpreted as a measure of the degradation of the signal-to-noise ratio as the signal passes through the receiver.

The input signal is

$$
S_{\text {in }}=\frac{k T_{0} B F_{n} S_{\text {out }}}{N_{\text {out }}}
$$

If the minimum detectable signal $S_{\min }$ is that value of $S_{\text {in }}$ which corresponds to the minimum detectable SNR at the output of the IF, $\left(\frac{S_{\text {out }}}{N_{\text {out }}}\right)_{\min }$ becomes

$$
S_{\min }=k T_{0} B F_{n}\left(\frac{S_{\text {out }}}{N_{\text {out }}}\right)_{\min }
$$

Use of this gives us another expression for the maximum range of the radar

$$
R_{\max }^{4}=\frac{P_{t} G A_{e} \sigma}{(4 \pi)^{2} k T_{0} B F_{n}(S / N)_{\min }}
$$

From the above equation we do observe that the minimum detectable signal is replaced in the radar equation by the minimum detectable $\operatorname{SNR}(S / N)_{\min }$. The advantage is
that $(S / N)_{\min }$ is independent of the receiver bandwidth and noise figure. It can be expressed in terms of the probability of detection and probability of false alarm, two parameters that can be related to the radar user's needs.

The signal-to-noise ratio in the above is that at the output of the IF amplifier, since maximizing the signal-to-noise ratio at the output of the IF is equivalent to the video output where threshold decision is made.

## TRANSMITTER POWER

The power $P_{t}$ is the peak pulse power. The average power $P_{a v}$ of a radar is also of interest since it is a more important measure of a radar performance than the peak power. It is defined as the average transmitter power over the duration of the total transmission. If the transmitter waveform is a train of rectangular pulses of width $\tau$ and constant-pulse-repetition period $T_{p}=1 / f_{p}$, the average power is related to the peak power by

$$
P_{a v}=\frac{P_{t} \tau}{T_{p}}=P_{t} f_{p}
$$

The radar duty cycle can be expressed as $P_{a v} / P_{t}$ or $\tau / T_{p}$. Pulse radars might typically have duty cycles of from 0.001 to 0.5 , more or less. A CW radar has a duty cycle of unity. The duty cycle depends on the type of waveform, the pulse width, whether or not pulse compression is used, and problems associated with range and Doppler ambiguities, and the type of transmitter employed.

Writing the radar range equation in terms of average power, we obtain

$$
R_{\max }^{4}=\frac{P_{a v} G A_{e} \sigma n E_{i}(n)}{(4 \pi)^{2} k T_{0} F_{n}(B \tau)(S / N)_{1} f_{p}}
$$

For simplicity, the fluctuation loss $L_{f}$ has been set to unity in this equation. The energy per pulse is $E_{p}=P_{t} \tau$ is also equal to $P_{a v} / f_{p}$. The radar range equation can also be written as, in terms of energy

$$
R_{\max }^{4}=\frac{E_{p} G A_{e} \sigma n E_{i}(n)}{(4 \pi)^{2} k T_{0} F_{n}(B \tau)(S / N)_{1}}=\frac{E_{T} G A_{e} \sigma E_{i}(n)}{(4 \pi)^{2} k T_{0} F_{n}(B \tau)(S / N)_{1}}
$$

where $E_{T}$ is the total energy of the $n$ pulses, which equals $n E_{p}$.

## PULSE REPETITION FREQUENCY

The pulse repetition frequency (prf) is determined by the maximum unambiguous range beyond which targets are not expected. The prf corresponding to a maximum unambiguous range $R_{u n}$ is given by $f_{p}=2 R_{u n} / c$ where $c$ is the velocity of propagation. There are times, however, when echoes might appear from beyond the maximum unambiguous range, especially for some unusually large target or clutter source (such as a
mountain), or when anomalous propagation conditions occur to extend the normal range of the radar beyond the horizon. Echo signals that arrive at a time later than the pulserepetition period are called second-time-around echoes. They are also called multiple-time-around echoes, particularly when they arrive from ranges greater than $2 R_{u n}$. The apparent range of these unambiguous echoes can result in error and confusion. Another problem with multiple-time-around echoes is that clutter echoes from ranges greater than $R_{u n}$ can mask unambiguous target echoes at the shorter ranges.

Some types of radars, such as pulse Doppler radars, always operate with a prf that can result in range ambiguities. Range ambiguities are tolerated in a pulse Doppler radar in order to achieve the benefits of a high PRF when detecting moving targets in the midst of clutter. Resolving the range ambiguities is an important part of operation of pulse Doppler radars.

The existence of multiple-time-around echoes cannot be readily recognized with a constant prf waveform. Let us consider three targets labeled $\mathrm{A}, \mathrm{B}$ and C in the following figure. Target A is within the unambiguous range interval $R_{u n}$, target B is at a distance greater than $R_{u n}$ but less than $2 R_{u n}$, while target C is greater than $2 R_{u n}$ but less than $3 R_{u n}$. Target B is a second-time-around echo; target C is a multiple-time-around echo. When these three pulse repetition intervals, or sweeps, are superimposed on a radar display, the ambiguous echoes look no different from the unambiguous range echo of A. Only the range of A is correct, but it cannot be determined from this display that the other two are not at their apparent range.

Ambiguous-range echoes can be recognized by changing the PRF of the radar. When the PRF is changed, the unambiguous echo (at a range less than $R_{u n}$ ) remains at its true range. Ambiguous-range echoes, however, appear at different apparent ranges for each PRF. An example of how these three echoes might appear on an A-scope is shown in the figure below. A similar effect would be seen on the PPI. Thus the ambiguous target ranges can be readily identified.

If the first pulse repetition frequency $f_{1}$ has an unambiguous range $R_{u n 1}$ and if the apparent range measured with prf $f_{1}$ is denoted $R_{1}$, then the true range is one of the following

$$
R_{\text {true }}=R_{1,} \text { or }\left(R_{1}+R_{u n 1}\right) \text {, or }\left(R_{1}+2 R_{u n 1}\right), \text { or } \ldots
$$

Anyone of these might be the true range. To find which is correct, the prf is changed to $f_{2}$ with an unambiguous range $R_{u n 2}$, and if the apparent measured range is $R_{2}$, the true range is one of the following

$$
R_{\text {true }}=R_{2}, \text { or }\left(R_{1}+R_{\text {un } 2}\right) \text {, or }\left(R_{1}+2 R_{u n 2}\right), \text { or } \ldots
$$

The correct range is that value which is the same with the two prfs. In theory, two prfs can resolve the range ambiguity; but in practice, three prfs are often used for increased accuracy and avoiding false values.

The pulse repetition frequency may be changed pulse to pulse, every half beamwidth (with a scanning antenna), or on every rotation of the antenna.

## ANTENNA PARAMETRS

Almost all radars use directive antennas with relatively narrow beamwidths that direct the energy in a particular direction. The antenna is an important part of a radar. It serves to place energy on target during transmission, collect the received echo energy reflected from the target, and determine the angular location of the target. There is always a trade between antenna size and transmitter size when long-range performance is required. If one is small the other must be large to make up for it. This is one reason why large antennas are generally desirable in most radar applications when practical considerations do not limit their physical size. Thus far, the antenna has been thought of as a mechanically steered reflector. Radar antennas can also be electronically steered phased arrays.

Antenna Gain The antenna gain $G(\theta, \phi)$ is a measure of the power per unit solid angle radiated in a particular direction by a directive antenna compared to the power per unit solid angle which would have radiated by an omnidirectional antenna with $100 \%$ efficiency. The gain of an antenna is

$$
G(\theta, \phi)=\frac{\text { power radiated per unit solid angle at an azimuth } \theta \text { and anelevation } \phi}{(\text { power accepted by the antenna from the transmitter }) / 4 \pi}
$$

This is the power gain and is a function of direction. If it is greater than unity in some directions, it must be less than unity in other directions. There is also the directive gain, which has a similar definition except that the denominator is the power radiated by the antenna per $4 \pi$ steradians rather the power accepted from the transmitter. The difference between the two is that the power gain accounts for losses within the antenna. The power gain is more appropriate for the radar equation than the directive again, although there is usually little difference between the two in practical radar antennas, except for the phased array. The power gain and the directive antenna of a radar antenna are usually considered to be the same here. When they are significantly different, then the distinction between the two must be made. The maximum power gain is denoted as $G$.

Effective Area and Beamwidth

The directive gain $G$ and the effective area $A_{e}$ of a lossless antenna are related as

$$
G=\frac{4 \pi A_{e}}{\lambda^{2}}=\frac{4 \pi \rho_{a} A}{\lambda^{2}}
$$

where $\rho_{a}$ is the antenna aperture efficiency and $A$ is the physical area of the antenna. The gain of an antenna is approximately equal to

$$
G \approx \frac{26,000}{\theta_{B} \phi_{B}}
$$

where $\theta_{B}$ and $\phi_{B}$ are the azimuth and elevation half-power beam widths, respectively, in degrees. This results in a gain of 44 dB for a one-degree pencil beam. The half-power beamwidth of an antenna also depends on the nature of the aperture illumination and, therefore, the sidelobe level. When no specific information is available regarding the nature of the antenna, the following relation holds good between beamwidth and antenna dimension

$$
\theta_{B}=\frac{65 \lambda}{D} \text { degrees }
$$

where the wavelength $\lambda$ has the same units as the aperture dimension $D$. When $D$ is the horizontal dimension of the antenna, the beamwidth $\theta_{B}$ is the azimuth beamwidth; when $D$ is the vertical dimension, $\theta_{B}$ is the elevation beamwidth. The above expression might apply for an antenna with 25 to 28 dB peak sidelobe level.

The half-power beamwidth of an antenna can be measured somewhat accurately, but the antenna gain "is probably one of the least accurate measurements made on an antenna system".

## Revisit Time

The revisit time is the time that an antenna takes to return to view the same region of space. It usually represents a compromise between (1) the need to collect sufficient energy (a sufficient number of pulses) for the detection of weak targets and (2) the need to have a rapid re-measurement of the location of a moving target so as to quickly determine its trajectory. The revisit time is also called the scan time; and both are inversely related to the rotation rate (rpm) of a scanning antenna. The revisit time of longrange civil air-traffic-control radars are generally in the vicinity of 10 to 12 s , corresponding to an antenna rotation rate of 6 to 5 rpm . Military air-surveillance radars, unlike civil radars, have to detect and track high-speed maneuvering targets. A revisit time of 10 to 12 seconds is too long. Revisit times for long-range military radars are more like 4 seconds ( 15 rpm ). Short-range military radars that must detect an quickly respond to low-flying high-speed targets that pop up over the horizon generally require revisit times of 1 to 2 seconds ( 60 to 30 rpm ), depending on the radar type and design. A small civil marine radar commonly found on boats and ships might have a rotation rate of about 20 rpm (3-s revisit time). High-resolution radars which monitor the ground traffic at major airports, such as the ASDE generally have rotation rates 60 rpm .

## BEAM SHAPE

Radars employ either fan beams or pencil beams. The beam width of the pencil beam shown in diagrams below in the horizontal plane is equal or almost equal to the beam width in the vertical plane. Its beamwidth is generally less than a few degrees; one degree might be typical. It is found in radars that must have accurate location measurement and resolution in both azimuth and elevation. The pencil beam is popular for tracking radars, 3-D radars (rotating air-surveillance radars that obtain elevation angle measurement as well as azimuth and range), and many phased array radars.

The fan-beam antenna shown in figure below has one angle small compared to the other. In air-surveillance radars that use fan beams, the azimuth beamwidth might typically be one or a few degrees, while the elevation beamwidth might be perhaps four to ten times the azimuth beamwidth. Fan beams are found with 2-D (range and azimuth) airsurveillance radars that have to search out a large volume of space. The narrow beamwidth is in the horizontal coordinate so as to obtain a good azimuth angle measurement. The elevation beamwidth is broad in order to obtain good elevation coverage, but at the sacrifice for an elevation angle measurement.

A single pencil beam has difficulty searching out a large angular volume. Employing a number of scanning pencil beams can solve this problem, as is found in some 3-D radars. Sometimes in a 3-D radar a stacked-beam coverage is used in the vertical dimension. This consists of a number of contiguous fixed pencil beams as shown in the following figure. Six to sixteen contiguous beams have been typical in the past.

Usually the shape of a fan beam has to be modified to obtain more complete coverage. An example is the cosecant-squared shaped beam as shown below.

## RECEIVER BANDWIDTH REQUIREMENTS

The bandwidth of the receiver corresponds to the bandwidth of the transmitter and its pulse width. The narrower the pulses, the greater is the IF and video bandwidth required. The RF bandwidth is normally greater than these, as in other receivers. With a given pulse duration $T$, the receiver bandwidth may still vary, depending on how many harmonics of the pulse repetition frequency are needed to provide a received pulse having a suitable shape. If vertical sides are required for the pulses in order to give a good resolution, a large bandwidth is required. The bandwidth must be increased if more information about the target is required, but too large a bandwidth will reduce the maximum range by admitting more noise.

The IF bandwidth of a radar receiver is made $n / T$, where $T$ is the pulse duration and $n$ is a number whose value ranges from under 1 to over 10 , depending on the circumstances. Values of $n$ from 1 to about 1.4 are the most common. Because pulse widths normally range from 0.1 to $10 \mu \mathrm{~s}$, the receiver bandwidth may lie between 200 KHz to over 10 MHz . Bandwidths from 1 to 2 MHz are the most common.

## ANTENNAS AND SCANNING

The majority of radar antennas use dipole or horn-fed paraboloid reflectors, or at least reflectors of a basically paraboloid shape, such as those shown in diagrams below. In each of the cut paraboloid, parabolic cylinder or pillbox reflectors, the beamwidth in the vertical direction will be much worse than in the horizontal direction, but this is immaterial in ground-to-ground or even air-to-ground radars. It has the advantages of allowing a significantly reduced antenna size and weight, reduced wind loading and smaller drive motors.

## ANTENNA SCANNING

Radar antennas are often made to scan a given area of the surrounding space, but the actual scanning pattern depends on the application. In the diagrams below, we show some typical scanning patterns.

The first of these is the simplest but has the disadvantage of scanning in the horizontal plane only. However, there are many applications of this type of scan in searching the horizon, e.g., in ship-to-ship radar. The nodding scan of Fig.b is an extension of this; the antenna is now rocked rapidly in elevation while it rotates more slowly in azimuth, and scanning in both planes is obtained. The system can be used to scan a limited sector or else it can be extended to cover the complete hemisphere. Another system capable of search over the complete hemisphere is the helical scanning system of Fig.c, in which the elevation of the antenna is raised slowly while it rotates more rapidly in azimuth. The antenna is returned to its starting point at the completion of the scanning cycle and typical speeds are a rotation of 6 rpm accompanied by a rise rate of $20^{\circ} /$ minute. Finally, if a limited area of more or less circular shape is to be covered, spiral scan may be used as shown in Fig.d.

## AUTOMATIC DETECTION

An operator viewing a PPI display or an A-scope "integrates" in his/her eye-brain combination the echo pulses available from the target. Although an operator in many cases can be as effective as an automatic integrator, performance is limited by operator fatigue, boredom overload and the integrating characteristics of the phosphor of the CRT display. With automatic detection by electronic means, the operator is not depended on to make the detection decision. Automatic detection is the name applied to the part of the radar that performs the operations required for the detection decision without operator intervention. The detection decision made by an automatic detector might be presented to an operator for action or to a computer for further processing.

In many aspects, automatic detection requires much better receiver design than when an operator makes the detection decision. Operators can recognize and ignore clutter echoes and interference that would limit the recognition abilities of some automatic devices. An operator might have better discrimination capabilities than automatic methods for sorting clutter and interference; but the automatic, computer-based decision devices can operate with far greater number of targets than an operator can handle.

Automatic detection of radar signals involves the following:

- Quantization of the radar coverage into range, and maybe angle, resolution cells.
- Sampling of the output of the range-resolution cells with at least one sample per cell, more than one sample when practical
- Analog-to-digital conversion of the analog signals
- Signal processing in the receiver to remove as much noise, clutter echoes, and interference as practicable before the detection decision is attempted.
- Integration of the available samples at each resolution cell.
- Constant-false-alarm rate (CFAR) circuitry to maintain the false-alarm rate when the receiver cannot remove all the clutter and interference
- Clutter map to provide the location of clutter so as to ignore known clutter echoes
- Threshold detection to select target echoes for further processing by an automatic tracker or other data processor
- Measurement of range and angle after the detection decision is made

Track While Scan (Limited Sector Scan) This is the name given to the tracking performed by a rotating-antenna air-surveillance radar which obtains target location updates each time the antenna beam rotates past the target, which might be from about 1 to 12 seconds. The name track while scan for this type of a radar is now seldom used since almost all modern air-surveillance radars provide the equivalent with what is called automatic detection and track (ADT). This type of antenna is used to scan a relatively narrow angular sector, usually in both azimuth and elevation. It combines the search function and the track function. Scanning may be performed with a single narrow-beamwidth pencil beam that might cover a rectangular sector in a raster fashion. Scanning can also be performed with two orthogonal fan beams, one hat scans in the azimuth and the other in elevation. TWS radars have been used in airport landing radars, airborne interceptors; and air-defense systems.

A difference between a continuous tracker and the TWS radar is that the angle-error signal in a continuous tracker is used in a closed-loop servo system to control the pointing of the antenna beam. In the TWS radar, however, there is no closed-loop positioning of the antenna. Its angle output is sent directly to a data processor. Another significant difference is that the TWS radar can provide simultaneous tracks on a number of targets within the sector of coverage, while the continuous tracker observes only a single target, which is why it is sometimes called a single-target tracker. With comparable transmitters and antennas, the energy available to perform tracking is less in a TWS radar than a STT since the TWS shares its radiated energy over an angular sector rather than concentrate it in the direction of a single target. In airborne-interceptor applications, the TWS radar might be preferred when multiple targets have to be maintained in track and the tracking accuracy only has to be good enough to launch missiles which contain their own guidance systems to home on the target. On the other hand, if highly accurate tracking is needed, a single-target tracker might be preferred.

Limited sector-scan TWS radars have been used in Precision Approach radars (PAR) or Ground-Controlled Approach (GCA) systems that guide aircraft to a landing. These radars allow aground controller to direct an aircraft to a safe landing in bad whether by tracking it as it lands. The ground controller communicates to the pilot directions to change his or her heading up,down,right, or left. In the control of aircraft landing, fan beams have been used which are electromechanically scanned over a narrow sector at a rate of twice per second. The azimuth sector that is scanned might be $20^{\circ}$ and the elevation sector 6 or $7^{0}$. Landing radars using limited-scan phased array antennas, such as the AN/TPS-19, operate differently than scanning fan beams
since they electronically scan a pencil beam over a region of $20^{\circ}$ in azimuth and $15^{0}$ in elevation at a rate of twice per second. The AN/TPN-19 used multiple beams to obtain a monopulse angle measurement. Being a phased array, the AN/TPN-19 could simultaneously track up to six aircraft at a $20-\mathrm{Hz}$ data rate. In the past, radars for the control of landing aircraft have been mainly used by the military. Civilian pilots prefer to use landing systems in which control of landing is in the aircraft's cockpit rather from a voice originating from the ground.

TWS radars have also been used successfully for the control of weapons in surface-to-air missile systems for both land and ship-based air defense, especially by the former Soviet Union.

Generally TWS systems using fan beams have some limitations compared to systems that operate with one or more pencil beams. The fan-beam system can see more rain and surface clutter, it is more vulnerable to electronic countermeasures, and there might be problems with associating multiple targets that appear in the two beams.

The advantage of TWS compared to a continuous tracker is that multiple targets can be tracked. Because it shares its energy over a region of space, the TWS radar needs to have a larger transmitter to obtain the same detection and tracking capabilities of a STT that dwells continuously on a single target. If the target's angle is found from the centroid of the angle measurements obtained as the TWS antenna beam scans past the target, inaccuracies can occur if the target signal fluctuates in amplitude. TWS radars are also more vulnerable to angle jamming than are continuous trackers. TWS radars can use monopulse angle measurements in an open-loop manner, similar to a phasedarray. Since the monopulse measurement is not made with closed-loop tracking, a TWS radar should not experience the wild fluctuations in angle caused by glint when there multiple scatterers within the resolution cell or when there is multipath at low elevation angles.

## MONOPULSE TRACKING

A monopulse tracker is defined as one in which information concerning the angular location of a target is obtained by comparison of signals received in two or more simultaneous beams. A measurement of angle may be made on the basis of a single pulse; hence the name monopulse. In practice, however, multiple pulses usually employed to increase the probability of detection, improve the accuracy of the angle estimate, and provide resolution in Doppler when necessary. By making an angle measurement based on the signals that appear simultaneously in more than one antenna beam, the accuracy is improved compared to time-shared single-beam tracking systems (such as conical scan or sequential lobing) which suffer degradation when the echo signal amplitude changes with time. Thus the accuracy of monopulse is not affected by amplitude fluctuations of the target echo. It is the preferred tracking technique when accurate angle measurements are required.

The monopulse angle method may be used in a tracking radar to develop an angle error signal in two orthogonal angle coordinates that mechanically drive the boresight of the tracking antenna using a closed-loop servo system to keep the boresight positioned in the direction of the moving target. In radars such as the phased-array, angle measurements can be obtained in an open-loop fashion by calibrating the errorsignal voltage in terms of angle.

There are several methods by which a monopulse angle measurement can be made. The most popular by far has been the amplitude-comparison monopulse which compares the amplitudes of the signals simultaneously received in multiple squinted beams to determine the angle. When the term monopulse is used by itself with no other description, it generally refers to the amplitude-comparison version.

